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CORONA J  
PERFORMANCE EVALUATION REPORT  
MISSION 1018-1 and 1018-2  
31 OCTOBER 1966

Approved:

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Manager  
Advanced Projects

Declassified and Released by the N R O  
In Accordance with E. O. 12958  
on NOV 26 1997

Approved:

[REDACTED]  
Manager~~TOP SECRET~~

## FOREWORD

This report details the performance of the payload system during the operational phase of the Program [REDACTED] Flight Test Vehicle 1612.

Lockheed Missiles and Space Company has the responsibility for evaluating payload performance under the Systems Integration and "J" System contracts.

This document is the final payload test and performance evaluation report for Missions 1018-1 and 1018-2 which was launched on 25 March 1965.

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## INTRODUCTION

This report presents the final performance evaluation of Missions 1018-1 and 1018-2 of the Corona Program. The purpose of this report is to define the performance characteristics of the J-19 payload system and to identify the source of in-flight anomalies.

The performance evaluation was jointly conducted by representatives of Lockheed Missiles and Space Company (LMSC) and ITEK at the facilities of NPIC and AFSPPF. The off-line evaluation using Corona engineering photography acquired over the United States was performed at the individual contractors plants.

The quantitative data used for this report is obtained from government organizations. The diffuse density data, and MTF/AIM resolution are produced by AFSPPF. The vehicle attitude error values, frame correlation times are made at NPIC who also supply the Processing Summary and MTF/AIM resolution reports published by [REDACTED]

[REDACTED]

Computer programs developed by A/P are utilized to calculate and plot the frequency distribution of the various contributors to image smear to permit analysis and correlation of the conditions of photography to the information content and quality of the acquired pictures. Computer analysis of the exposure, processing and illumination data provides the necessary data to analyze the exposure criteria selected for the mission.

## SECTION I

## SYSTEM PERFORMANCE

## A. MISSION OBJECTIVES

The payload section of Mission 1018, placed into orbit by Flight Test Vehicle #1612 and SLV-2A booster #429, consisted of two panoramic cameras, two Stellar-Index Cameras, two Mark 5A recovery capsules and a space structure to enclose the cameras and provide mounting surfaces for all equipments. Figure 1-1 presents an inboard profile of the J-19 payload system. This Corona "J" system is designed to acquire search and reconnaissance photography of selected areas of the earth from orbital altitudes. The planned mission was two, four day photographic periods with no deactivate period.

## B. MISSION DESCRIPTION

The payload was launched from Vandenberg Air Force Base (VAFB) at 2111:17 Z (1311:17 PST) on 25 March 1965. Ascent and injection were normal and the achieved orbit within nominal tolerances. Tracking and command support was effected by the Air Force Satellite Control Facility consisting of tracking and command stations at [REDACTED]

[REDACTED] under central control of the Satellite Test Center at Sunnyvale, California. Mission 1018-1 consisted of four days operation and was completed by air recovery on 29 March 1965. Mission 1018-2 followed immediately with no deactivate, and consisted of two days operation concluding with a successful air recovery on 31 March 1965.

The comparison of the planned and actual orbit parameters is tabulated as follows:

## ORBITAL PARAMETERS

<u>Parameter</u>	<u>Predicted</u>	<u>Orbit 33 Actuals</u>
Period (Min.)	88.98	88.96
Perigee (N. M.)	100.00	100.26
Apogee (N. M.)	144.20	148.60
Inclination (Deg.)	96.00	96.01
Perigee Latitude (Deg. N.)	22.06	40.37
Eccentricity	0.0063	0.0067

SCHEMATIC DRAWING - GOOGOL 1 SYSTEM

MISSION 1613

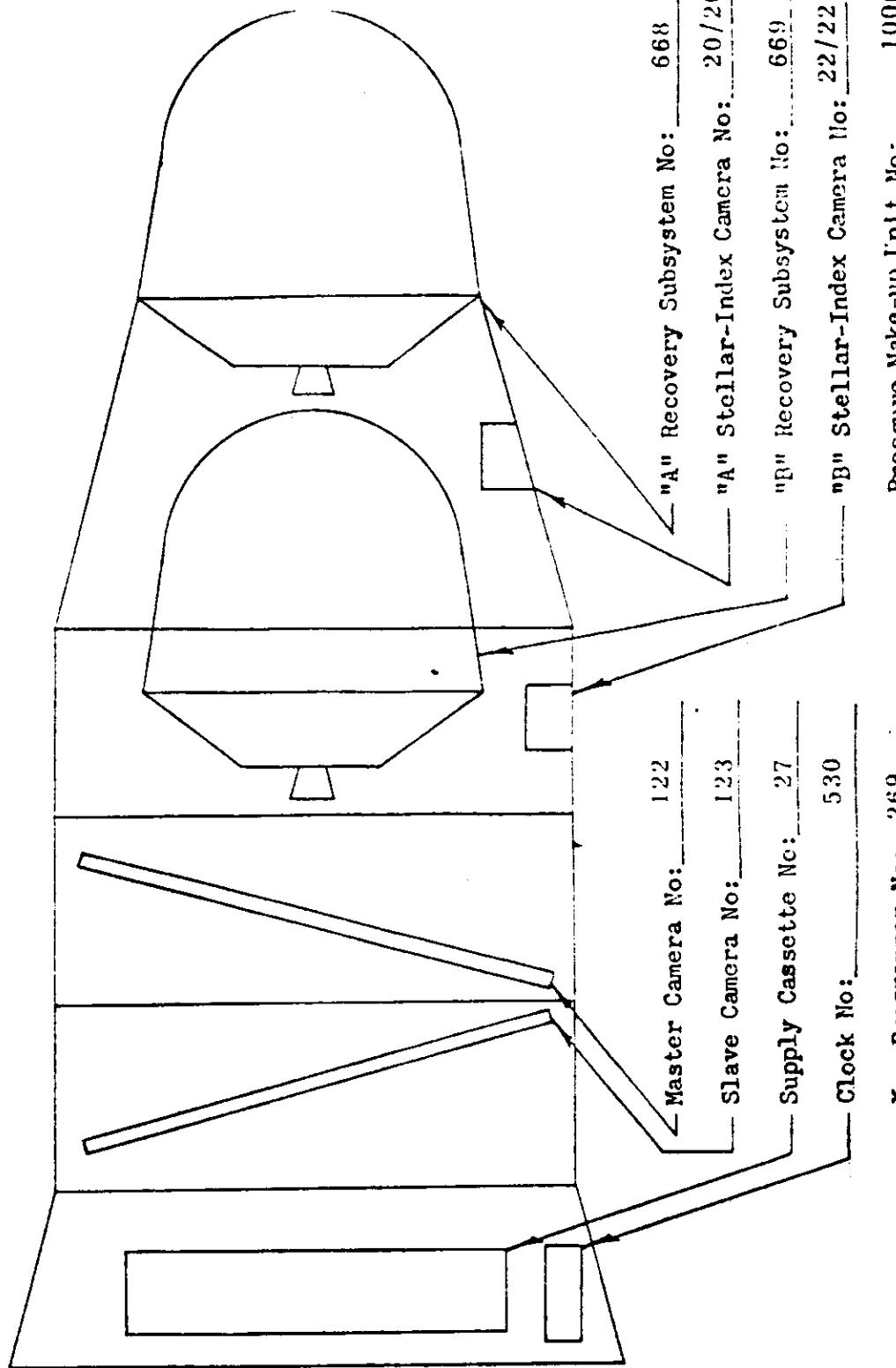


FIGURE 1-1

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### C. PANORAMIC CAMERAS

The Master and Slave panoramic cameras operated throughout both missions with no significant problems and produced good photographic coverage. The cloud cover and atmospheric haze observed in the photography was nominal. The thermal environment of the instruments were fairly constant throughout the flight due to a constant  $\beta$  angle.

### D. STELLAR-INDEX CAMERAS

The S/I Programmer failed during the third operation of Rev. 8 resulting in the loss of command capability to both the -1 and -2 stellar-index cameras.

### E. OTHER SUBSYSTEMS

The clock, instrumentation, pressure make-up, command and thermal control subsystems performed satisfactorily through both missions. The yaw programmer was activated at orbit 8 and functioned properly throughout both missions.

### F. CONCLUSIONS

Mission 1018 achieved the objective of acquiring high quality search and reconnaissance photography from orbital altitudes.

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## SECTION 2

## PRE-FLIGHT SYSTEMS TEST

## A. ENVIRONMENTAL TESTING

1. Test Objective

As a standard procedure, the J payload systems are subject to thermal/altitude environmental testing which simulates orbital environment. One of the purposes of this test is to demonstrate the system susceptibility to corona discharge. Such discharge fogs the film thus degrading the operational photography.

2. Test Summary

The Environmental Test for the J-19 payload system was conducted in the TASC chamber at Sunnyvale from 5 December to 12 December 1964. The test consisted of 3 days of operation in the "A" mode; a 1 day soak; and 3 days of "B" operation. The payload system was only operated during the daytime.

The J-19 payload system consisted of panoramic cameras 122 and 123. There were no Stellar-Index cameras installed.

Panoramic camera operation was satisfactory. Both instruments ran approximately 1.5 to 2% fast in the "A" mode and approximately 1 to 1.5% slow in the "B" mode.

Both "A" and "B" recovery sequences were satisfactory with the exception of the tape recorder data for the master instrument. This channel went to an out of hand voltage during cut and wrap. Transfer to the "B" bucket was normal and both instruments stowed properly.

A deactivate command was given between the "A" and "B" operations. Both instruments operated for 5 cycles and stowed normally.

The lens rotation monitor on the slave instrument became intermittent during the "B" mode. The 1000 count cycle counter on the slave instrument read 0.2 volts high in the 6 position.

The PMU operation appeared normal.

The clock performance was satisfactory. The Corona level on the processed payload met the acceptance criteria and is acceptable for flight.

### 3. Panoramic Camera Performance

Both instruments operated satisfactorily throughout the test. Film transport and instrument dynamic operation as monitored by the lens rotation, center of format, monitors, and supply and take-up idlers was normal. Instrument start and shutdown appeared normal.

The 99/101 percent clutch ratios were 5/5 to 5/7 for both instruments.

The cycle periods for both instruments were approximately 1.5 to 2% fast for the "A" mission and from 1 to 1.5% slow for the "B" mission. A tabulation of the rates are included as Tables 2-1 and 2-2.

The lens rotation monitor for the slave instrument became intermittent during the "B" mission. Post TASC adjustment and cleaning corrected this condition. The 6 position of the 1000 count cycle counter read 0.2 volts high. No corrective action was taken as the significance of the 1000 count does not constitute a primary flight item.

Corona discharge marking was not present in the master and only an occasional trace amount of start up Corona in the slave photography. The system was altitude tested between 1.8 and 50 microns internal pressure.

Scratches in the master and slave camera photography are attributed to the scan head rollers. A fix was made by Itek.

Heavy intermittent light leak fogging of master and slave photography was traced to a light leak in the silver dollar. A fix and subsequent lighting test verifying the light tight integrity of the system was performed.

Dendritic static spotting was observed at random points throughout the master payload to a density of approximately 0.06 above the base fog level. Some static marks occurred along the film edge. Dendritic marks are attributed to retrieval static and do not represent the expected performance of the flight system.

The horizon camera exposure lamps were inoperative for master and slave. Verification of flight acceptable horizon shutter operation was made prior to generation of the theodolite test film exhibit.

Slave instrument data recording was acceptable except for the 200 PPS timing track. Serial number, binary lamps, index lamps, and horizon camera fiducials were good. The 200 PPS timing track was acceptable for all formats except 17. The timing track became discontinuous exposing in spurts for 17 frames, approximately 2000 feet from the cut and wrap event.

#### 4. Instrumentation

The instrumentation system performance was satisfactory with the exception of the rotation monitor and cycle counter discussed above.

#### 5. Clock Performance

The clock performance was normal with the recorded error being approximately 4 milliseconds in 24 hours.

#### 6. Temperature Environment

Typical instrument temperatures recorded through the test are as follows:

<u>Mission</u>	<u>Orbit</u>	<u>Master Camera</u>	<u>Slave Camera</u>
A	1	85°	81°
A	15	88°	89°
B	5	70°	71°
B	16	72°	73°

The self-heating characteristics of the temperature sensors were determined during the TASC test. Results of these calibrations are included in Tables 2-3 and 2-4.

#### 7. Pressure Environment

The PMU system operated satisfactorily throughout the test. Figure 2-1 is a plot showing the pressure supply as a function of the operate time. This plot indicates no leakage in the system. The pressure increased to 85 microns when the PMU system was operated.

#### B. RESOLUTION TEST

The dynamic resolution test of the J-19 payload system was performed at the A/P facility on 15 December 1964. Each panoramic camera photographed high and low contrast resolution targets. The resulting through focus resolution data is shown in Figure 2-2 for the Master camera and in Figure 2-3 for the Slave camera.

#### C. LIGHT LEAK TEST

The examination of the film threaded in the J-19 system during the light leak test determined that no film fogging was present. The light leak integrity of the system was considered acceptable for flight.

J-19 122/123 ENVIRONMENTAL TEST CYCLE RATES 12-06-64 3840

REV/MOD	RAMP	T.U.R.	INST 122			INST 123			122/123	
			ACT.	NOM.	DEV.	ACT.	NOM.	DEV.	S:IFF.	
4 A 7 7	2280	2.450	2.492	-1.70		2.470	2.495	1.02	0.82	
4 A 8 2	340	5.400	5.478	1.42		5.480	5.531	0.92	1.48	
5 A 8 2	1730	2.200	2.215	0.67						
5 A 8 2	2250					2.270	2.315	1.95		
5 A 11 1	1460	2.620	2.668	1.81		2.630	2.669	1.47	0.38	
6 A 11 1	2040	2.290	2.330	1.73		2.290	2.335	1.94	-0.	
6 A 5 8	1090	2.850	2.867	0.58		2.860	2.866	0.22	0.35	
6 A 5 8	1440	2.440	2.466	1.05		2.460	2.469	0.37	0.82	
7 A 7 7	1180	2.930	2.944	0.47		2.950	2.943	-0.24	0.66	
8 A 7 7	2530	2.700	2.752	1.90		2.720	2.753	1.19	0.74	
8 A 4 1	1015	2.900	2.914	0.50		2.910	2.914	0.13	0.34	
8 A 4 1	1750	2.170	2.191	0.94						
9 A 11 1	890	4.350	4.386	0.82		4.380	4.400	0.46	0.69	
9 A 4 1	3210	3.710	3.797	2.26		3.750	3.800	1.30	1.08	
9 A 4 1	2620					2.460	2.505	1.79		
10 A 11 1	1880	2.270	2.310	1.73		2.270	2.315	1.96	-0.	
10 A 11 1	2970	4.360	4.469	2.44		4.410	4.485	1.67	1.15	
10 A 7 7	140	4.300	4.376	1.74		4.350	4.390	0.91	1.16	
11 A 7 7	1910	2.340	2.358	0.75		2.350	2.362	0.52	0.43	
11 A 7 7	2280	2.460	2.492	1.30		2.470	2.495	1.02	0.41	
12 A 8 2	1150	3.010	3.068	1.89		3.025	3.067	1.37	0.50	
12 A 8 2	1850					2.200	2.218	0.81		
13 A 4 1	2630	2.470	2.521	2.03		2.500	2.524	0.94	1.21	

O: \*Time up ramp not verified

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REV/RUD	RAMP	I.U.R.	1	1975	INST 122			INST 123			122/123	
					ACT.	NOM.	DEV.	ACT.	NOM.	DEV.	DIFF.	
13	A	11	1	1975	2.250	2.312	2.69	2.250	2.312	2.91	-0.	
14	A	11	1	3080	4.850	4.965	2.32	4.910	4.997	1.74	1.24	
1	0	7	7	390	4.140	4.140	0.04	4.160	4.154	-0.14	0.40	
2	8	4	1	1430	2.200	2.212	0.53					
2	8	4	1	1610				2.190	2.203	0.61		
4	0	7	7	2280	2.460	2.492	1.30	2.470	2.495	0.02	0.40	
5	8	11	1	1460	2.670	2.668	-0.06	2.670	2.669	-0.02	-0.	
6	8	11	1	2040	2.310	2.330	0.87	2.330	2.335	0.23	0.87	
6	8	5	8	1090	2.900	2.867	-1.16	2.900	2.866	-1.17	-0.	
6	8	5	8	1440	2.450	2.466	-0.97	2.500	2.469	-1.25	0.40	
7	8	7	7	1180	3.000	2.944	-1.91	3.000	2.943	-1.94	-0.	
7	0	7	7	1590	2.450	2.471	-0.78	2.500	2.474	-1.06	0.40	
8	8	7	7	2530	2.760	2.752	-0.28	2.770	2.753	-0.63	0.35	
8	8	4	1	1015	2.950	2.914	-1.22	2.960	2.914	-1.59	0.34	
5	8	11	1	890	4.410	4.386	-0.54	4.420	4.400	-0.45	0.23	
10	6	11	1	1880	2.300	2.310	0.43	2.320	2.315	-0.20	0.87	
13	8	11	1	1975	2.330	2.312	-0.77	2.340	2.318	-0.97	0.43	
45	8	5	8	1930	2.270	2.263	-0.30	2.300	2.269	-1.35	1.02	

DEV. AND DIFF. ARE IN PERCENT

THE (-) SIGN INDICATES THAT THE INST IS SLOWER THAN  
PREDICTED OR THAT INST 1 IS SLOWER THAN INST 2.

\*Time up ramp not verified. -10-

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Table 2-2

## VEHICLE 1612 PAYLOAD J-19 SELF HEATING TEST

## SUMMARY OF SELF HEATING CORRECTION CURVES

10	106	40	SS2	03	103	43	208	28	113	23	111
39	TC2	20	110	08	105				55	213	50
21	SSI	37	TC1	52	212				21		
		38	206	46	209						
		36	205	41	207						
		31	203	33	204						
		48	210	05	104						
		25	112	15	108						
		18	109	44	CA2						
		13	107								
TIME	NO. 1	NO. 2	NO. 3	NO. 4	NO. 5	NO. 6					
0.10	0.6	1.4	2.1	2.9	2.4	4.3					
0.13	0.8	1.7	2.6	3.6	3.0	5.4					
0.16	1.0	2.2	3.3	4.5	3.8	6.8					
0.20	1.1	2.3	3.5	4.8	4.2	7.9					
0.25	1.1	2.4	3.7	4.9	4.5	9.1					
0.32	1.2	2.5	3.8	5.0	4.9	10.5					
0.40	1.3	2.6	4.1	5.2	5.4	12.3					
0.50	1.4	2.8	4.3	5.4	6.0	14.6					
0.63	1.5	3.0	4.5	5.6	6.4	15.6					
0.79	1.6	3.2	4.6	5.8	6.9	16.9					
1.00	1.8	3.4	5.1	6.1	7.6	18.6					
1.26	1.9	3.6	5.3	6.4	8.0	19.5					
1.58	2.1	3.9	5.6	6.7	8.6	20.6					
2.00	2.4	4.2	6.0	7.1	9.3	22.1					
2.51	2.6	4.5	6.3	7.5	9.8	23.0					
3.16	2.8	4.8	6.6	7.9	10.4	24.1					
3.98	3.1	5.1	7.0	8.4	11.0	24.9					
5.01	3.4	5.5	7.4	8.9	11.6	25.9					
6.31	3.6	5.8	7.7	9.3	12.0	26.5					
7.94	3.9	6.1	8.2	9.8	12.5	27.2					
10.00	4.2	6.5	8.7	10.4	13.2	26.1					
12.59	4.4	6.8	9.0	10.8	13.4	26.4					
15.85	4.7	7.0	9.4	11.3	13.8	28.8					
19.95	5.0	7.4	10.0	11.9	14.2	29.2					
25.12	5.4	7.8	10.6	12.6	14.7	29.8					
31.62	5.8	8.3	11.3	13.5	15.2	30.3					
39.81	6.0	8.6	11.8	14.0	15.4	30.4					
50.12	6.4	8.9	12.5	14.7	15.6	30.6					
63.10	6.8	9.3	13.2	15.6	15.9	30.7					
79.43	7.1	9.7	13.9	16.3	16.0	30.6					
100.00	7.5	10.1	14.7	17.1	16.1	30.5					

## VEHICLE 1612 PAYLOAD J-19 SELF HEATING TEST

## SUMMARY OF SELF HEATING CORRECTION CURVES

37 TC1A

40 SS2 31 203 15 108 55 213 50 211

37 TCI 41 207 33 204 28 113 23 111

39 TC2 38 206 08 105

21 SSI 36 205 44 CA2

39 TCZA 46 209

20 110

48 210

52 212

03 103

18 109

25 112

13 107

05 104

10 106

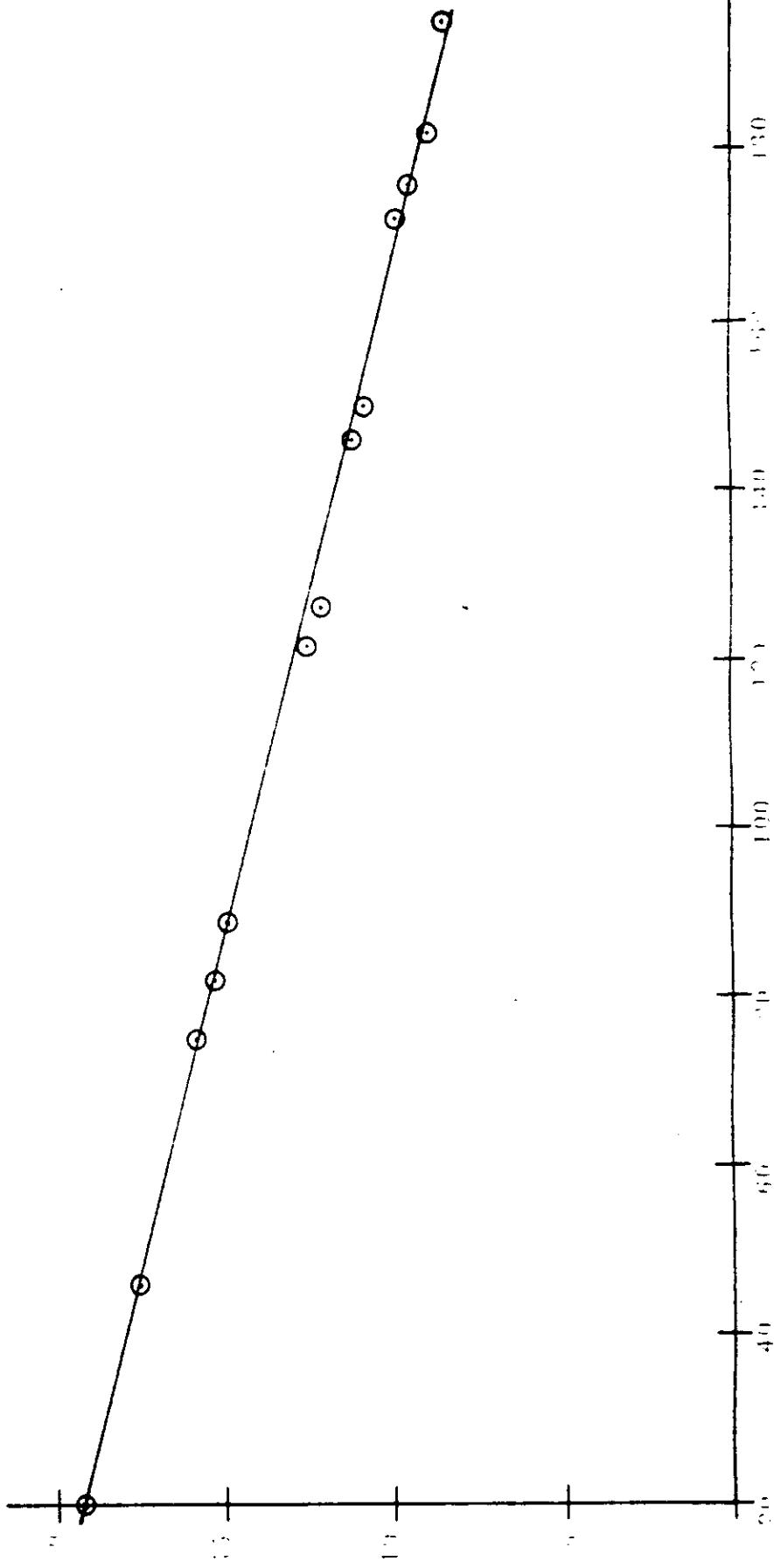
43 208

TIME	NO. 1	NO. 2	NO. 3	NO. 4	NO. 5
0.10	0.1	0.4	0.7	1.0	3.3
0.13	0.1	0.6	0.8	1.2	4.2
0.16	0.2	0.7	1.1	1.5	5.3
0.20	0.2	0.8	1.2	1.8	6.2
0.25	0.3	1.0	1.4	2.1	7.2
0.32	0.4	1.1	1.6	2.4	8.5
0.40	0.5	1.4	1.9	2.8	10.1
0.50	0.7	1.6	2.3	3.3	12.1
0.63	0.8	1.8	2.5	3.8	13.1
0.79	1.0	2.1	2.9	4.4	14.3
1.00	1.2	2.4	3.3	5.1	15.9
1.26	1.4	2.7	3.6	5.6	16.8
1.56	1.6	3.0	4.1	6.3	17.9
2.00	1.8	3.4	4.6	7.1	19.4
2.51	2.0	3.8	5.0	7.8	20.3
3.16	2.3	4.2	5.5	8.5	21.4
3.98	2.5	4.5	6.0	9.2	22.3
5.01	2.8	5.0	6.5	10.0	23.4
6.31	3.0	5.4	7.0	10.6	24.1
7.94	3.3	5.9	7.5	11.3	24.9
10.00	3.6	6.5	8.2	12.1	26.0
12.59	3.8	6.9	8.5	12.5	26.4
15.85	4.0	7.3	9.0	13.0	26.9
19.95	4.3	7.9	9.6	13.6	27.5
25.12	4.7	8.6	10.3	14.3	28.2
31.62	5.1	9.4	11.1	15.0	29.0
39.81	5.3	10.0	11.6	15.4	29.3
50.12	5.6	10.7	12.2	15.8	29.7
63.10	6.0	11.5	12.9	16.2	30.1
79.43	6.3	12.2	13.5	16.4	30.3
100.00	6.7	13.2	14.3	16.7	30.5

Table 2-4



Graph showing  
the relationship  
between  
operating time  
and temperature.



OPERATE TIME - MINUTES

TEMPERATURE

Figure 2-1

• PULSED DYNAMIC RANGE TEST

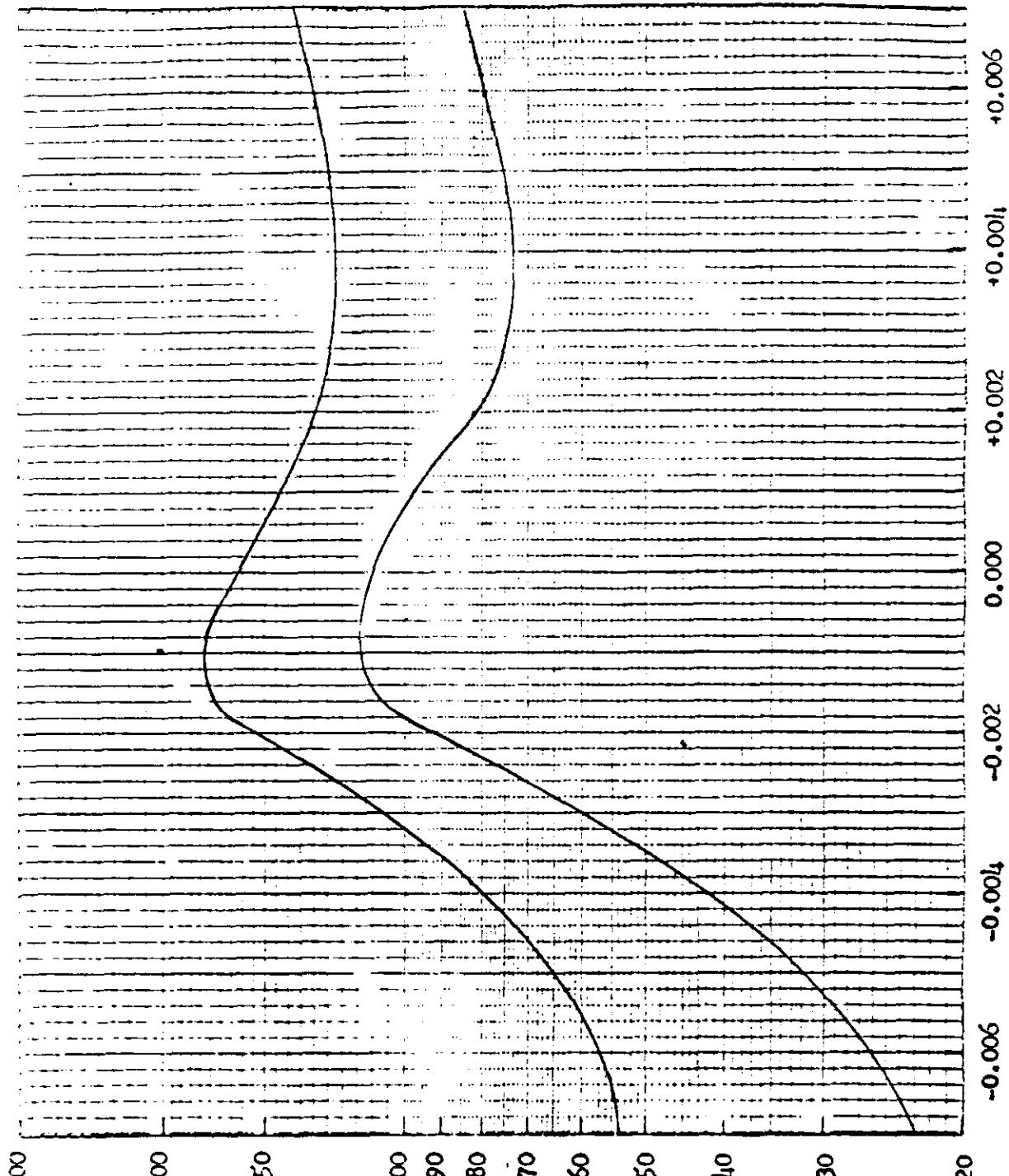


Figure 2-2

**PRE-FLIGHT DYNAMIC RESOLUTION**

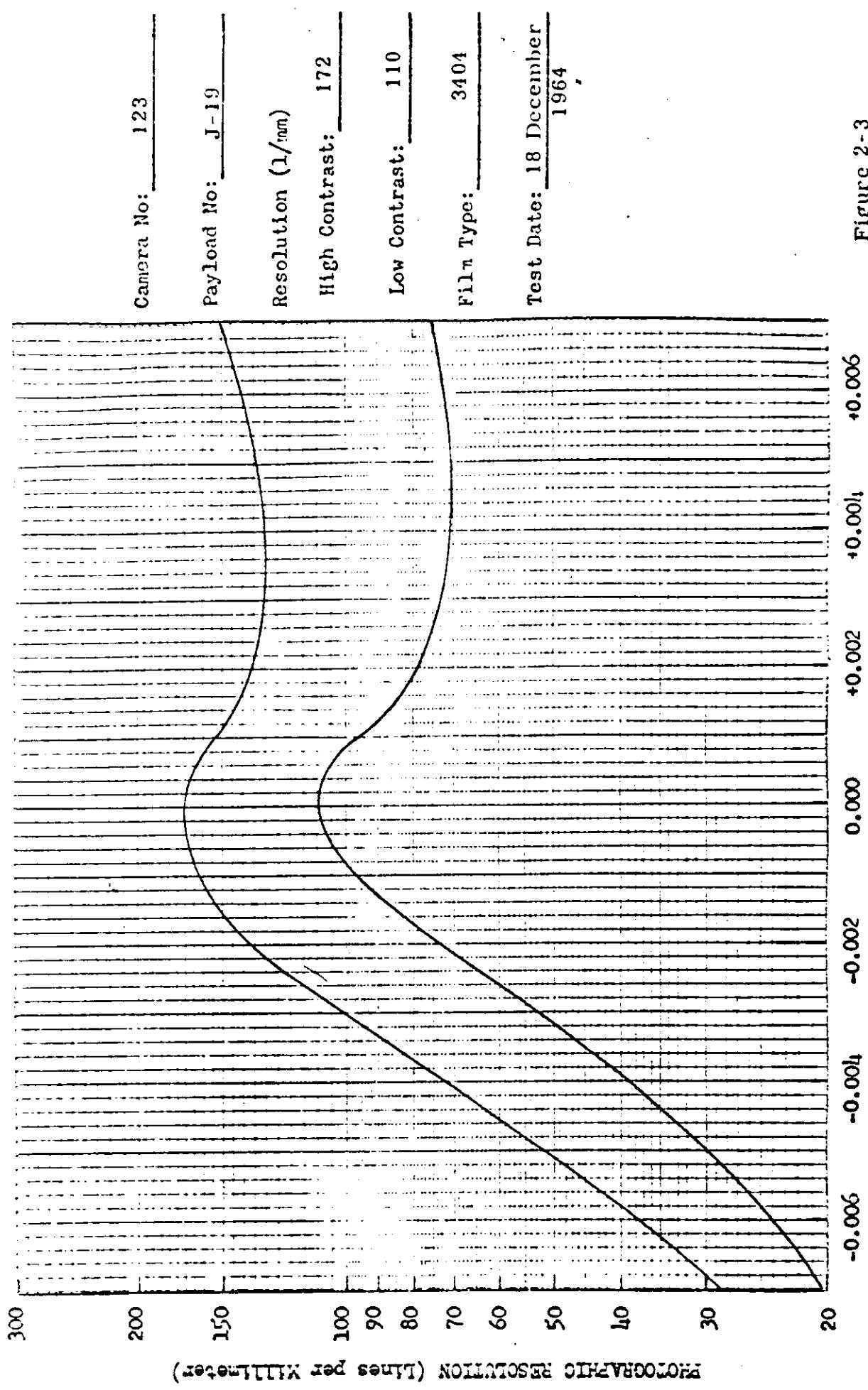


Figure 2-3

THROUGH FOCUS INCREMENTS (Inches)

TOT SPEC HLT

## SECTION 3

## FLIGHT OPERATIONS

## A. INSTRUMENTATION AND COMMAND PERFORMANCE

The telemetry instrumentation system was satisfactory throughout the flight. The only exceptions being the lens rotation monitor (Ch 10, Lk I, cont.) and the cycle counter units monitor (Ch 13, Lk I, Pt 27) for Instrument #2.

The Instrument #2 lens rotation monitor pulse randomly changed length and at times was missing completely. The trailing edge of the pulse when present was always in the proper position.

The units cycle count for Instrument #2 read a maximum of 3.7V (9 count) on and after orbit 32. Step levels were proper on orbits 16 and 24. Analysis of the cycle counter circuit shown on Figure 3-1 indicates that the most probable cause of failure is a high resistance short from either the wiper or high side of counter resistor divider network to TLM ground. The resistance of the short would be approximately 6K from data observed on orbit 32 and as low as 1K from data observed on orbit 73.

Good correlation was obtained for the cycle counters and film footage potentiometers during the first mission. The potentiometers agreed within plus or minus 50 cycles which is the minimum resolution obtainable from reading the TLM records. During the second mission the readings of both potentiometers were off from the corresponding cycle counters by values up to 150 cycles.

## B. PANORAMIC CAMERA PERFORMANCE

Engineering operations were programmed for acquisition over [REDACTED] Tracking Station during Orbits 8, 16, 24, 32, 48 and 65 of Mission 1018-1, and 3, 81 and 97 of Mission 1018-2.

Instrument operation and payload metering was satisfactory throughout both Missions 1018-1 and 1018-2 as indicated on TLM by the center-of-format, lens rotation, and payload supply and take-up monitors. Start-up and shut-down was normal.

Cycle period data for these operations are contained in Table 3-1. The instruments ran consistently slower than predicted throughout the flight. The predictability error was approximately -1.5% for Instrument #1 and -.75% for Instrument #2. The difference in operating rates between the instruments was generally within  $\pm 1\%$ .

Film consumption for the flight was as follows:

MISSION 1018-1

	<u>Master</u>	<u>Slave</u>
Cycles	2856	2804
Feet	8023*	7887*

MISSION 1018-2

Cycles	2956	2964
Feet	7786	7832

\* Includes pre-flight test film footage.

The general instrument response, based upon nominal vs. actual consumption, was approximately 3% slow during the first mission, and 3% decaying to 5% slow during the second mission. The slave camera was slightly faster ( $< 1\%$ ) than the master.

FMC match was within 5% of rates required after the early revs when orbit parameters and camera response were being determined.

<u>Orbit</u>	<u>R-A-D</u>	<u>Approx. Match</u>	<u>Remarks</u>
1 - 5	4-7-7	5 % slow	Settings for nominal orbit and response
6 - 18	4-7-6	4% slow	Change on rev 6 to move perigee northwards; valid settings except for camera response.

<u>Orbit</u>	<u>R-A-D</u>	<u>Approx. Match</u>	<u>Remarks</u>
19 - 66	3-7-6	1% mean	Increase rates for camera response; good actual match for first mission, but errors increasing as orbit decays.
67 - 83	2-8-6	2 1/2% mean	Increase rates to compensate orbit decay; errors increase as orbit decays.
84 - 99	2-7-5	1% mean	(D-5 effective on rev 85) increase rates to compensate slower cameras and to adjust limiter action to desired latitude range.

### C. STELLAR-INDEX PERFORMANCE

The Stellar-Index camera for the first mission operated normally for eight and one-half orbits and metered 55 frames. During the third operation of orbit eight the S/I command programmer failed. This failure resulted in loss of all commands to the S/I unit and the loss of the smear pulse on the main panoramic camera film. With the failure of the programmer both the first and second mission S/I units were rendered inoperative. Both units slew during recovery indicating that both units were functional.

### D. CLOCK PERFORMANCE

Satisfactory clock correlation was obtained for both missions. Clock/System time correlation data is contained in Table 3-2. The system time is fitted to a best fit curve of clock versus system time which is represented by a second order equation in this case. The table includes the amounts that the recorded values of system time deviate from the corrected values.

The corrected system time values are computed from the equation -  
 $y = a_2x^2 + a_1x + a_0$ , where  $y$  = corrected system time and  $x$  = clock time.

## E. PRESSURE MAKE-UP SYSTEM PERFORMANCE

The supply pressure history for the PMU system is shown on Figure 3-2, where the supply pressure decay is plotted as a function of the total instrument operate time. The irregular slope of the curve indicates that gas is passing through the PMU system at times other than when the instruments are operating.

The average consumption rate was 7.0 psi/min. during flight. The value obtained during the environmental test was 6.25 psi/min. with no leakage being indicated.

## F. TEMPERATURE ENVIRONMENT

The temperature data obtained on TLM during the [REDACTED] acquisitions are summarized on Table 3-3.

Average instrument temperatures varied only 5° F on both instruments due to an almost constant  $\beta$  angle for the entire flight. The instruments averaged 79° F for the master and 69° F for the slave.

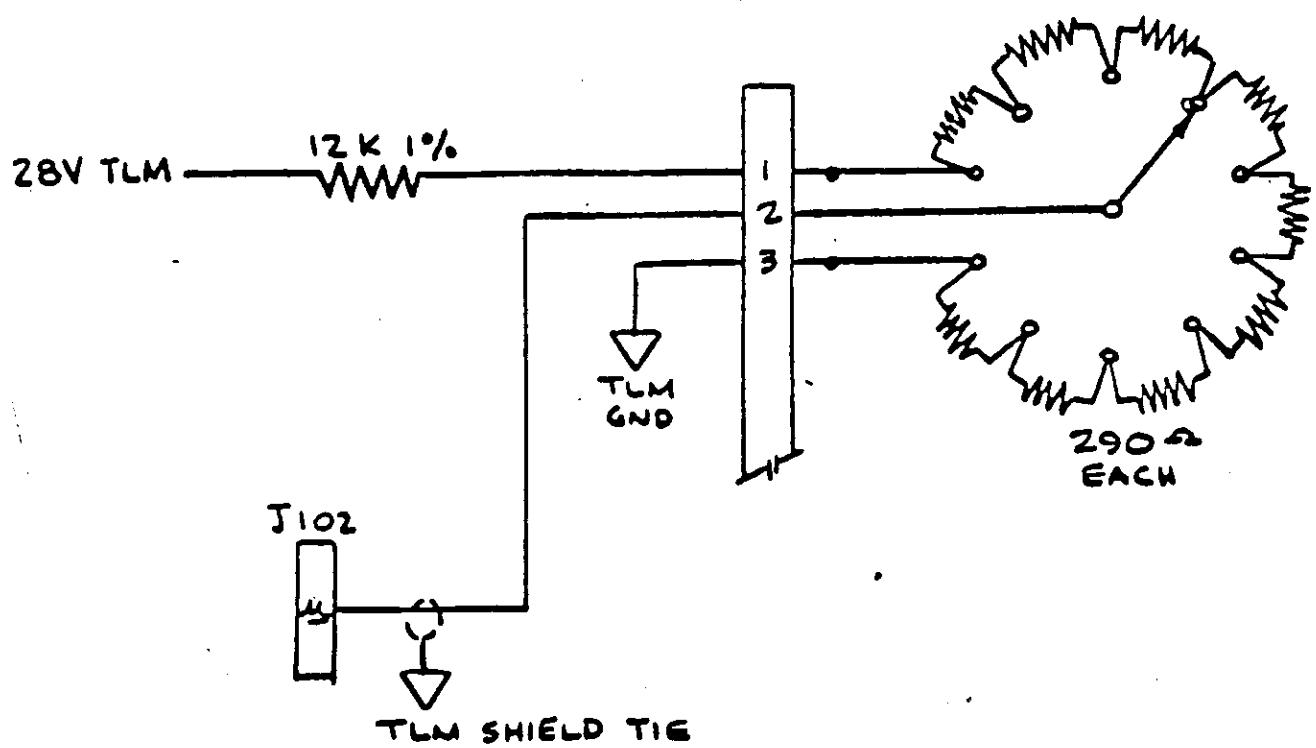
## G. RECOVERY SYSTEM PERFORMANCE

The -1 and -2 SRV units were successfully air recovered on orbit 66, 3/29/65 and on orbit 99, 3/31/65, respectively. Impact points were within tolerance for both recoveries.

## H. SINE FUNCTION GENERATOR PERFORMANCE

This orbital sine function generator (S/N 269) was the fourth unit to be flown with a Corona J system. The unit was first enabled by real time command on orbit 8. The output to the Agena guidance at this time should have been 7.5 milli-volts. The "enable" transient was not detected by the vehicle response. The telemetry monitor indicated the unit was in the correct phase position throughout both missions. Figure 3-3 shows the telemetry voltage vs. the actual elapsed time.

After second recovery the unit was disabled on orbit 112 and then enabled on orbit 113. At the "enable" the output of the unit should have been 21.5 millivolts. Again the vehicle did not detect the "enable" transient and could indicate that the output of the unit was not being received by the vehicle guidance system.



INST. #2 UNITS CYCLE COUNTER TLM CIRCUIT

## J-19/1612 FLIGHT CYCLE RATE SUMMARY - MISSION 1010

REV/MODE	RAMP	T.U.R.	INST 122			INST 123			122/123	
			ACT.	NOM.	DEV.	ACT.	NOM.	DEV.	DIFF.	
008 A	4 7	182	3.112	3.107	-0.15	3.120	3.106	-0.44	0.26	
016 A	4 7	1740	2.275	2.215	-2.73	2.255	2.222	-1.50	-0.88	
024 A	3 7	224	2.973	2.961	-0.41	2.974	2.960	-0.48	0.03	
032 A	3 7	1830	2.230	2.206	-1.08	2.230	2.213	-0.77	-0.	
048 A	3 7	1800	2.237	2.206	-1.39	2.221	2.213	-0.34	-0.72	
065 A	3 7	1845	2.238	2.206	-1.45	2.229	2.213	-0.73	-0.40	
073 D	2 8	355	2.700	2.699	-0.03	2.700	2.700	-0.00	-0.	
081 B	2 8	1940	2.245	2.207	-1.70	2.222	2.214	-0.34	-1.02	
097 E	2 7	2056	2.193	2.201	0.35	2.225	2.207	-0.79	1.46	

DEV. AND DIFF. ARE IN PERCENT

THE (-) SIGN INDICATES THAT THE INST IS SLOWER THAN  
PREDICTED OR THAT INST 1 IS SLOWER THAN INST 2

## PRELIMINARY CLOCK CORRELATION

ORDER FIT 1

SYS TIME I/P	CL TIME I/P	COMP SYS TM	DELTA ST	REV STA
56632.268	395404.54190	56632.27520	-0.00628	0 1
31321.622	456493.03690	31321.61980	0.00314	8 1
75649.309	500821.57590	75649.30850	0.00145	16 1
30360.145	5061.49890	30360.14320	0.00273	24 1
74671.579	49372.93390	74671.57790	0.00203	32 1
29276.552	90377.40990	29276.55360	-0.00069	40 1
73568.432	134669.79290	73568.43630	-0.00338	48 1
28333.667	175835.02690	28333.67010	-0.00210	56 1
77875.992	225377.34590	77875.98870	0.00424	65 1
32594.385	266495.73990	32594.38240	0.00352	73 1
76841.137	310742.49690	76841.13910	-0.00117	81 1
31482.427	351783.78690	31482.42880	-0.00089	89 1
75704.865	396006.22690	75704.86850	-0.00259	97 1

A0=-0.33877226400 06 A1= 0.9999999931370 00

SIGMA=0.00291 NO. POINTS= 13

RATIO OF CLOCK TIME TO SYS TIME= 0.100000006860 01

ORDER FIT 2

SYS TIME I/P	CL TIME I/P	COMP SYS TM	DELTA ST	REV STA
56632.268	395404.54190	56632.27280	-0.00389	0 1
31321.622	456493.08690	31321.61890	0.00404	8 1
75649.309	500821.57590	75649.30850	0.00150	16 1
30360.145	5061.49890	30360.14380	0.00217	24 1
74671.579	49372.93390	74671.57890	0.00101	32 1
29276.552	90377.90990	29276.55490	-0.00196	40 1
73568.432	134669.79290	73568.43770	-0.00474	48 1
28333.667	175835.02690	28333.67130	-0.00336	56 1
77875.992	225377.34590	77875.98960	0.00332	65 1
32594.385	266495.73990	32594.38290	0.00308	73 1
76841.137	310742.49690	76841.13890	-0.00092	81 1
31482.427	351783.78690	31482.42780	0.00019	89 1
75704.865	396006.22690	75704.86640	-0.00043	97 1

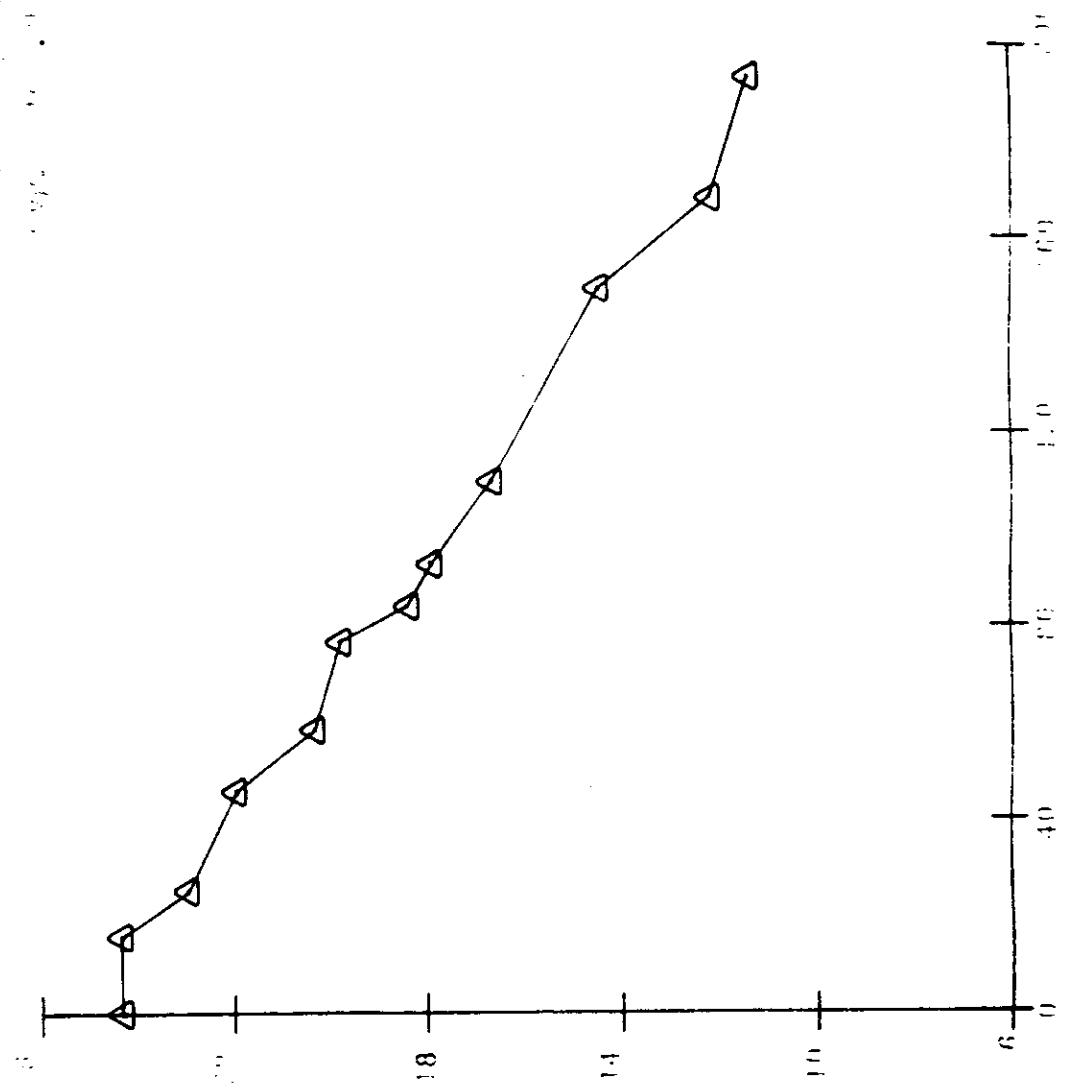
A0=-0.33877228510 06 A1= 0.1000000060410 01

A2=-0.50317405674850-13

SIGMA=0.00265 NO. POINTS= 13

FIGURE 1.22  
RESULTS OF A TEST

Figure 1.22



TEST NUMBER 1000

1000

TABLE 3-3  
J-19 TEMPERATURE SUMMARY

<u>SENSOR</u>		<u>ORBITS ACQUIRED</u>													
<u>Master Camera</u>		65 73 81 89 97 67 73 72 73 80 71 73 81 65													
AVG		72 79 86 89 96 77 84 87 90 93 81 84 86 76													
<u>Slave Camera</u>		70 78 86 89 96 77 84 87 90 93 81 84 86 76													
AVG		69 67 65 63 61 60 65 62 63 60 64 66 65 60													
<u>Supply Spool</u>		64 66 53 59 55 61 55 60 57 61 60 63 64 61													
Spool		55 58 59 59 55 61 55 60 57 61 60 63 64 61 59													

NOTE: All data corrected for self-heating, except injection.

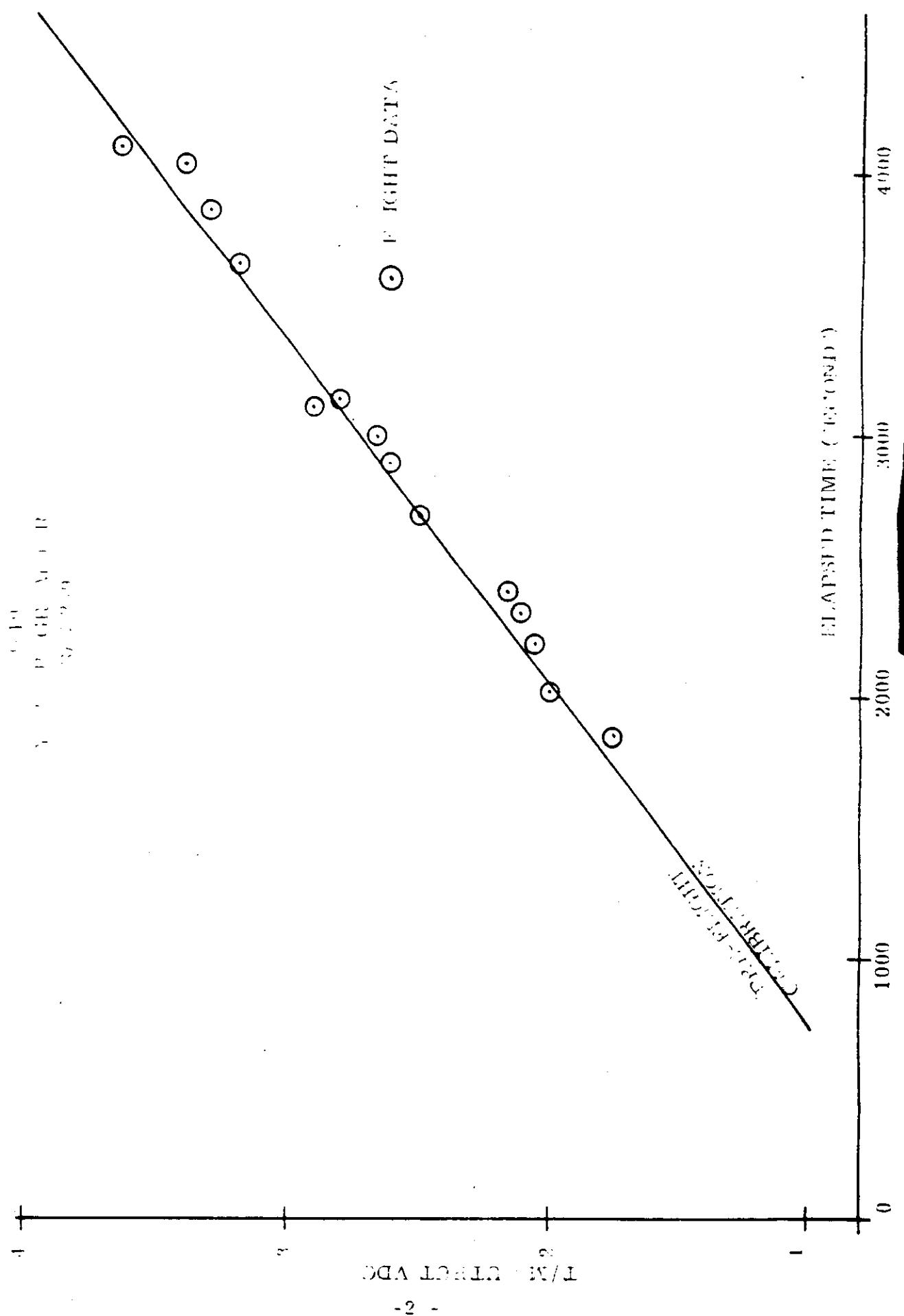
TABLE 3-3  
J-19 TEMPERATURE SUMMARY

SENSOR														
<u>Fair ("A")</u>		0	8	16	24	32	40	48	56	64	72	80	89	97
<u>Barrel #1 ("E")</u>		CBH	37	59	43	55	43	40	43	55	33	53	--	--
1		CBH	11	13	17	10	16	2	17	10	67	74	--	--
2		215	2	20	7	20	7	12	10	20	77	131	--	--
3		248	53	68	60	71	58	62	62	68	44	83	--	--
4		CBH	52	57	57	61	57	52	57	61	46	63	--	--
5		CBH	52	61	57	57	57	48	57	61	--	--	--	--
<u>Barrel #2</u>														
1		162	39	51	38	51	38	42	41	51	31	47	31	33
2		162	36	77	45	77	45	66	48	77	45	81	45	69
3		225	53	117	62	114	62	101	65	114	62	117	58	109
4		205	56	74	61	74	64	62	67	71	61	70	61	59
5		178	48	73	56	72	56	67	56	72	47	63	50	54
<u>Conic Adapter</u>														
1		187	37	43	46	42	46	34	46	46	35	35	35	27
<u>Clock</u>														
1		94	64	67	67	67	67	62	70	67	63	59	63	53
2		96	64	67	67	67	70	64	70	70	63	59	63	53
<u>Thrust Cone "A" to "B" SRV</u>														
1		107	32	35	33	31	33	34	28	30	57	57	56	56
2		72	52	44	43	42	41	43	43	43	63	63	62	63
<u>Stellar/Index "A" to "B"</u>														
1		90	56	65	61	64	61	56	61	65	65	62	62	57
2		78	56	62	59	58	59	53	62	59	75	65	68	57
<u>Recovery Battery "B" SRV</u>														
1		70	69	68	66	66	66	64	66	68	87	89	85	86
<u>Master Cassette "A" SRV</u>														
2		92	51	52	45	44	44	48	45	51	--	--	--	--

NOTE: Only Thrust Cone Data corrected for Self-heating.

CBH - Out of Range High

Figure 3-3



## SECTION 4

## MISSION 1018-1 RECOVERY SYSTEM

SRV #668 was received at A/P on 30 July 1964. The receiving weight was 149.7 pounds. After modifications and incorporation of outstanding E.O.'s, the SRV was delivered to Systems Test for incorporation into the J-19 system.

The capsule was shipped to VAFB on 13 January 1965.

The -1 recovery system was successfully recovered by air catch from orbit 66 at 15:38 PST on 29 March 1965. The impact point was as follows:

Predicted Impact             $22^{\circ}03.2'N/151^{\circ}12.4'W$

Actual Impact             $21^{\circ}58'N/151^{\circ}15'W$

The condition of the recovered capsule was satisfactory with no damage other than normal paint blistering due to the re-entry environment.

Event times are shown in Table 4-1.

~~TOP SECRET~~

## MISSION 1018-1

## RECOVERY SEQUENCE OF EVENTS

<u>Event</u>	<u>Delta Time (Seconds)</u>	
	<u>Actual</u>	<u>Nominal</u>
* Arm	74.66	$75.0 \pm 1.0$
* Transfer	2.02	$2.0 \pm 0.25$
Electrical Disconnect	0.92	$0.900 \begin{matrix} +0.430 \\ -0.400 \end{matrix}$
Separation	---	---
** Spin	3.43	$3.4 \pm 0.30$
Retro	7.57	$7.55 \pm 0.45$
Despin	10.74	$10.75 \pm 0.54$
T C Separation	1.49	$1.5 \pm 0.15$
*** "G" Switch Open	467.85	$515.0 \begin{matrix} +65.0 \\ -50.0 \end{matrix}$
Parachute Cover Off	33.98	$34.0 \pm 1.5$
Drogue Chute Deployed	0.74	$0.63 \pm 0.08$
Main Chute Bag Separate	9.91	$10.14 \begin{matrix} +0.48 \\ -0.40 \end{matrix}$
Main Chute Deployed	0.54	$0.52 \pm 0.13$
Main Chute Disreef	4.67	$4.46 \begin{matrix} +0.49 \\ -0.29 \end{matrix}$

\* From Separation

\*\* From Electrical Disconnect

\*\*\* From Retro

Spin Rate 66 RPM

Despin Rate 9.1 RPM

Retro Velocity 1024 ft. /sec.

TABLE 4-1

~~TOP SECRET~~

## SECTION 5

## MISSION 1018-2 RECOVERY SYSTEM

SRV #669 was received at A/P on 30 July 1964. The receiving weight was 151.9 pounds. After modifications and incorporation of outstanding E.O.'s the unit was delivered to Systems Test for mating to the J-19 system.

The capsule was shipped to VAFB on 13 January 1965.

The -2 recovery system was successfully recovered by air catch from orbit 99 at 16:29 PST on 31 March 1965. The impact point was as follows:

Predicted Impact	$20^{\circ}01.1'N/164^{\circ}34'W$
Actual Impact	$20^{\circ}00'N/164^{\circ}24'W$

Event times are shown in Table 5-1.

The condition of the recovered capsule indicated no abnormal re-entry effects.

## MISSION 1018-2

## RECOVERY SEQUENCE OF EVENTS

<u>Event</u>	<u>Delta Time (Seconds)</u>	
	<u>Actual</u>	<u>Nominal</u>
Arm	74.94	75.0 <u>±</u> 1.0
Transfer	2.03	2.0 <u>±</u> 0.25
Electrical Disconnect	0.93	0.900 <sup>+0.430</sup> <sub>-0.400</sub>
Separation	--	--
Spin	3.37	3.4 <u>±</u> 0.30
Retro	7.83	7.55 <u>±</u> 0.45
Despin	10.74	10.75 <u>±</u> 0.54
T/C Separation	1.51	1.5 <u>±</u> 0.15
"G" Switch Open	468.82	515.0 <sup>+65.0</sup> <sub>-50.0</sub>
Parachute Cover Off	33.30	34.0 <u>±</u> 1.5
Drogue Chute Deployed	0.69	0.63 <u>±</u> 0.08
Main Chute Bag Separate	9.66	10.14 <sup>+0.48</sup> <sub>-0.40</sub>
Main Chute Deployed	0.60	0.52 <u>±</u> 0.13
Main Chute Disreef	4.00	4.46 <sup>+0.49</sup> <sub>-0.29</sub>

\* From Separation

\*\* From Electrical Disconnect

\*\*\* From Retro

Spin Rate 66 RPM

Despin Rate 10 RPM

Retro Velocity 1024 ft. / sec.

TABLE 5-1

## SECTION 6

## MASTER (FWD) PANORAMIC CAMERA

## A. COMPONENT ASSIGNMENT

Component	Serial Number
Main Camera	122
Main Camera Lens	0652435
Supply Horizon Camera	149B
Supply Horizon Camera Lens	813523
Take-up Horizon Camera	163A
Take-up Horizon Camera Lens	814011
Supply Cassette	SC-27

## B. CAMERA DATA AND FLIGHT SETTINGS

## Main Camera:

Lens	24" f/3.5
Slit Width	0.250"
Filter Type	Wratten 25
Film Type	Eastman Type 3404

## Supply (Port) Horizon Camera:

Lens	55 mm f/6.8
Aperture Setting	f/6.8
Exposure Time	1/100 second
Filter Type	Wratten 25

## Take-up (Starboard) Horizon Camera:

Lens	55 mm f/6.8
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

## C. POST FLIGHT PERFORMANCE EVALUATION

This camera produced 2857 frames (8023 feet, including preflight test footage) of photography during Mission 1018-1, and 2956 frames (7786 feet) during Mission 1018-2. The overall photographic quality was judged to be about the same for both missions. Compared to past missions, the present ones are considered better than 1014, but not quite as good as 1015, 1016, and 1017.

Image contrast of the master camera was about the same as that of the slave although the filter that was used provided higher wavelength cutoff. Customer personnel using the material expressed no preference between master and slave results in terms of either information content or image contrast. The MIP frames selected for both missions were however produced by the master camera.

The only probable explanation for the slightly less than optimum image quality appears to be unfavorable atmospheric conditions over the areas of interest. The basis for this explanation, which is far from conclusive, is: (1) weather conditions observed on the small number of index images obtained at the beginning of the mission, and (2) the exposure and preliminary processing data tends to suggest atmospheric reflection of solar illumination.

The master panoramic camera and its auxiliary equipment operated in a generally satisfactory manner throughout both missions. The horizon cameras functioned normally and provided imagery of good quality. Lights for the time word, serial number, and index marks, and fiducial marks functioned normally throughout. The 200 cps timing marks appeared normally except that the density was less than desirable.

Emulsion scratches occurred at each edge of the format under the camera number on all frames of both missions. Similar scratches have

been observed on past missions, however the scratches generally are not detected on preflight test materials. These scratches match the ends of the scan rollers. There were also many very small longitudinal scratches in the region within one inch of the timing mark edge throughout both missions. Such marks have been observed occasionally on past missions, but in much smaller regions of the format. Similar scratches appeared on the slave photography and are described in the following section.

Minor light leak fogging during instrument shut down periods is observed at three places on the first or last frames of most passes. The approximate locations of the leaks have been identified and are discussed further in the following section on the slave camera.

## SECTION 7

## SLAVE (AFT) PANORAMIC CAMERA

## A. COMPONENT ASSIGNMENT

Component	Serial
Main Camera	123
Main Camera Lens	0632435
Supply Horizon Camera	129B
Supply Horizon Camera Lens	812270
Take-up Horizon Camera	130A
Take-up Horizon Camera Lens	812277
Supply Cassette	SC-27

## B. CAMERA DATA AND FLIGHT SETTINGS

## Main Camera:

Lens	24" f/3.5
Slit Width	0.175"
Filter Type	Wratten 21
Film Type	Eastman Type 3404

## Supply (Starboard) Horizon Camera:

Lens	55mm f/6.8
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

## Take-up (Port) Horizon Camera:

Lens	55mm f/6.8
Aperture Setting	f/6.8
Exposure Time	1/100 second
Filter Type	Wratten 25

### C. POST FLIGHT PERFORMANCE EVALUATION

This camera produced 2804 frames (7887 feet, including preflight test footage) photography during Mission 1018-1, and 2964 frames (7832 feet) of photography during Mission 1018-2. The overall photographic quality was judged to be about the same as that produced by the master instrument.

Operation of the slave panoramic camera system was normal except for a low intensity mark and fluctuations in the 200 cps timing marks. The horizon cameras functioned normally throughout both missions, and lights for the time word, serial number, and fiducial marks were all normal. The low intensity index mark had been observed in preflight test but could not be corrected by field adjustments. The timing mark fluctuations, previously reported as intermittent operation, were observed only in flight material. In many places where the timing mark appeared to be missing, careful study showed that the marks were very faintly imaged. The cause of the fluctuating timing light intensity is not known.

Emulsion scratches occurred at each edge of the format under the camera number on all frames. Similar scratches occurred with the master camera. Also the small longitudinal scratches appeared on the slave panoramic material under the serial number and at the take-up end of the format; extending into the horizon image.

Minor light leaks were also observed near the end or start of passes. While the light leak fogging occurs at different places on master and slave material with respect to the end of a pass, common light sources are identified by comparison of film paths. One light leak occurring on the first frame of a pass from both instruments results from a leak at the drum of each instrument. Fogging of frames near the end of either master or slave passes can be traced to light leaks at one of two points: the interface between the master instrument barrel and the recovery barrel, or some point near the No. 1 SRV water seal. The latter of course only affects material during the first mission.

## SECTION 8

## PANORAMIC CAMERA EXPOSURE

The Master camera contained a 0.250 inch slit with a Wratten 25 filter while the Slave camera contained a 0.175 inch slit with a Wratten 21 filter. These conditions place the nominal exposure between the intermediate and full level processing curves, as published by [REDACTED] for their [REDACTED]

The illumination conditions during the mission were relatively broad as the flight was conducted near the autumnal equinox. The frequency distributions of the solar elevations and solar azimuths encountered during the photographic operations are shown in Figures 8-1 to 8-4.

The nominal exposure times are shown as a function of latitude for passes D-08, D-56 and D-89 in Figures 8-5 through 8-7 for the Master instrument, and Figures 8-8 through 8-10 for the Slave. The predicted level of processing for the original negative is based on the in-flight performance estimate and is tabulated below with the processing levels reported by [REDACTED]

<u>Mission</u>	<u>Camera</u>		<u>Primary</u>	<u>Intermediate</u>	<u>Full</u>
1018-1	FWD	Predicted	0	0	100
		Reported	9	63	28
1018-1	AFT	Predicted	0	22	78
		Reported	9	53	38
1018-2	FWD	Predicted	0	0	100
		Reported	18	74	8
1018-2	AFT	Predicted	0	15	85
		Reported	1	50	49

The variation in the predicted and reported processing levels is generally consistent with the data observed from recent missions. The use of greater percentages of full processing has been experienced throughout the Corona program. Further analysis and calculations are in process to attempt to ascertain the optimum exposure-processing conditions.

~~TOP SECRET~~

**SOLAR ELEVATION FREQUENCY DISTRIBUTION**

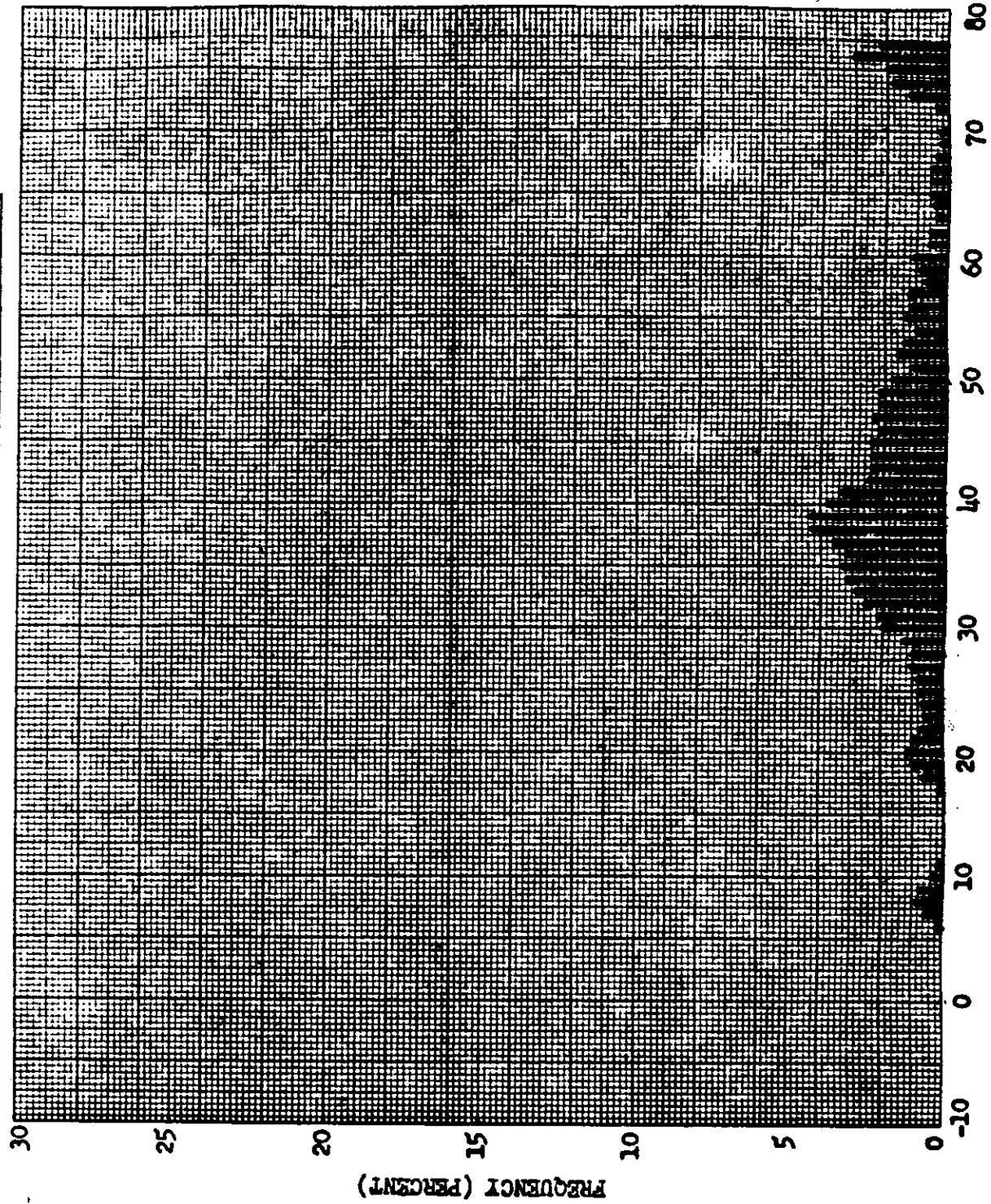


Figure 8-1

TOP SECRET

SOLAR ELEVATION FREQUENCY DISTRIBUTION

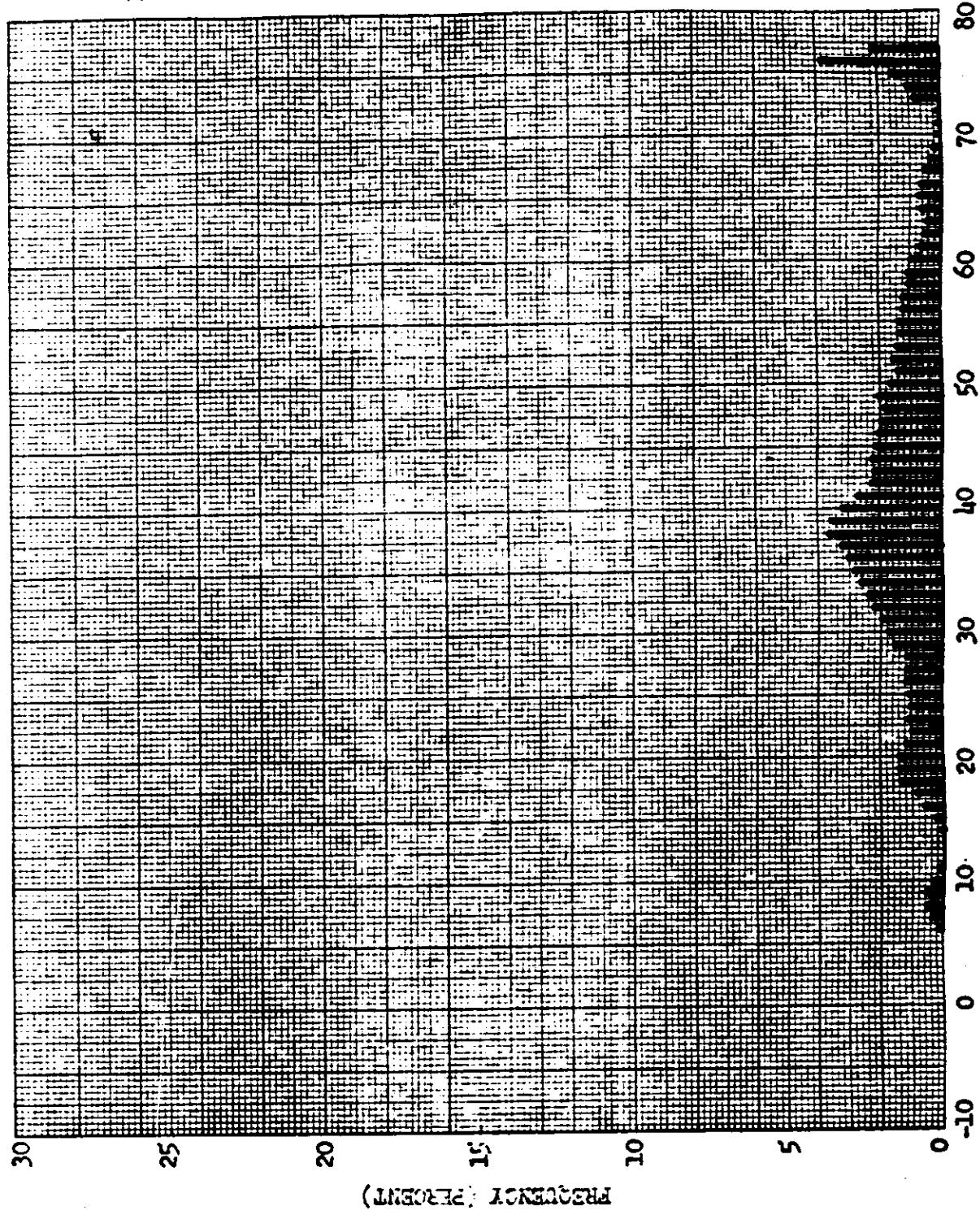


Figure 8-2

TOP SECRET

SOLAR AZIMUTH FREQUENCY DISTRIBUTION

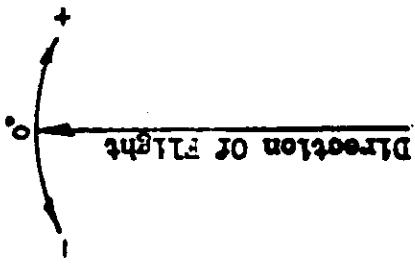
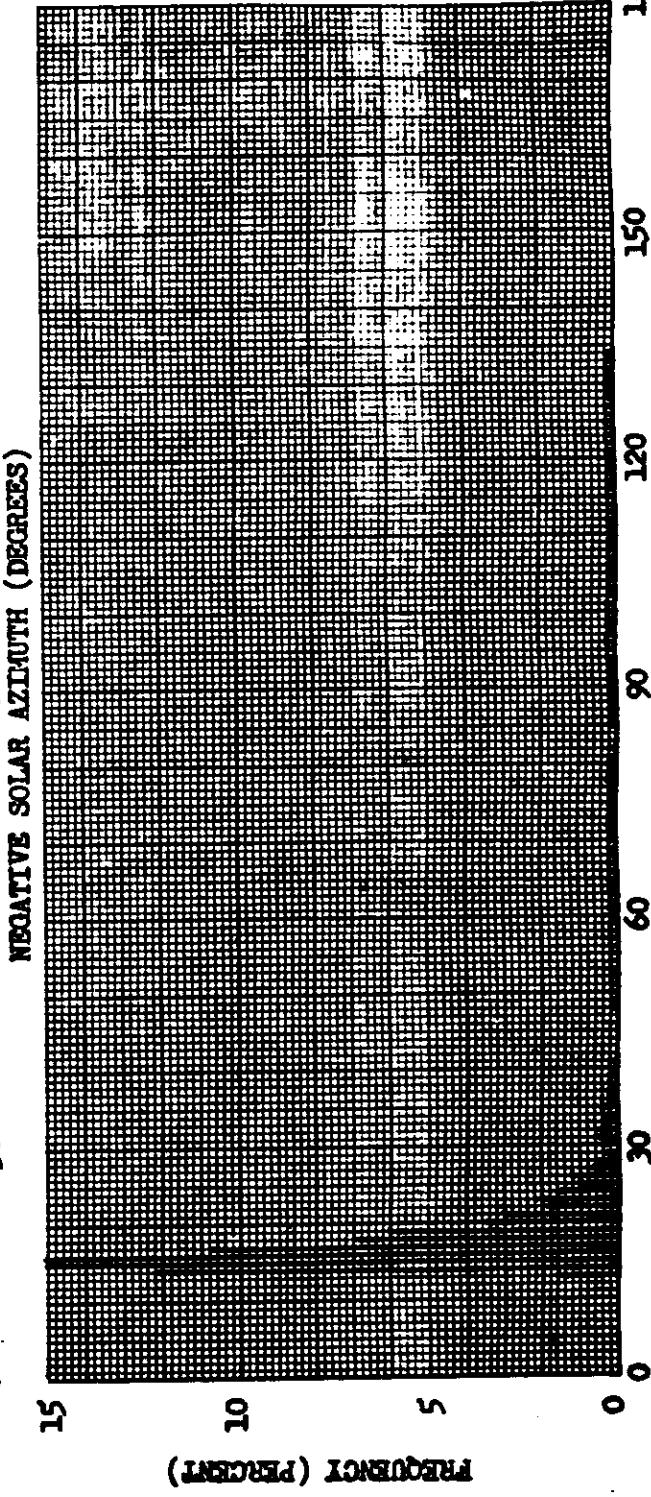
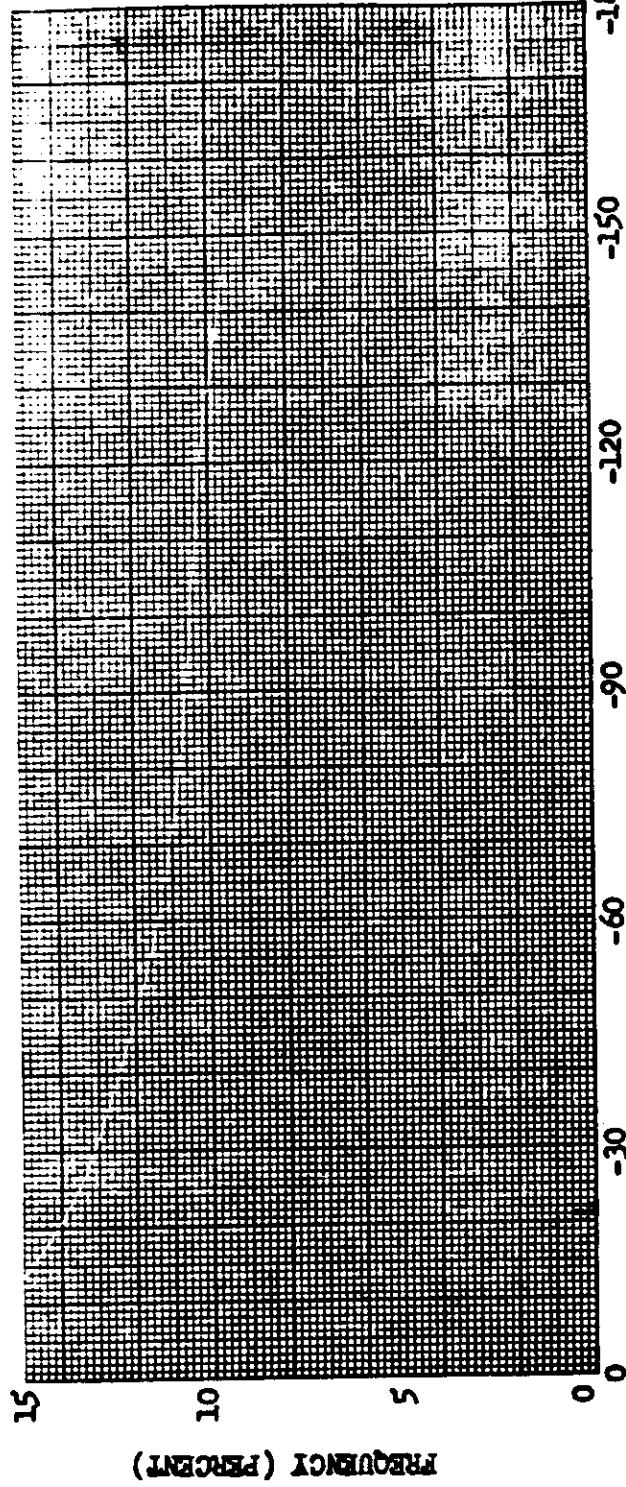
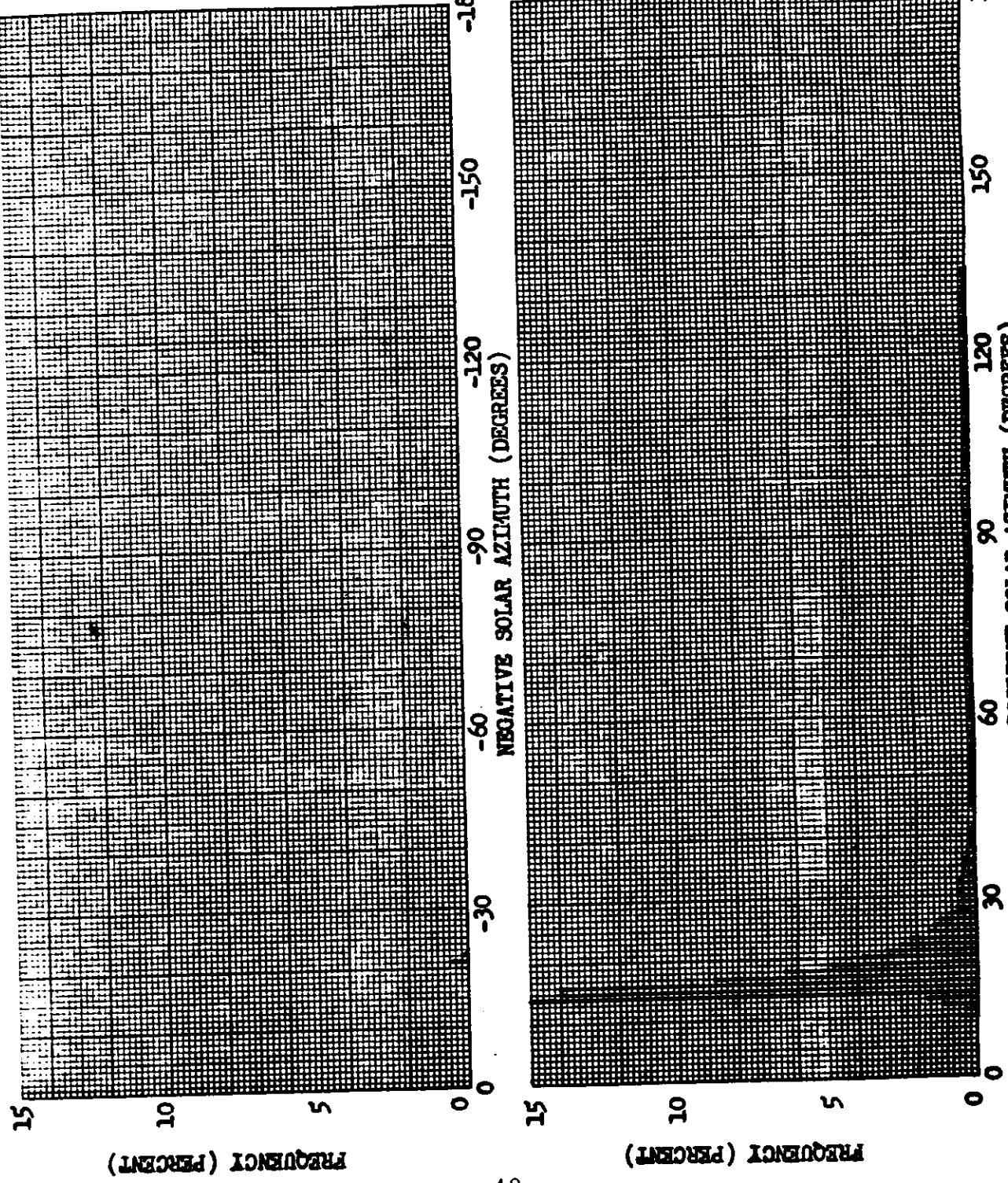


Figure 8-3

~~TOP SECRET~~

SOLAR AZIMUTH FREQUENCY DISTRIBUTION



**EXPOSURE POINTS**

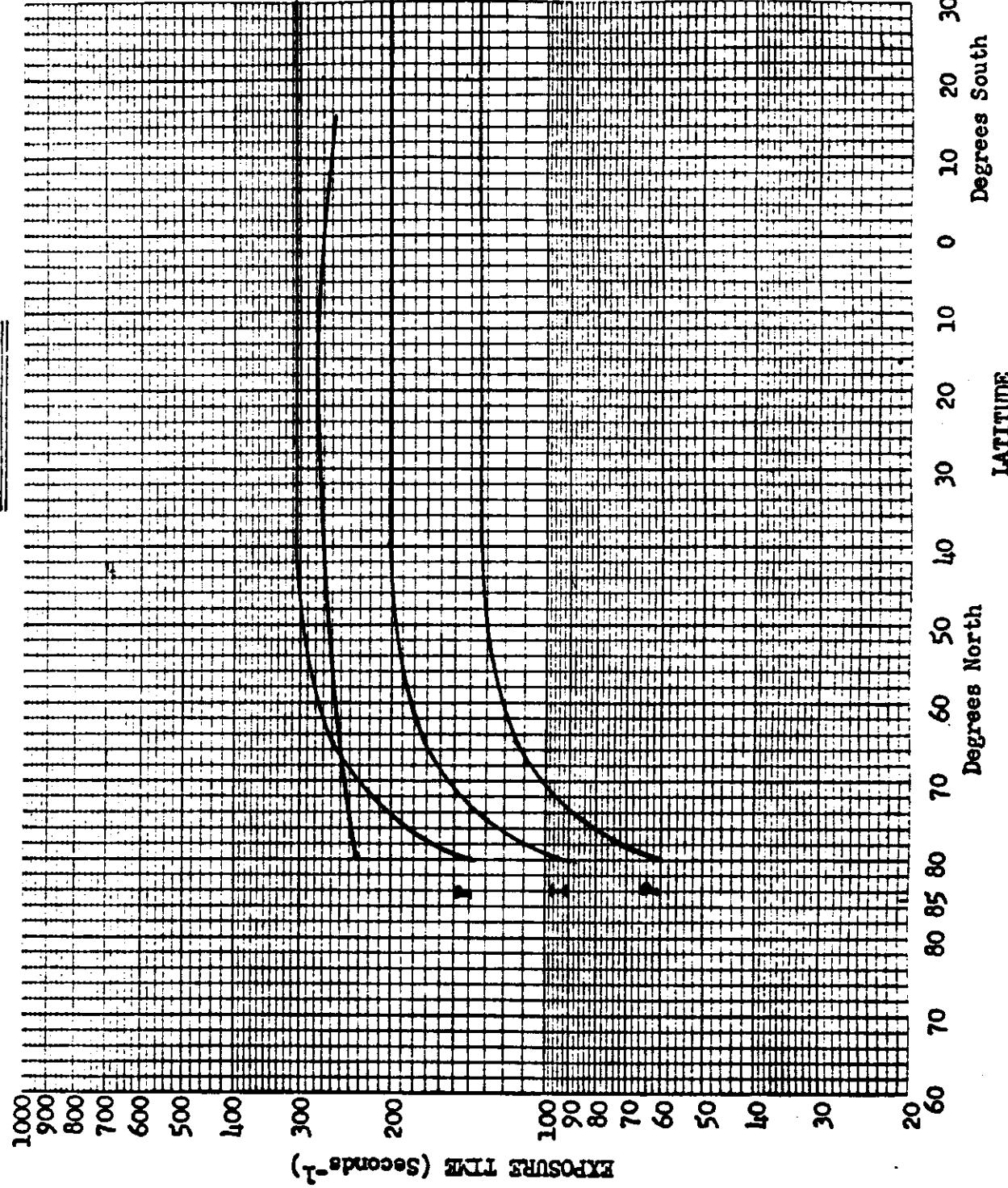
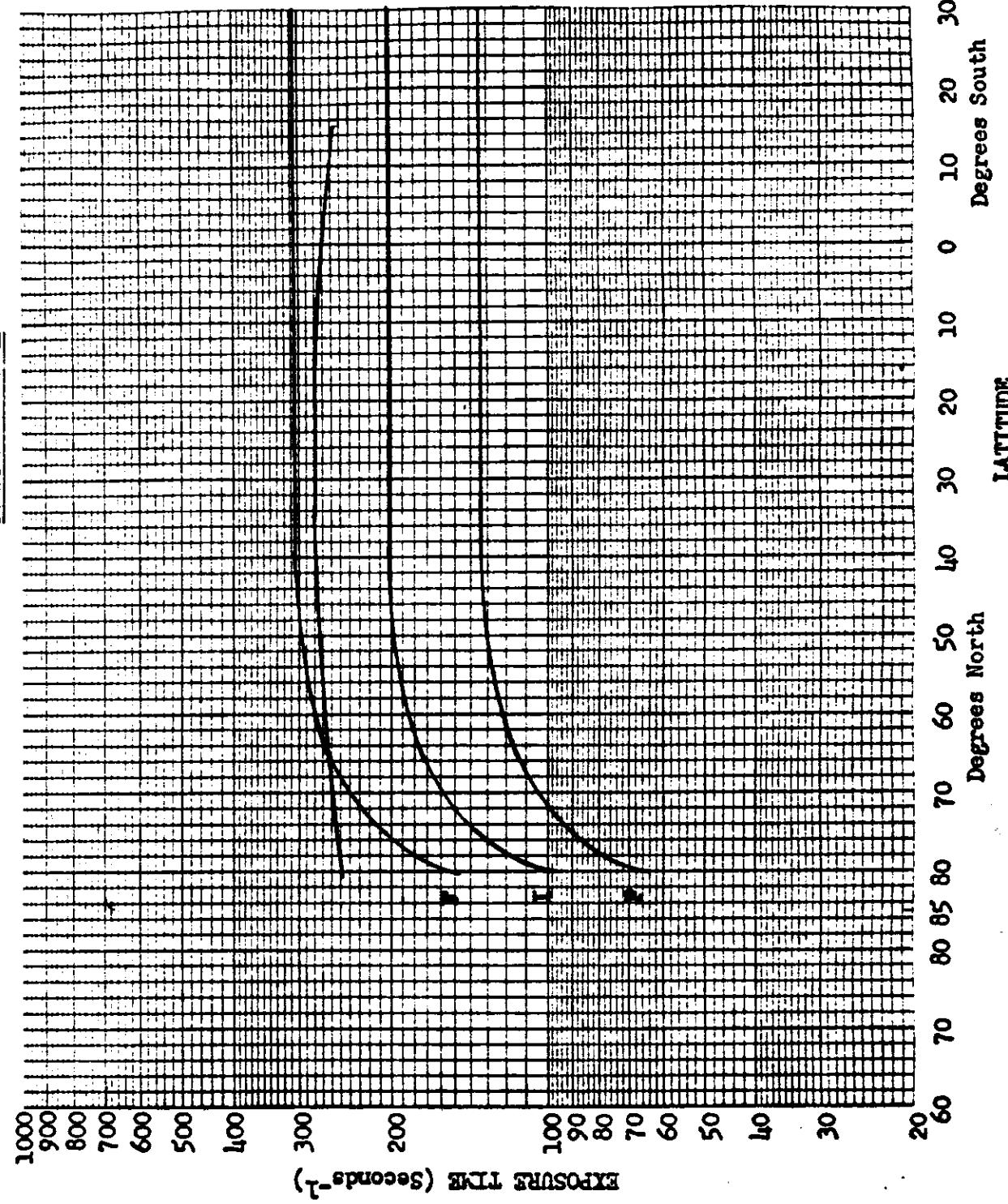


Figure 8-5

**EXPOSURE POINTS**



Mission No: 1018

Payload No: J-19

Camera No: FID

Pass No: 56

Launch Date: 3/25/65

Launch Time: 2111 2

Slit Width: .250

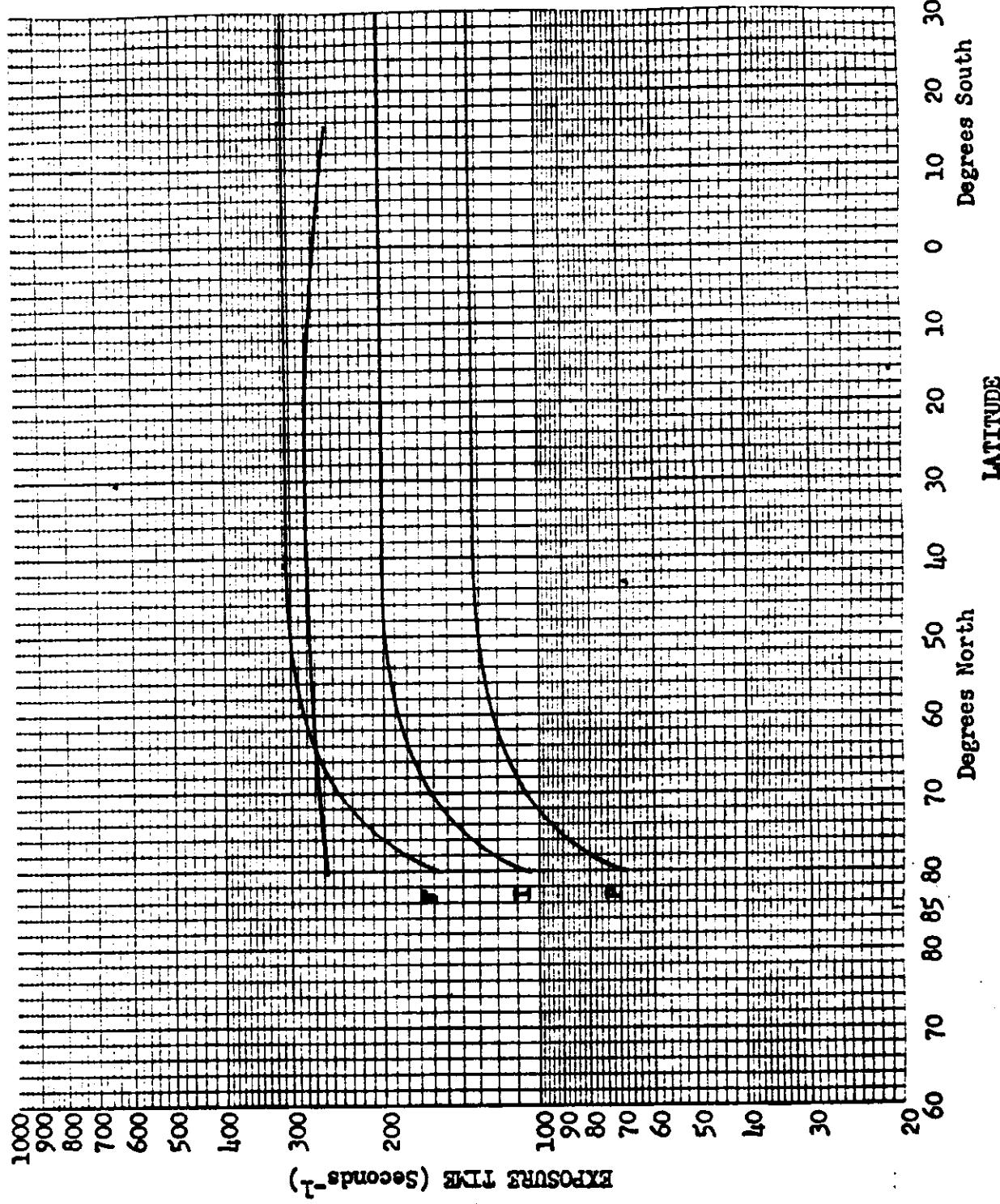
Filter Type: Wratten 25

Film Type: HQI

**NOTE:** Inclination 96°

Figure 8-6

**EXPOSURE POINTS**



Mission No: 1018

Payload No: J-19

Camera No: FWD

Pass No: 89

Launch Date: 3/25/65

Launch Time: 2111 Z

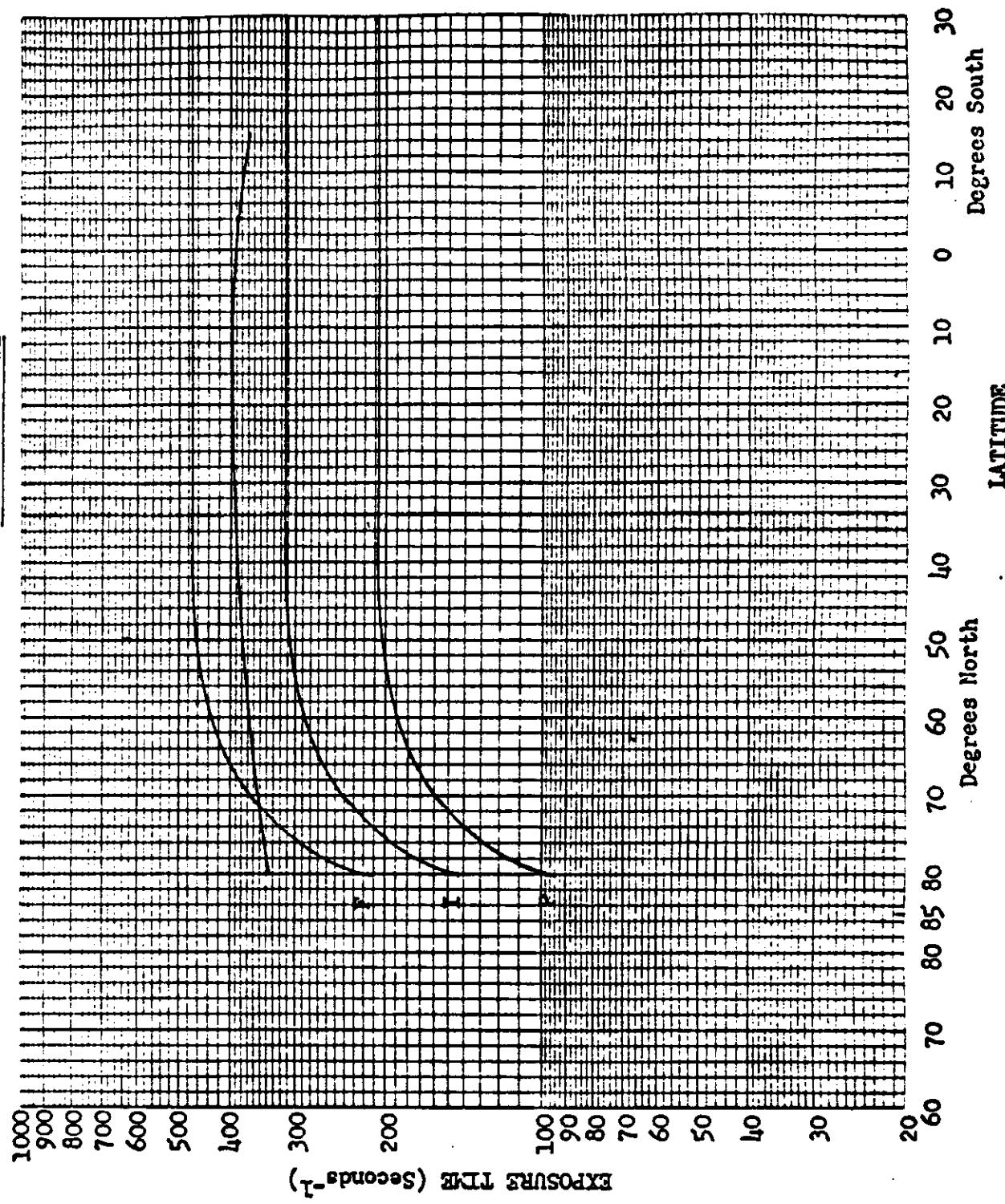
Slit Width: .250

Filter Type: Wratten 25

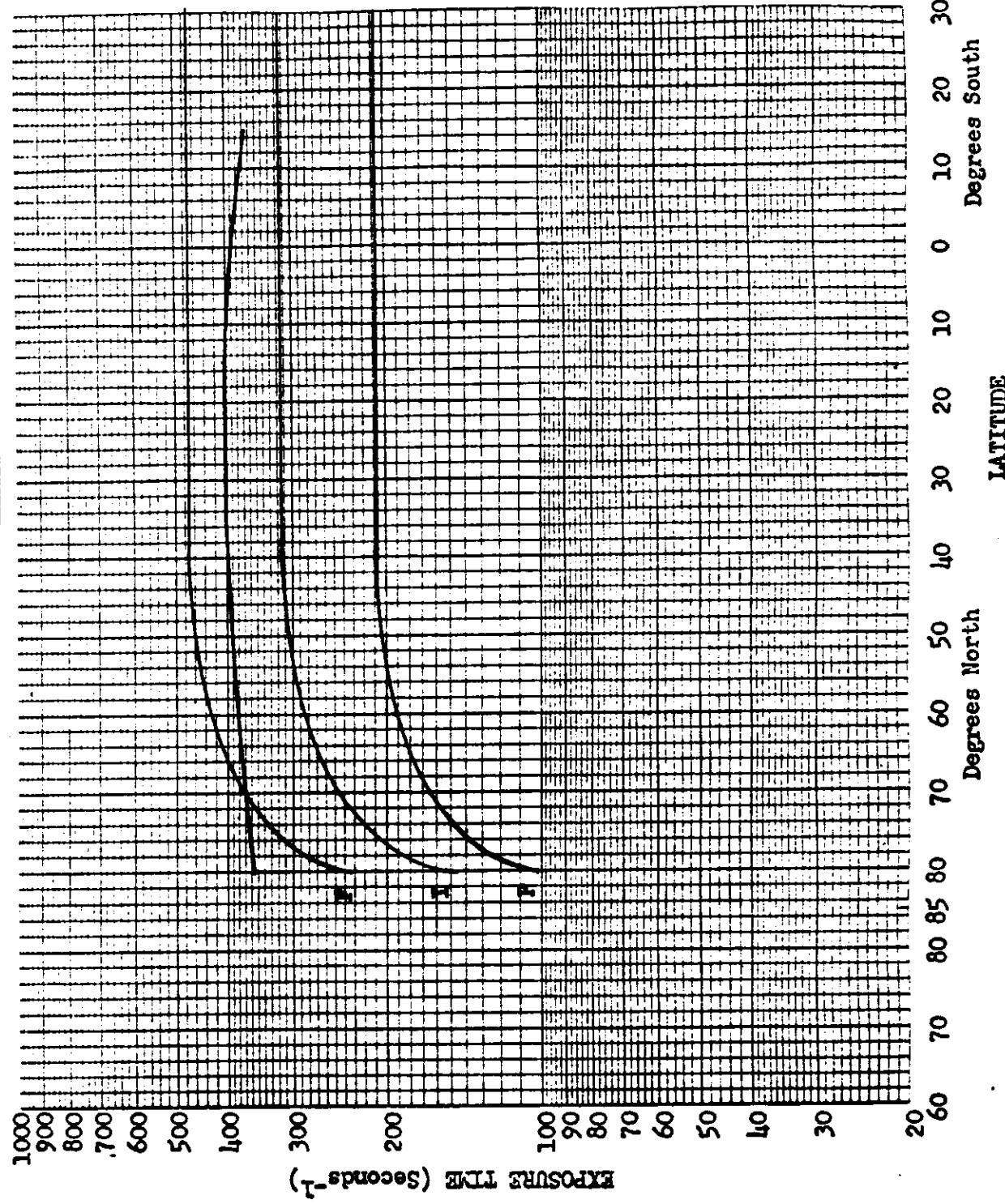
Film Type: Mgoh

NOTE: Inclination 96°

Figure 6-7  
(1)



EXPOSURE POINTS



Mission No: 1018

Payload No: J-19

Camera No: AFT

Pass No: 56

Launch Date: 3/25/65

Launch Time: 2111 Z

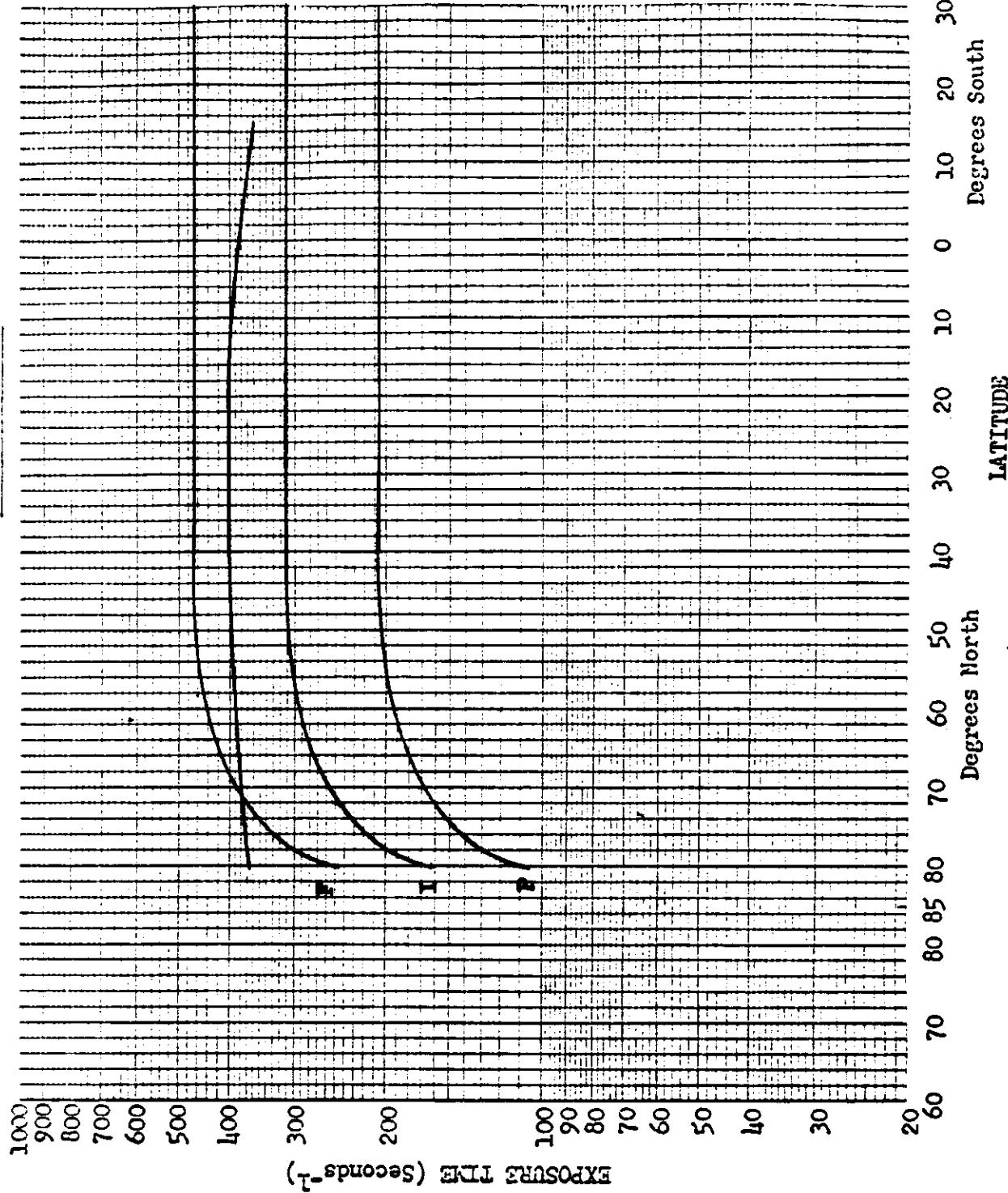
Slit Width: .175

Filter Type: Wratten 21

Film Type: U-404

NOTE: Inclination 96°

**EXPOSURE POINTS**



Mission No: 1018

Payload No: J-19

Camera No: AFT

Pass No: 89

Launch Date: 3/25/65

Launch Time: 2111 Z

Slit Width: .175

Filter Type: Wratten 21

Film Type: Mot

NOTE: Inclination 96°

Degrees South  
LATITUDE

Figure 3-1)

## SECTION 9

## DIFFUSE DENSITY MEASUREMENTS

The diffuse density measurements made by AFSPPF were computer sorted at A/P to permit analysis of the density ranges encountered at the three processing levels. A study of sorting techniques showed that no absolute method was available to separate the density values as the accuracy of the Processing History published by [REDACTED] appears rather low and processing transition phases are not accounted for. The sorting technique selected uses the base plus fog density values where measurements up to 0.09 density are considered as having received Primary processing, 0.10 to 0.17 as Intermediate and above 0.17 density as Full. The percentage of original negative that was processed at each level, based on the computer sort, is tabulated below with the predicted and reported processing percentages.

<u>Mission</u>	<u>Camera</u>		<u>Primary</u>	<u>Intermediate</u>	<u>Full</u>
1018-1	FWD	Predicted	0	0	100
		Reported	9	63	28
		Computed	1	84	15
1018-1	AFT	Predicted	0	22	78
		Reported	9	53	38
		Computed	5	72	23
1018-2	FWD	Predicted	0	0	100
		Reported	18	74	8
		Computed	17	80	3
1018-2	AFT	Predicted	0	15	85
		Reported	1	50	49
		Computed	0	74	26

The tabulations of density frequency distributions for Missions 1018-1 and 1018-2 are included in Appendix A, Table A-1 thru A-4. The graphical presentation of the density distribution are computed plotted in Appendix A Figures A-1 thru A-45.

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A summary of the processing and exposure analysis is shown in Table 9-1. The terrain D-Min criteria, (range) for proper exposure and processing is 0.40 to 0.90 density units. The area measured for D-Min is selected subjectively and is not necessarily the absolute D-Min in the photography.

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[REDACTED] - CONTROL NO. [REDACTED]

MISSION 1018-1 INSTR - FHC 06-11-65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPUSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	3	0 PC	67 PC	33 PC	0 PC	0 PC
INTERMEDIATE	262	0 PC	22 PC	69 PC	8 PC	0 PC
FULL	37	11 PC	0 PC	86 PC	3 PC	0 PC
ALL LEVELS	242	12 PC	19 PC	71 PC	7 PC	0 PC

MISSION 1018-1 INSTR - AFI 06-11-65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPUSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	11	0 PC	36 PC	36 PC	0 PC	27 PC
INTERMEDIATE	176	1 PC	14 PC	75 PC	10 PC	1 PC
FULL	156	4 PC	10 PC	89 PC	7 PC	0 PC
ALL LEVELS	243	1 PC	12 PC	77 PC	9 PC	2 PC

MISSION 1018-2 INSTR - FHC 06-11-65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPUSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	44	0 PC	32 PC	61 PC	0 PC	7 PC
INTERMEDIATE	204	0 PC	32 PC	54 PC	11 PC	3 PC
FULL	229	22 PC	30 PC	78 PC	9 PC	0 PC
ALL LEVELS	257	21 PC	31 PC	56 PC	9 PC	4 PC

MISSION 1018-2 INSTR - AFI 06-11-65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPUSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	182	0 PC	7 PC	63 PC	28 PC	3 PC
FULL	65	0 PC	0 PC	45 PC	55 PC	0 PC
ALL LEVELS	247	0 PC	5 PC	58 PC	35 PC	2 PC

PROCESS LEVEL + FCG EXPUSED UNDER PROCESSED CORRECT EXP+PROC OVER PROCESSED OVER EXPOSED

PRIMARY 0.01-C-09 0.01-C-13 0.14-0.39 0.40-0.90 0.91-1.34 0.91 AND UP

INTERMEDIATE 0.10-C-17 0.01-C-20 0.21-0.39 0.40-0.90 0.91-1.69 1.35 AND UP

FULL 0.18 ANC UP 0.01-C-39 0.21-0.39 0.40-0.90 0.91-1.69 1.70 AND UP

~~TOP SECRET~~

## SECTION 10

## PERFORMANCE MEASUREMENTS

The photography acquired by both panoramic cameras during Missions 1018-1 and 1018-2 received a MIP rating of 85. A summary is tabulated below of the MTF/AIM resolution values measured by AFSPPF and [REDACTED]. The microdensitometer slit used by AFSPPF and [REDACTED] was 1 micron by 80 microns.

<u>Mission</u>	<u>Camera</u>	<u>AFSPPF</u>	
1018-1	FWD	70	82
1018-1	AFT	74	96
1018-2	FWD	75	77
1018-2	AFT	71	91

The details of the measurement and computing techniques, targets measured and target locations are fully reported in the evaluation report published by AFSPPF and are not normally included in this report.

## SECTION 11

## OBSERVED DATA

Pass D16 of Mission 1018 contained the only photographs of ground resolution CORN targets. Indian Springs, Nevada, fixed target was not activated for this mission and probably was partially covered by blown sand. It could not be resolved.

At Pahrump, Nevada, both the fixed and mobile targets were resolved to Group 2, indicating a ground resolution performance in the range of 12'.

Pass D31 contained the PI selected MIP frame for Mission 1018-1, forward camera frame 3. At Blytheville AFB, Arkansas, engine nacelles could be detected on some large aircraft but could not be seen in the corresponding aft frame. A few frames later, the X bracing members of 3 steel truss bridges across the Mississippi River at Memphis could be seen about equally in fore and aft formats. Individual cars were clearly discernable in both.

Pass D32 covered Luke AFB near Phoenix, Arizona. Here engine nacelles were clearly seen in the forward photo but not in the aft. The aft print was much lower contrast than the forward and could have been affected by faulty print exposure.

Passes D48, D80 and D81 contained too much cloud, water, snow and haze to be useful.

Pass D95 contained the PI selected MIP frame for Mission 1018-2, forward camera frame #25. This is an unusual choice because it is the third frame of operation 2 of this pass. One of the criteria for selecting an MIP frame is that the first 10 frames of an operation shall be eliminated from consideration to insure optimum camera speed for the selected frame. The calculated IMC error for this frame was -9%. For the corresponding aft frame the IMC error was -3.9%. Discounting vehicle attitude rates, these errors would have produced resolution limits of 4 1/4' for the forward camera and 2' for the aft and would have no discernable effect on the photography produced by this system. A careful examination of these 2 frames at 30X magnification shows this to be the case.

Airport runways painted to FAA standards for all weather runways have threshold markers which are 5' wide bars with 3' spaces and are 50' long. Because this aspect ratio is higher than the 5:1 ratio of MIL STD 150A ground resolution targets, they cannot be used as a measure of ground resolution in the usual way. They do serve the useful purpose of providing an indication of whether or not the system can isolate a 3' dimension.

D95 passed directly over Boston and the orbit inclination was such that the ground track was parallel to the threshold markers of Logan Airport and 90° to those at Hanscom Field a few miles to the northwest. In both the MIP frame and the corresponding aft frame all of the 3' spaces could be seen, along track and cross track.

Although the best measured performance on this mission was approximately 12', under some conditions the system was capable of detecting a dimension as small as 3'.

## SECTION 12

## MISSION 1018-1 STELLAR-INDEX CAMERA

## A. COMPONENT ASSIGNMENT

Component	Serial Number
Camera	D20
Index Reseau	20
Stellar Reseau	20

## B. CAMERA DATA AND FLIGHT SETTINGS

## Stellar Camera:

Lens	85mm f/1.8
Exposure Time	2 seconds
Filter Type	None
Film Type	Eastman Type 3401

## Index Camera:

Lens	38mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	Eastman Type 3400

## C. POST FLIGHT EVALUATION

The camera exposed 55 frames of stellar and index photography before a failure of the stellar-index programmer precluded further operation. The stellar formats contained thirty or more star images. Quality was generally good, although some of the elongated images caused by image motion that had been observed during previous missions did occur in a few cases. Quality of the index formats appeared very good, although turbulent weather conditions appeared to generally be present. Minor static marks were noted outside the stellar formats.

## SECTION 13

## MISSION 1018-2 STELLAR-INDEX CAMERA

## A. COMPONENT ASSIGNMENT

Component	Serial Number
Camera	D22
Index Reseau	22
Stellar Reseau	22

## B. CAMERA DATA AND FLIGHT SETTINGS

## Stellar Camera:

Lens	85mm f/1.8
Exposure Time	2 seconds
Filter Type	None
Film Type	Eastman Type 3401

## Index Camera:

Lens	39mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	East man Type 3400

## C. POST FLIGHT EVALUATION

This instrument did not function in flight due to the stellar-index programmer failure during Mission 1018-1.

## SECTION 14

## VEHICLE ATTITUDE

The vehicle attitude errors for both Mission 1018-1 and 1018-2 were derived from the reduction of the Stellar camera photography. This attitude data is supplied to A/P by NPIC.

The attitude errors for each frame and the attitude control rates are calculated at the A/P computer facility. The computer also plots the frequency distribution of the rates and errors. Figures 14-1 through 14-4 show these distributions for Mission 1018-1 and Figures 14-5 through 14-8 for Mission 1018-2.

The summary table below lists the maximum attitude errors and rates that were experienced during 90% of the FWD camera photographic operations, excluding the first six frames of each operation, and the total range of the errors and rates.

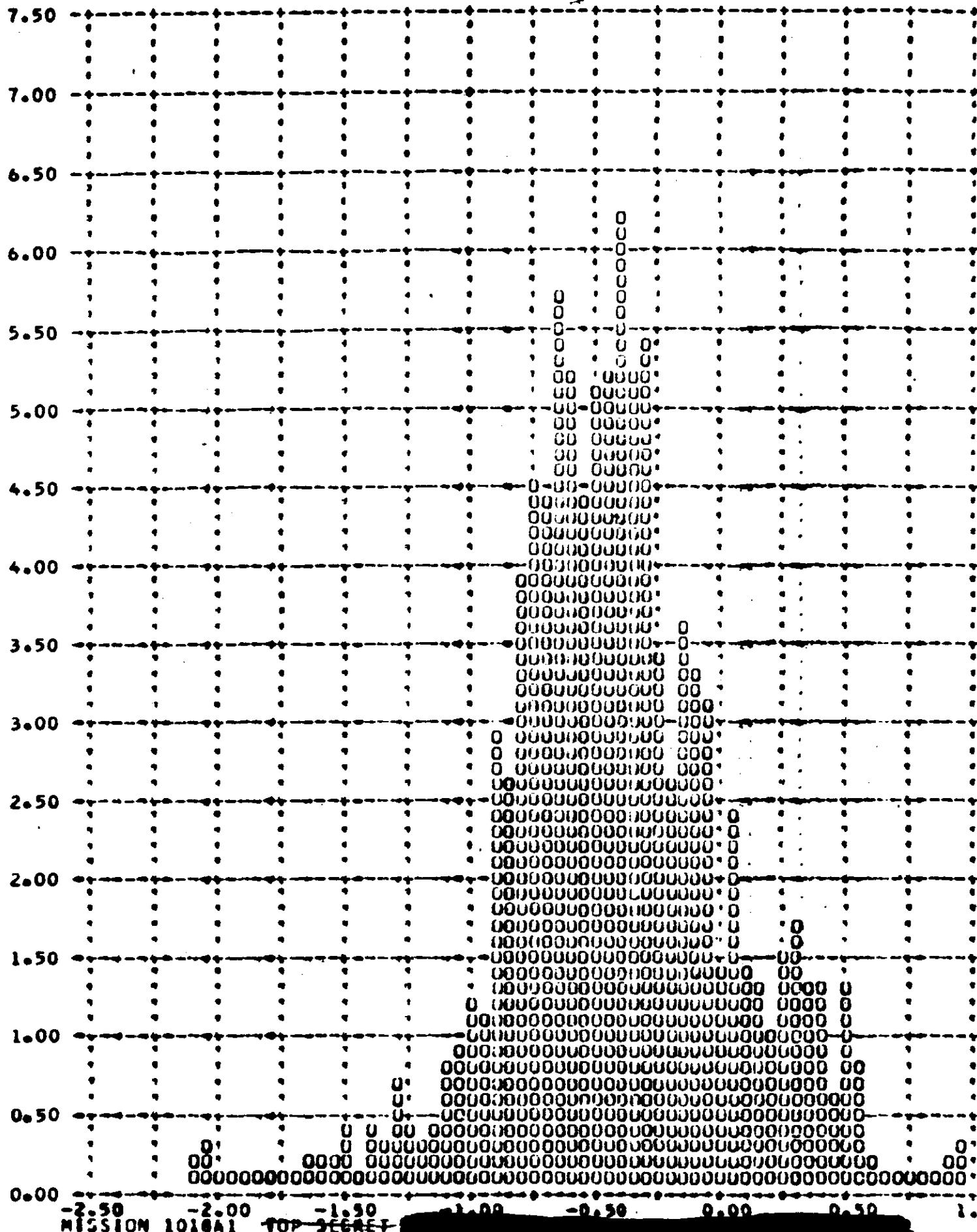
<u>Value</u>	Mission 1018-1		Mission 1018-2	
	<u>90%</u>	<u>Range</u>	<u>90%</u>	<u>Range</u>
Pitch Error ( $^{\circ}$ )	0.91	-2.1 to +0.95	0.84	-2.1 to +0.40
Roll Error ( $^{\circ}$ )	0.48	-0.8 to +0.60	0.63	-1.0 to +0.75
Yaw Error ( $^{\circ}$ )	N/A		N/A	
Pitch Rate ( $^{\circ}/hr.$ )	47.37	-95 to +85	34.74	-95 to +100
Roll Rate ( $^{\circ}/hr.$ )	36.73	-50 to +100	30.70	-50 to +100
Yaw Rate ( $^{\circ}/hr.$ )	N/A		N/A	

The performance of the attitude control system is comparable to the control systems used on recent missions. The panoramic photography was not degraded by the attitude control system.

J-19 A BUCKET - FWD INSTR

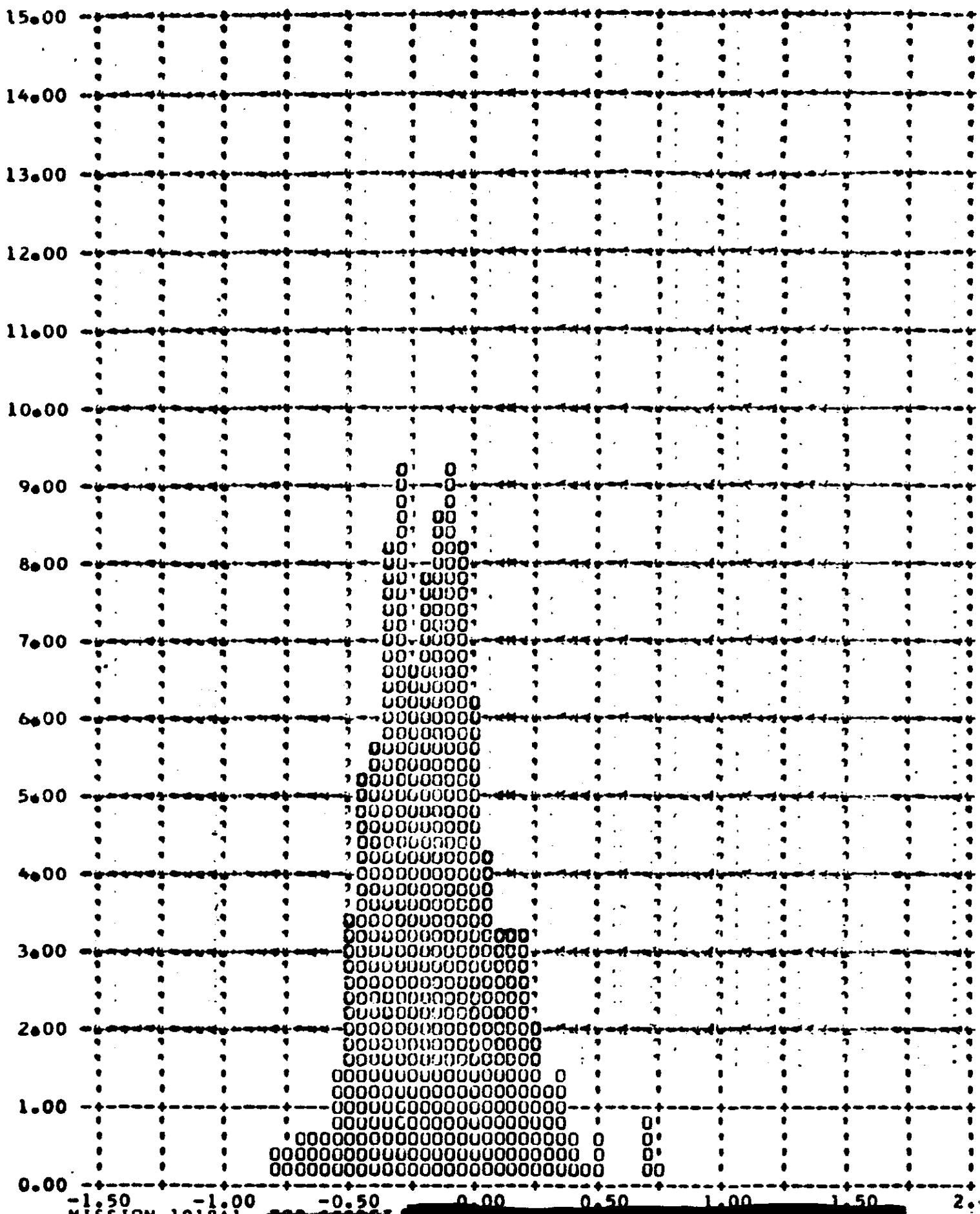
FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 0.9

Y PITCH ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



J-19 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 0.48

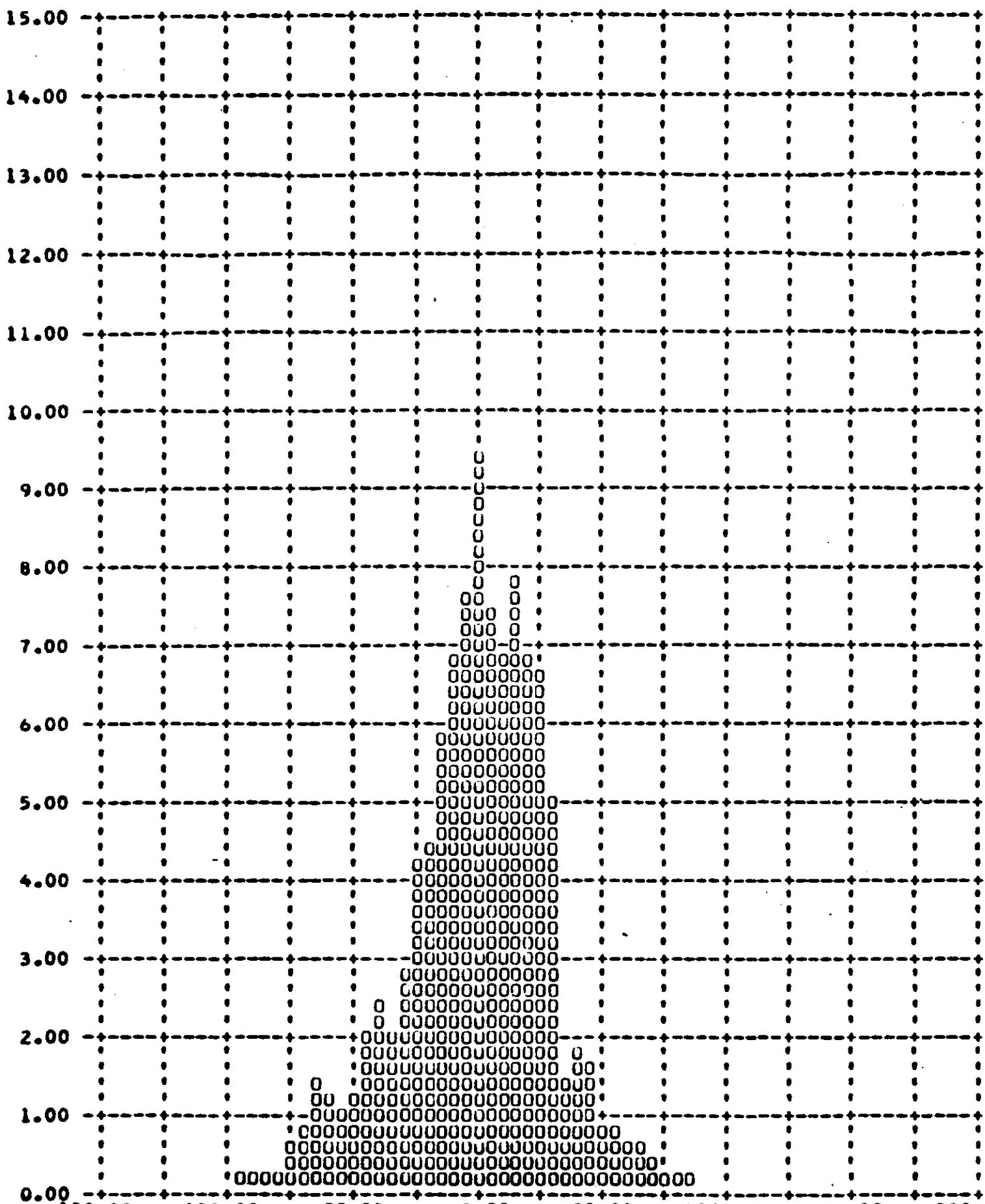
Y ROLL ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 1018A1 ~~TOP SECRET~~

J-19 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 47.37

Y PITCH RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)

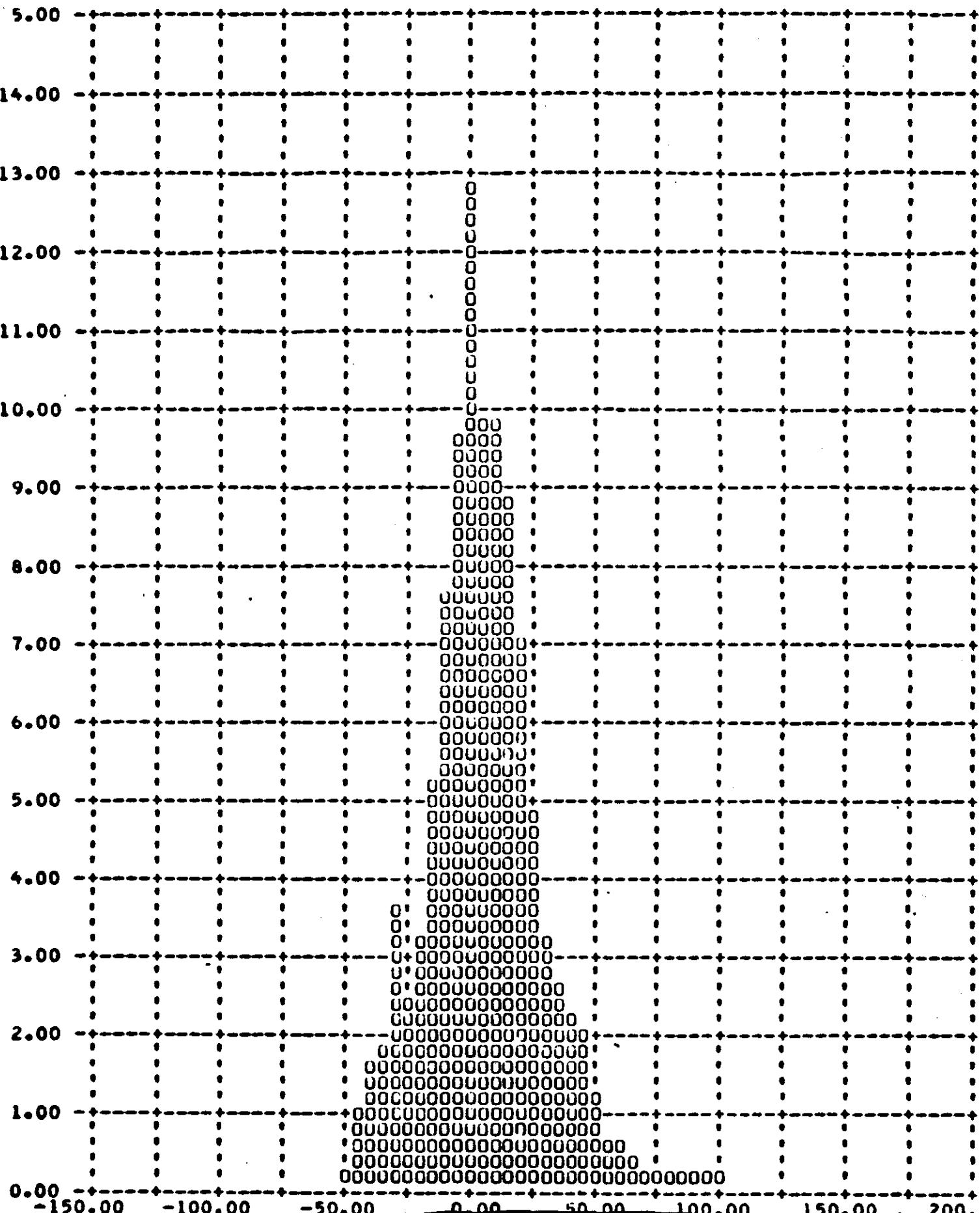


MISSION 1018A1 ~~TOP SECRET~~

MISSION 1018A1 ~~TOP SECRET~~

J-19 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 36.73

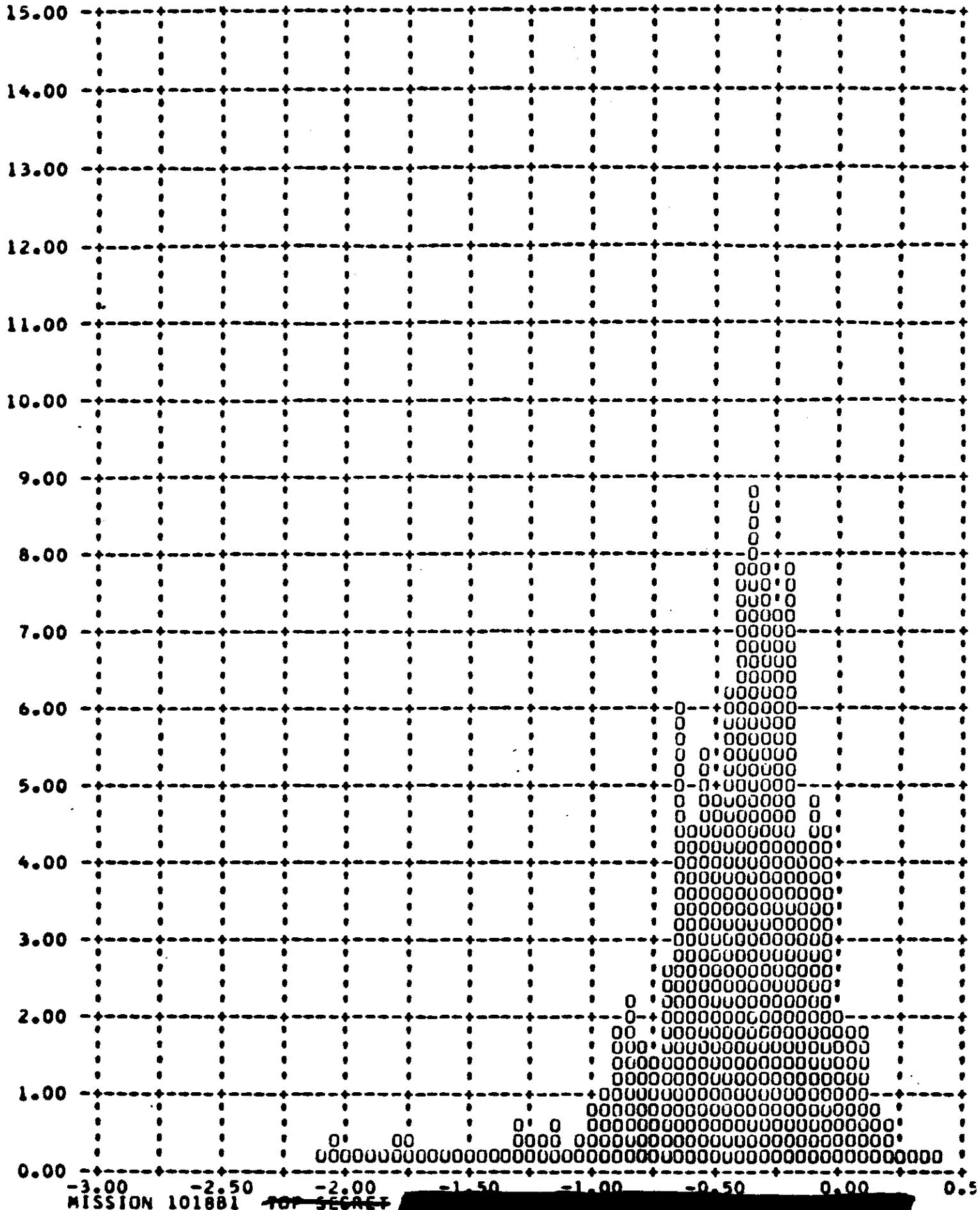
Y ROLL RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 1018A1 ~~TOP SECRET~~

J-19 B BUCKET - FWD INSTR      FRAMES 1-6 OF EACH OP OMITTED    90 PERCENT = 0.84

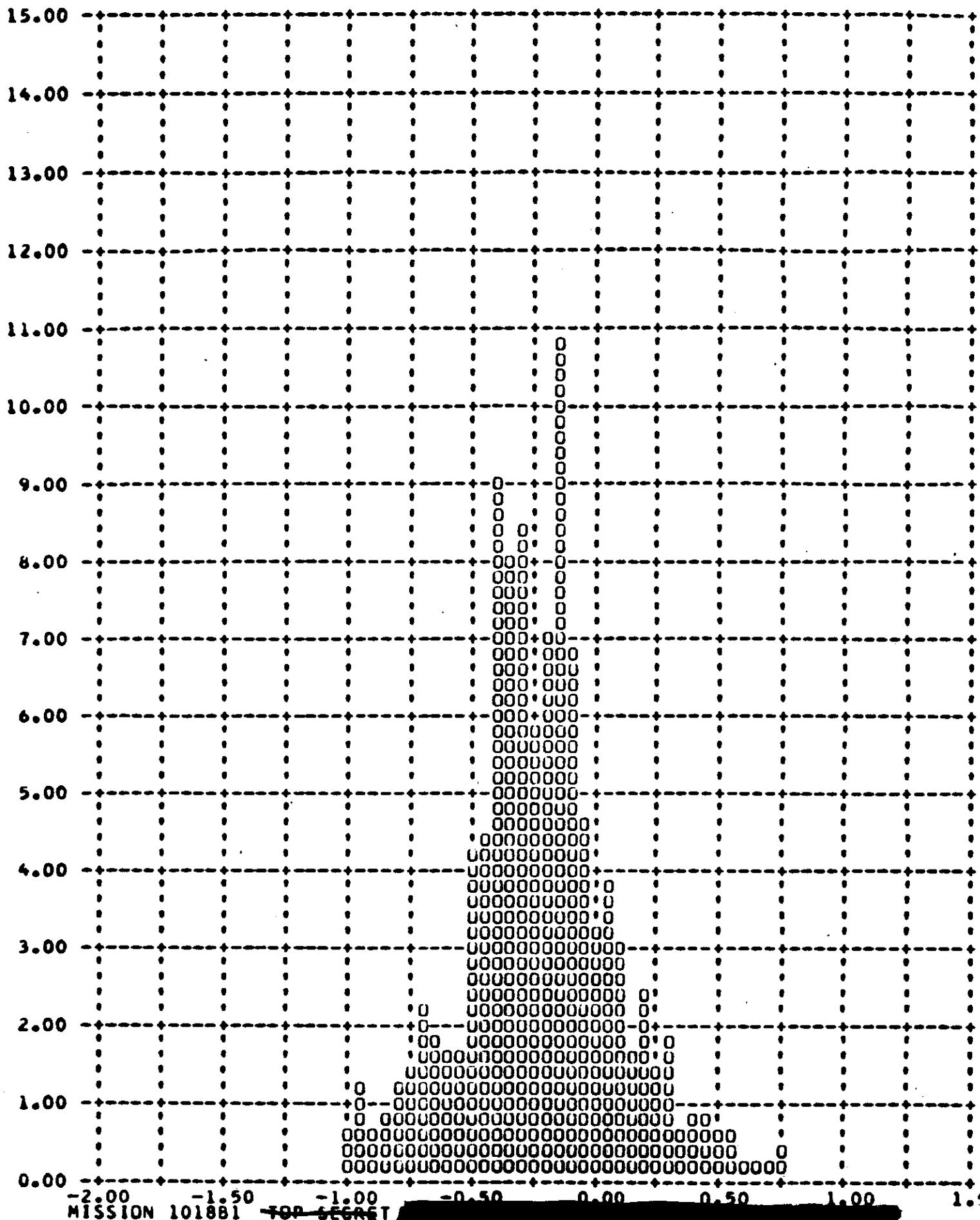
Y PITCH ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 101881 TOP SECRET

J-19 B BUCKET - FWD INSTR      FRAMES 1-6 OF EACH OP OMITTED    90 PERCENT = 0.63

Y   ROLL ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)

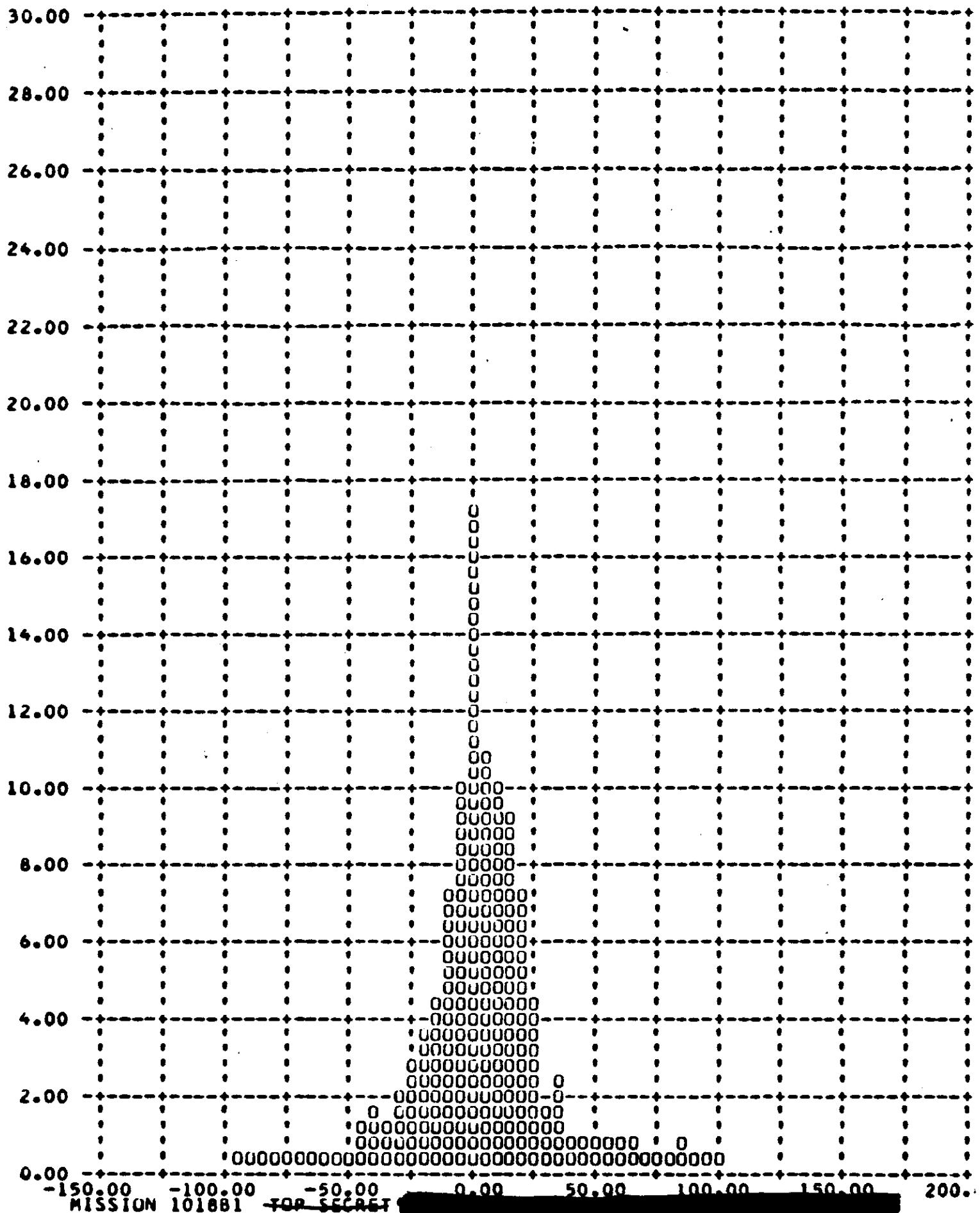


MISSION 101881 TOP SECRET

J-19 B BUCKET - FWD. INSTR

FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 34.74

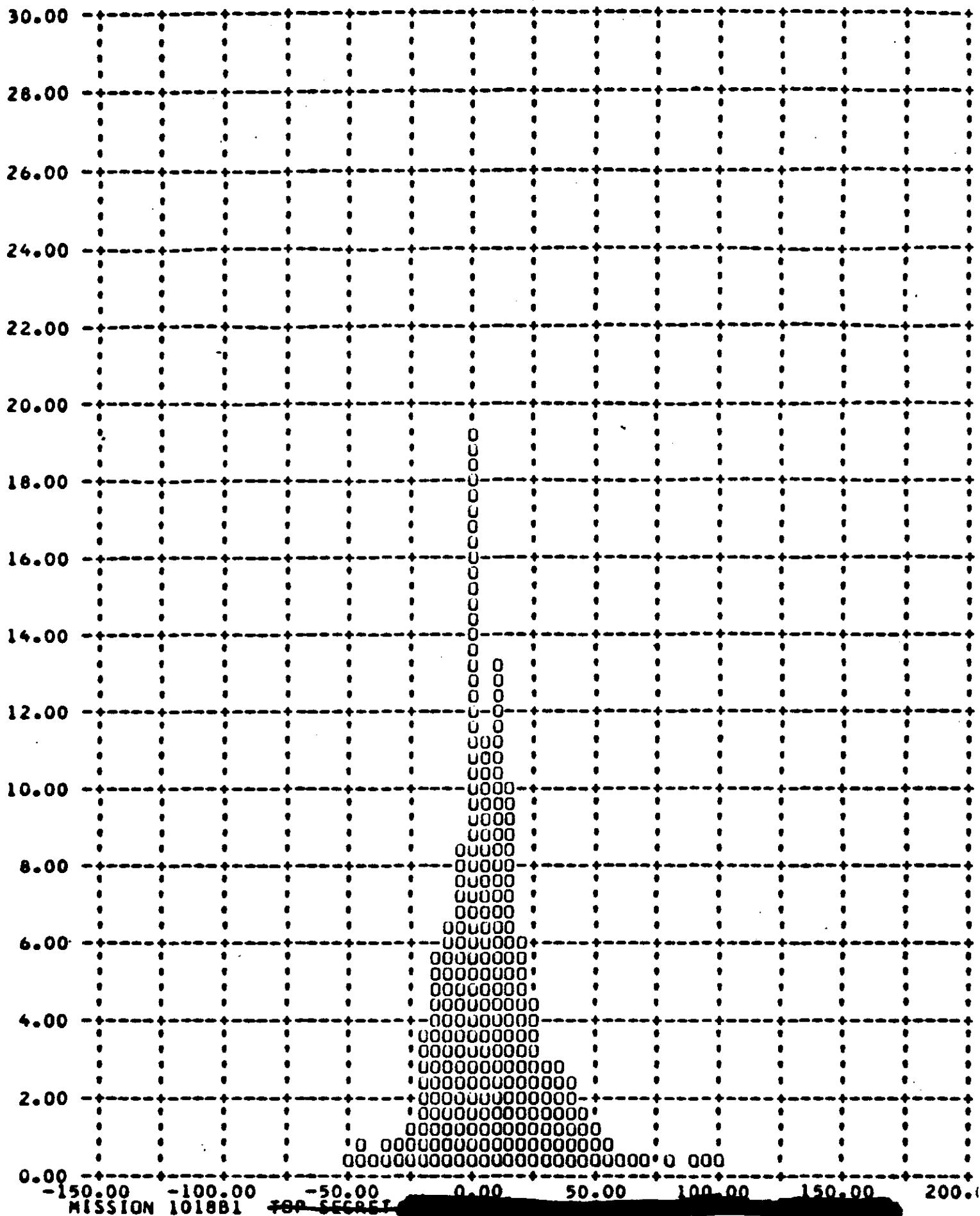
Y PITCH RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 101881 TOP SECRET

J-19 B BUCKET - FWD INSTR FRAMES 1-6 OF EACH OF OMITTED 90 PERCENT = 30.70

Y ROLL RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)



-150.00 -100.00 -50.00 0.00 50.00 100.00 150.00 200.00  
MISSION 1018B1 ~~TOP SECRET~~

## SECTION 15

## IMAGE SMEAR ANALYSIS

The frame correlation tape supplied to A/P by NPIC contains the binary time word of each frame of photography. A computer program has been assembled at A/P which calculates the exposure time of each frame and compares the camera cycle rate with the ephemeris to calculate the V/h mismatch. This data is combined with the vehicle attitude error and rate values of each frame and the crab error caused by earth rotation at the latitude of each frame. The program outputs the total along track and cross track IMC error and the limit of ground resolution that can be acquired by a camera regardless of focal length and system capabilities.

The computer rejects the first six frames of all operations as the large V/h error induced by camera start-up is not representative of the overall system operations. The frequency distribution of the V/h errors and resolution limits are computer plotted and are shown in Figures 15-1 through 15-8.

The summary table 15-1 presents the maximum V/h ratio errors and resolution limits that existed during 90% of the photographic operations and the total range of values during all operations that were computed.

~~TOP SECRET~~

MISSION 1018  
V/h RATIO AND RESOLUTION LIMITS

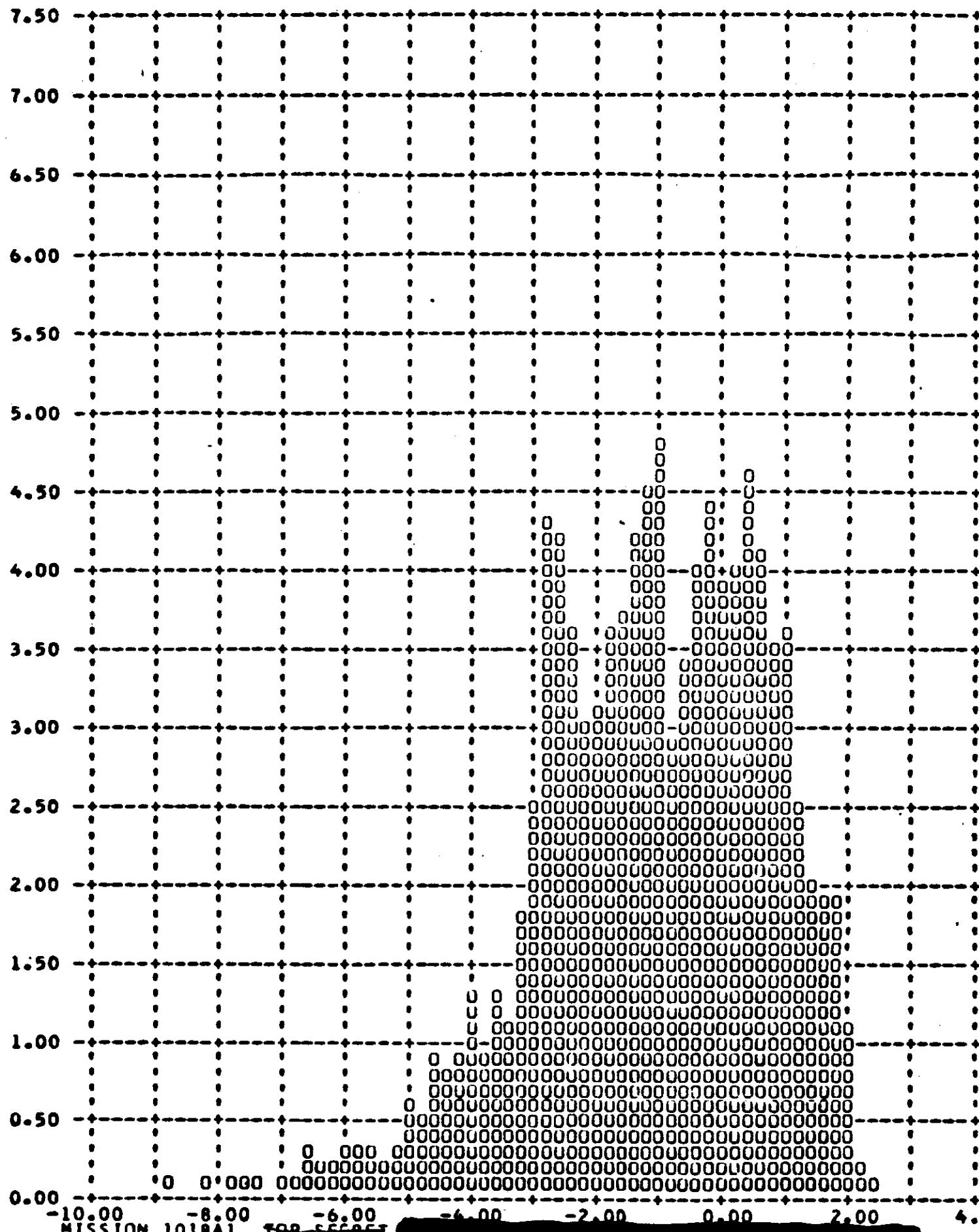
VALUE	UNITS	CAMERA	MISSION 1018-1		MISSION 1018-2	
			90% Range	90% Range	90% Range	90% Range
V/h Ratio Error	FWD		3.39	-8.8 to +2.4	3.09	-6.2 to +2.6
	AFT		3.16	-7.4 to +2.0	2.79	-4.4 to +3.0
Along Track Resolution Limit	FWD	Feet	5.64	0.2 to 9.0	5.57	0.2 to 7.6
	AFT		3.69	0.2 to 4.8	4.07	0.2 to 5.6
Cross Track Resolution Limit	FWD	Feet		No Data Available		
	AFT					

TABLE 15-1

~~TOP SECRET~~

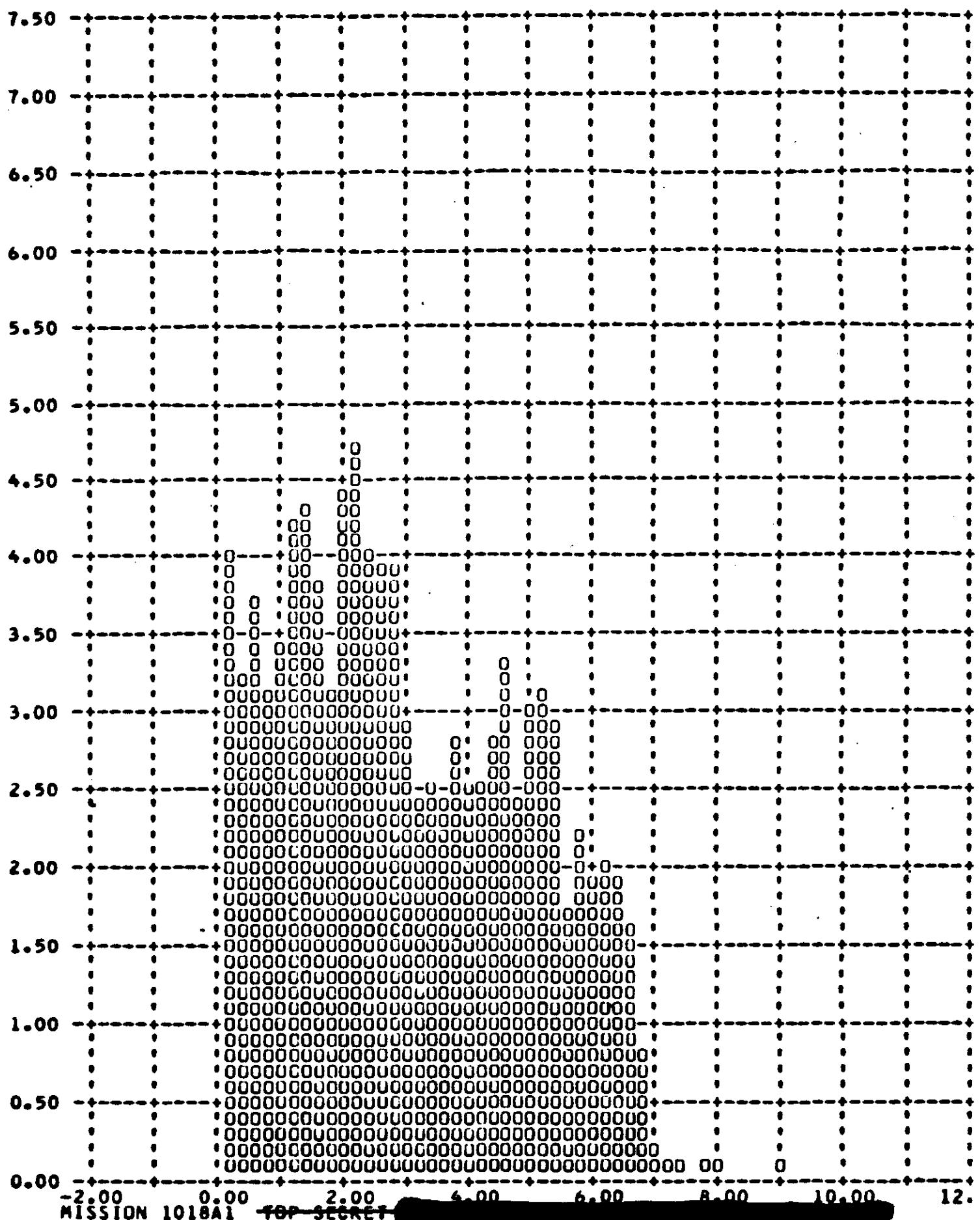
J-19 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 3.39

Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)



J-19 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 5.64

Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (Y)

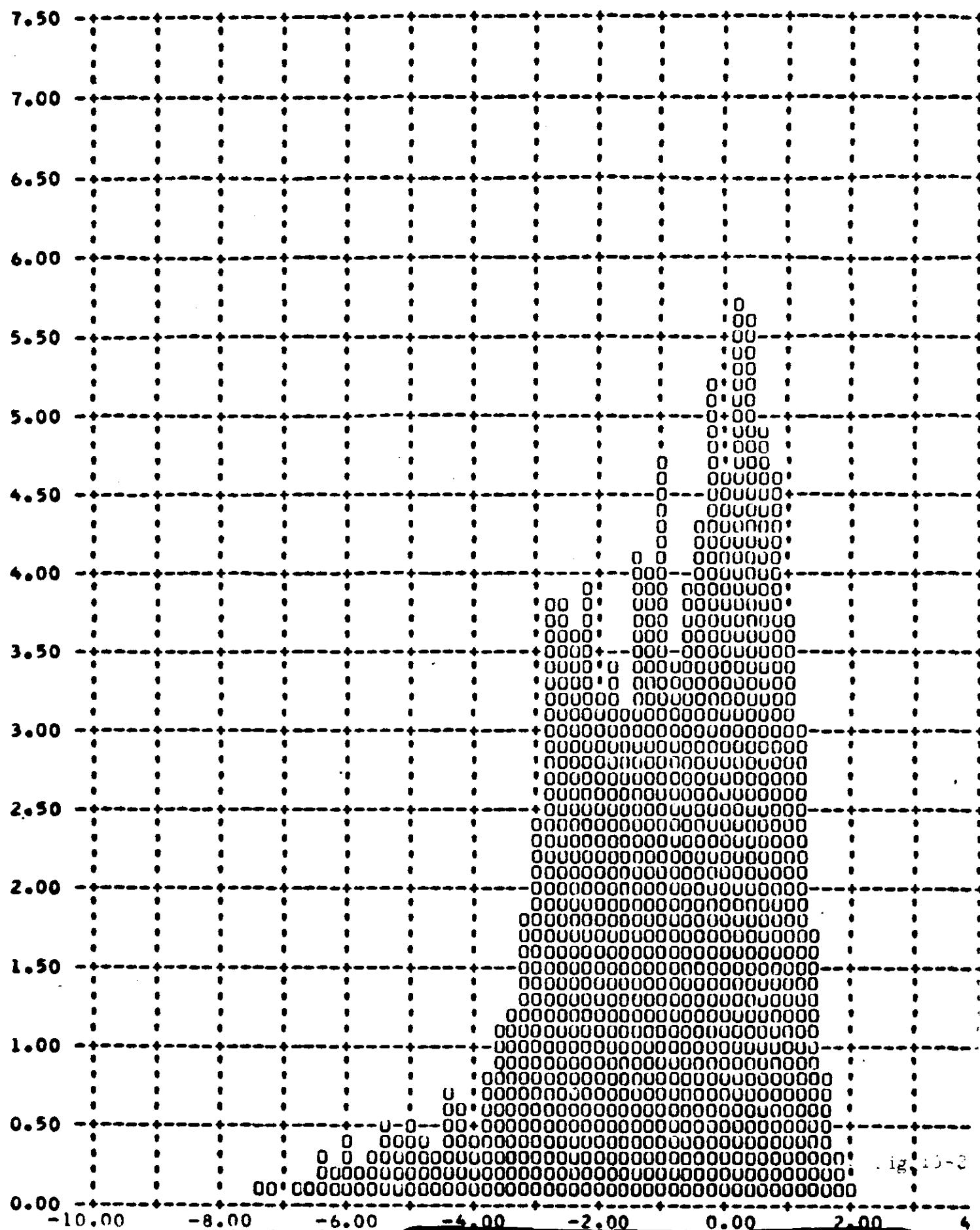


MISSION 1018A2 ~~TOP SECRET~~

J-19 A BUCKET - AFT INSTR

FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 3.16

Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)



J-19 A BUCKET - AFT INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 3.69

Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (Y)

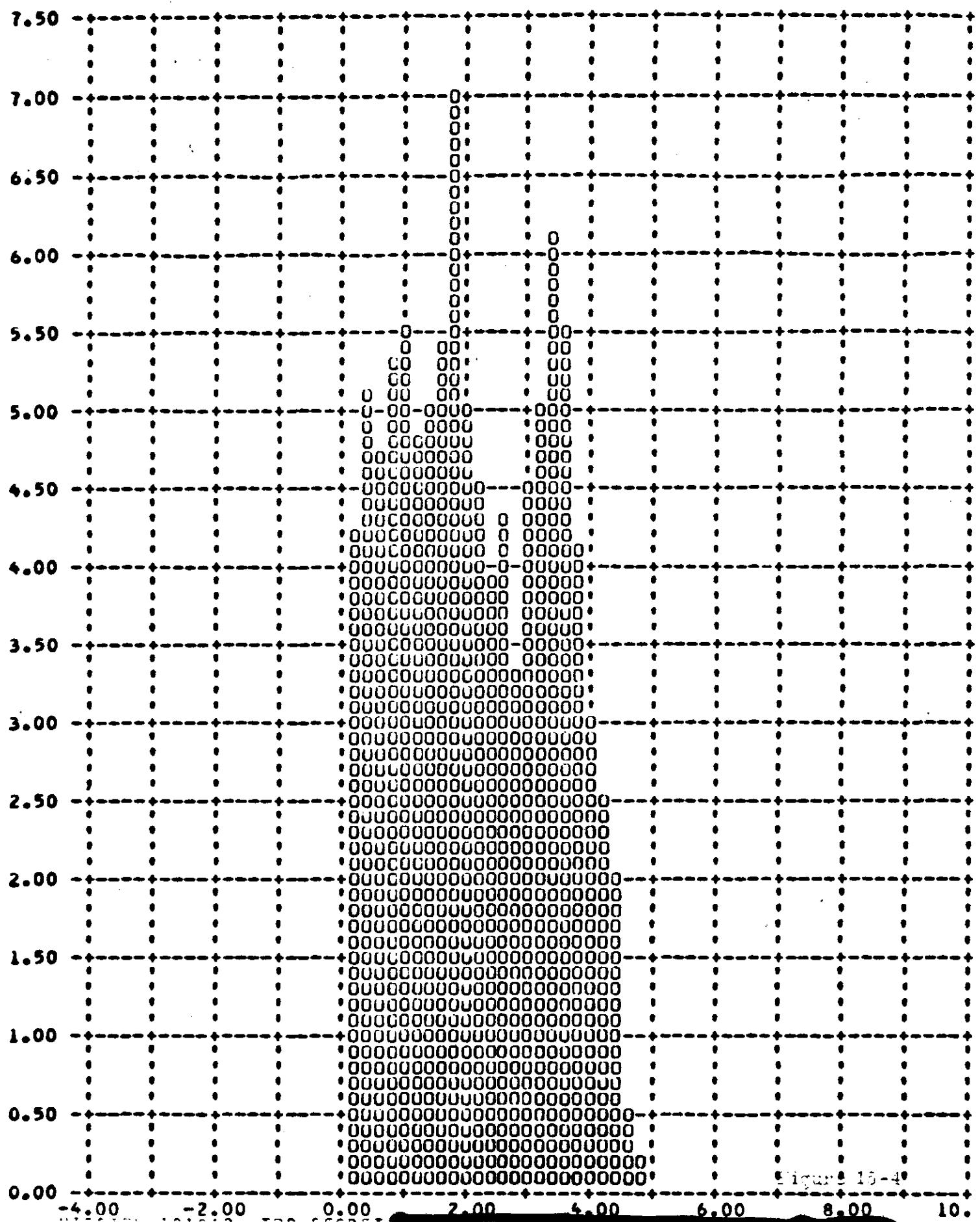
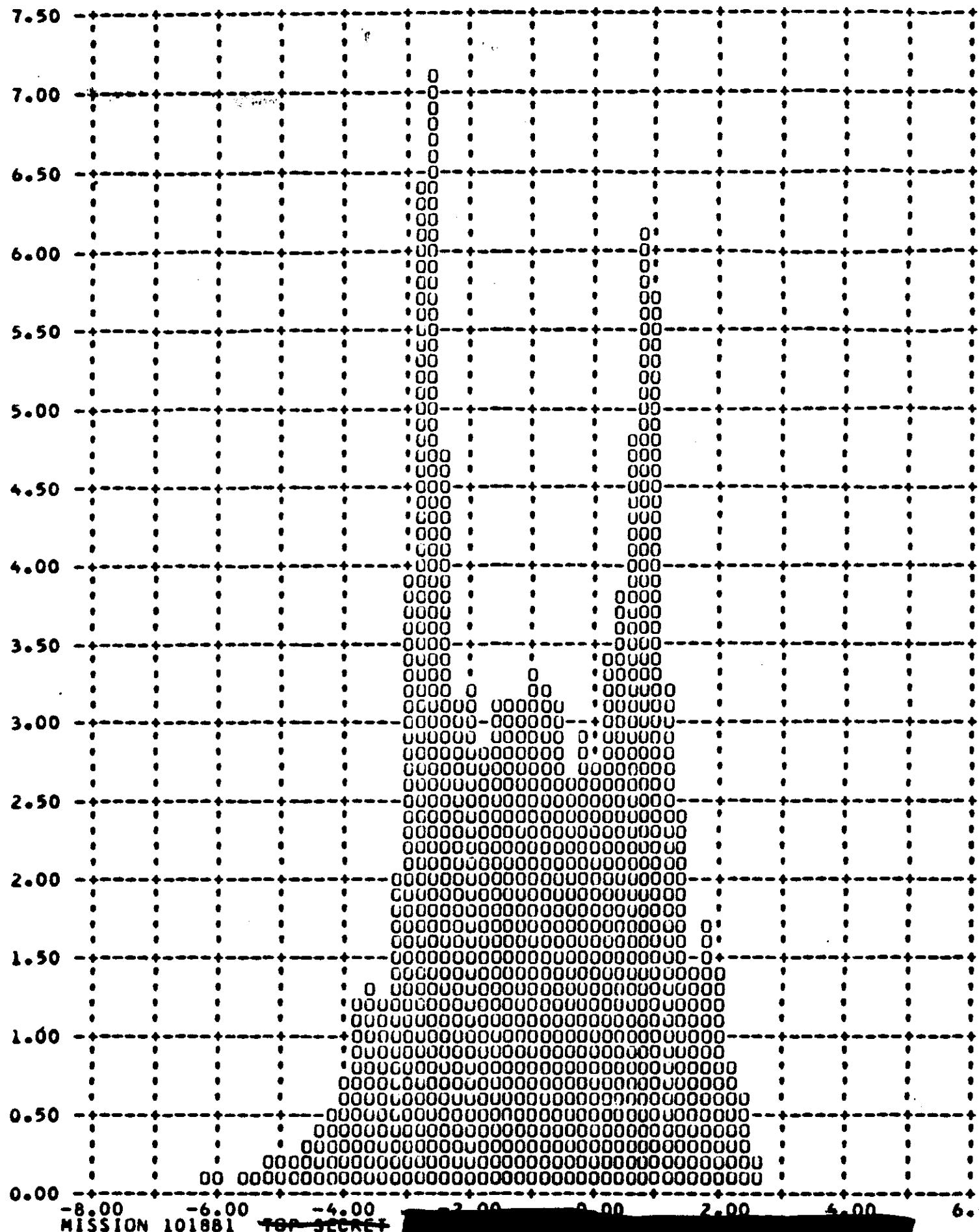


Figure 16-4

J-19 B BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 3.09

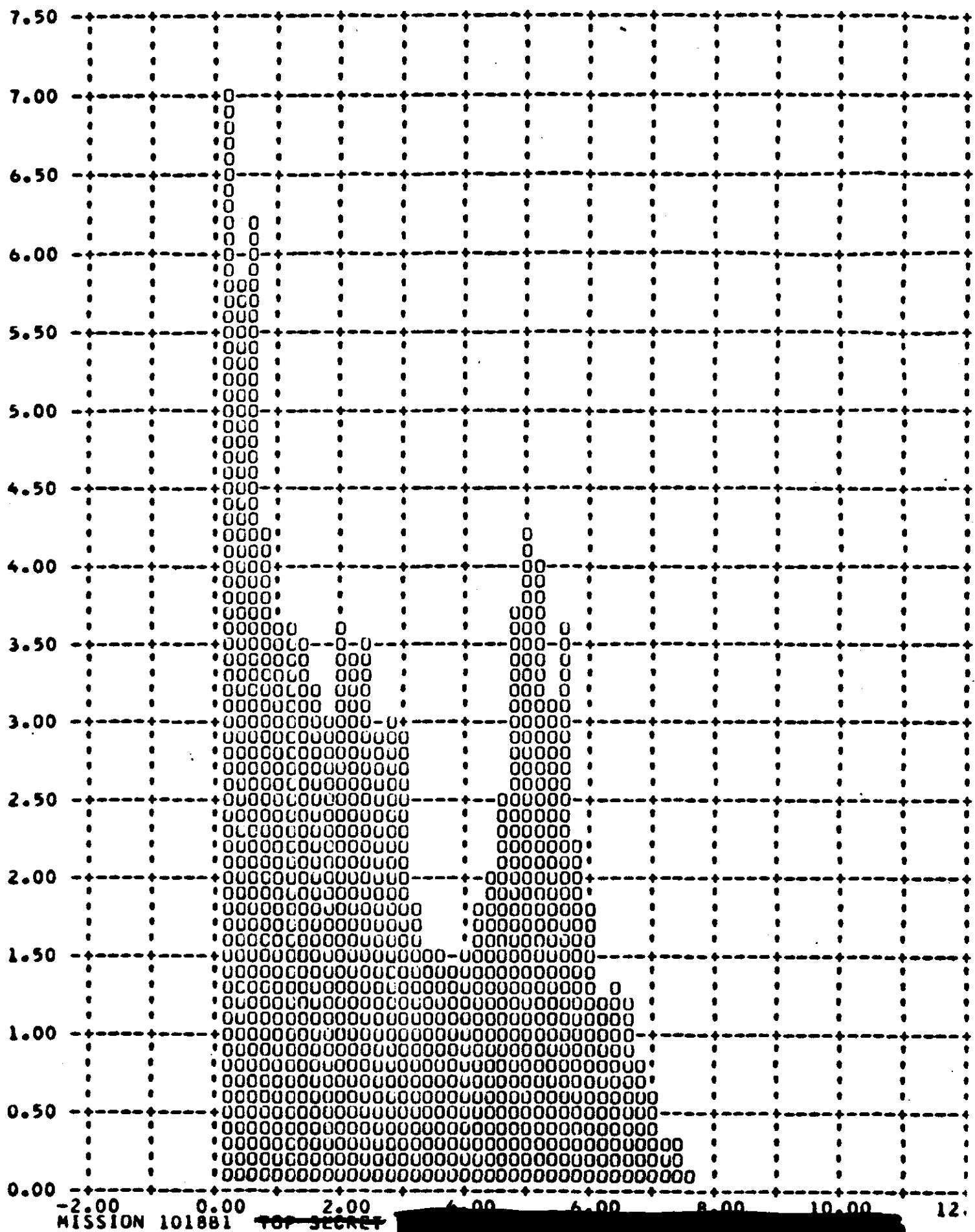
Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 101881 TOP SECRET

J-19 B BUCKET - FWD INSTR      FRAMES 1-6 OF EACH OP OMITTED    90 PERCENT = 5.57

Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (Y)

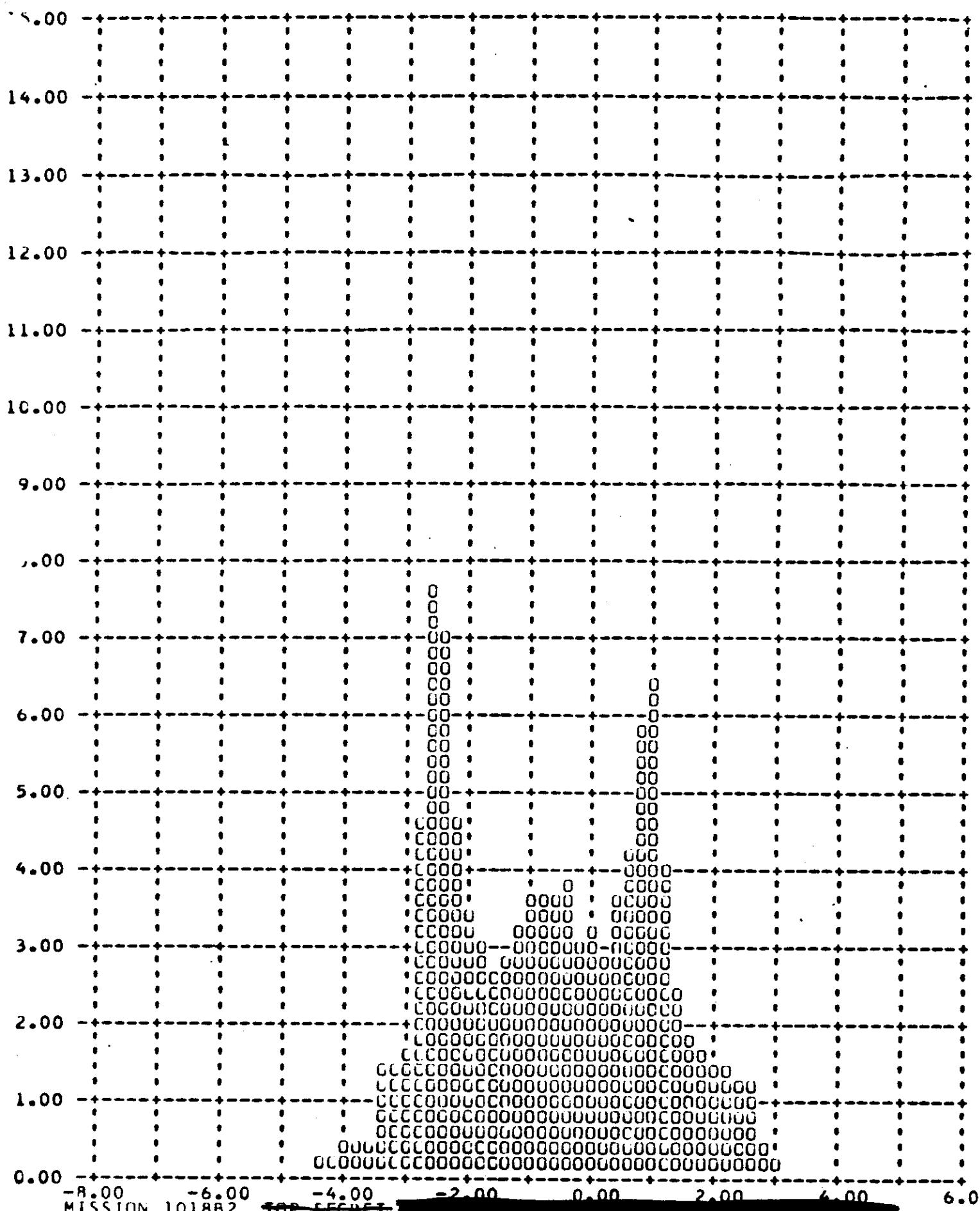


MISSION 101881 TOP SECRET

MISSION 101882 ~~TOP SECRET~~

-19 B - BUCKET AFT CAMERA FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 2.79

Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)

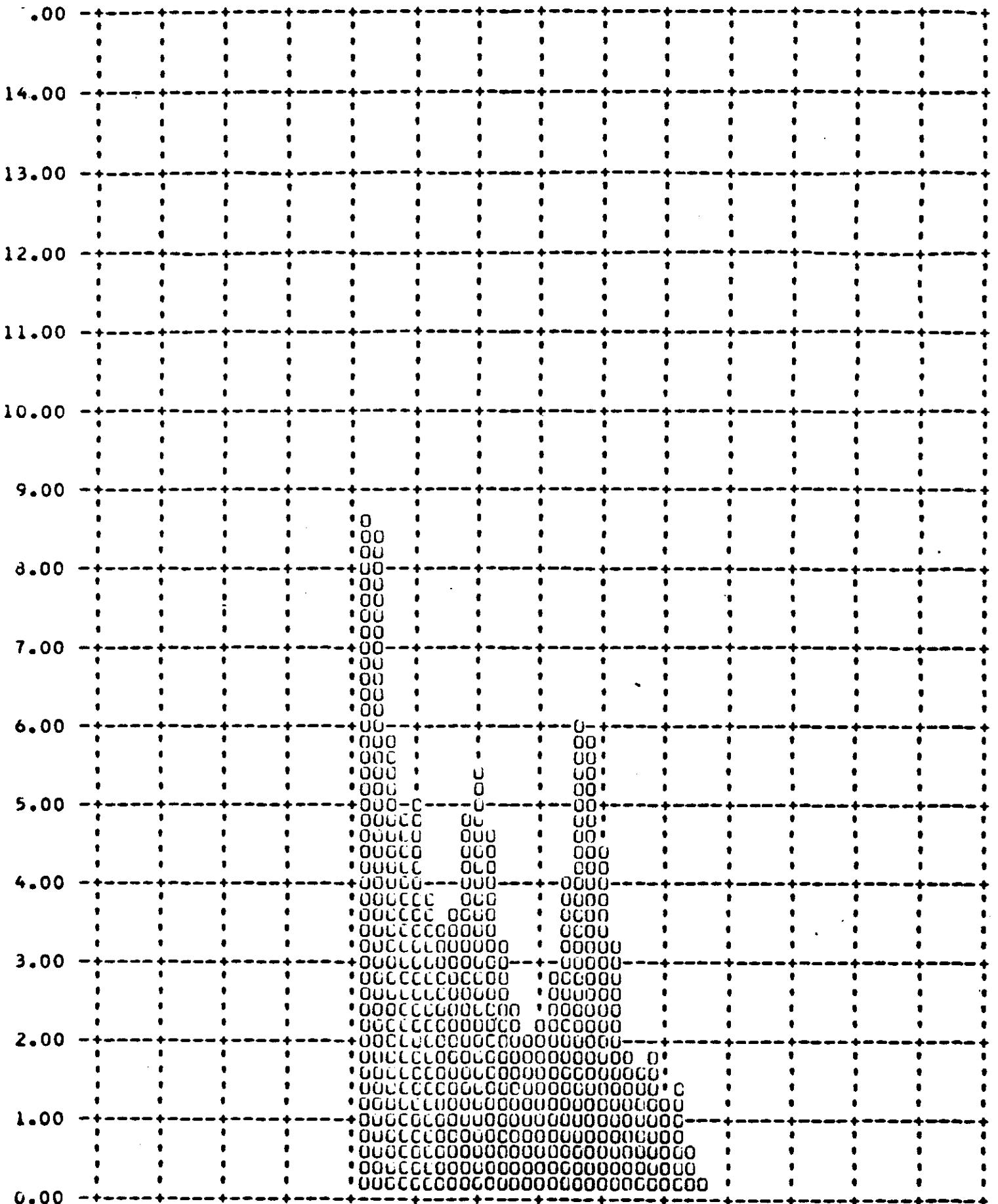


MISSION 101882 ~~TOP SECRET~~

MISSION 101882 ~~TOP SECRET~~

J-19 B - BUCKET AFT CAMERA FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 4.07

Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 101882 ~~TOP SECRET~~

## SECTION 16

## RADIATION DOSAGE

Each recovery system flown on a Corona mission contains a sealed packet of Eastman Type 3401 and Royal X Pan emulsions to determine the total radiation received at the take-up cassette. Both film types have been irradiated by LMSC at various levels and the base plus fog densities recorded after controlled processing.

Following recovery the film dosimeter packets are removed at A/P and processed with a pre-flight sample of the same film type and sensitometric control film. The resulting base plus fog density measurement of the dosimeter strips is used to ascertain the total radiation level. The table below-presents the base plus fog readings for the dosimeter strips and the radiation level equivalents.

<u>Emulsion</u>	Mission 1018-1	
	<u>B + F</u>	<u>Density</u>
Type 3401	.15	0.37 R
Royal X Pan	.16	0.13 R

Mission 1018-2		
Type 3401	0.16	0.43 R
Royal X Pan	0.16	0.13 R

The mean total radiation seen by the take-up cassettes during both missions was approximately 0.2 roentgens. This level is essentially the same as received during recent missions and is below the level that will degrade the panoramic photography.

## SECTION 17

## SYSTEM RELIABILITY

Reliability calculations for the payload are based on a sample beginning with M-7. Hence both the major part of the Mural Program and the "J" Program are covered in the calculation. For certain auxiliaries, i.e., the stellar-index camera and the horizon cameras, the sample size is changed to recognize incorporation of modified equipment or new designs where reliability was one of the principal reasons for the modification. However, for primary mission function, the sample size is consistent with reliability reporting for the vehicle.

The reliability estimates of this section deal exclusively with the payload. Failures to achieve orbit or vehicle induced failures are thereby excluded. Recoveries before a complete mission has been completed are considered as full missions providing that early termination was caused by reasons not connected with payload operation. Film quality is not considered in the reliability estimate calculation. Hence, only electrical and mechanical functioning are considered.

The reliability estimate is also divided into primary and secondary functions. The primary functions are operation of the panoramic cameras, main camera door operation, operation of the payload clock, and recovery operations. The secondary mission functions are horizon camera operation excluding catastrophic open shutter failure mode, auxiliary data recording, and stellar-index camera operation.

**Panoramic Camera Reliability**

Sample Size - 98 opportunities to operate.

One failure - S/I Programmer on System J-19.

Assume - 3000 cycles per camera per mission.

Estimated Reliability = 98.3% at 50% confidence level.

**Main Camera Door Reliability**

Sample Size - 36 vehicles x 2 doors = 72 opportunities to operate.

Estimated Reliability = 99.1% at 50% confidence level.

### Payload Command and Control Reliability

Sample Size: 4920 hours operation

1 Failure

Estimated Reliability - 96.8% at 50% confidence level

### Payload Clock Reliability

Sample Size - 4920 hours operation

No failures

Estimated Reliability - 98.7% at 50% confidence level

Estimated Reliability of Payload Functioning on orbit - 96.7% at 50% confidence level.

### Recovery System Reliability

38 opportunities to recover

1 Failure - improper separation due to water seal - cutter failure.

Estimated Reliability - 95.6% at 50% confidence level.

### Stellar-Index Camera Reliability

Sample begins with J-5

Sample Size - 8980 cycles

Number of failures - 1

Estimated Reliability - 92.3% at 50% confidence level

### Horizon Camera Reliability

Sample includes J5A and up

Sample Size: 40,500 cycles

Estimated Reliability of Single Camera - 97.5% at 50% confidence level.

Estimated Reliability of Four Horizon Cameras at a Parallel

Redundant System - 99.9% at 50% confidence level.

FOR STOCK

# ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS	SECONDARY FUNCTIONS		ON - ORBIT FUNCTIONS	RECOVERY SYSTEM	STELLAR - INDEX CAMERAS	HORizon ON CAMERAS
		SAMPLE FAILURES	RELIABILITY				
1008	PANORAMIC CAMERA COMMAND & CONTROL SYSTEM DOORS	0	99.0	3124 HOURS	0	98.1	99.7
1009	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	98.2	99.7
1010	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	98.4	99.7
1011	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	98.6	99.7
1012	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	98.9	99.7
1013	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	99.0	99.7
1014	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	99.3	99.7
1015	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	99.4	99.7
1016	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	99.4	99.7
1017	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	99.7	99.7
1018	PANORAMIC CAMERA DOORS	0	99.0	3214 HOURS	0	99.7	99.7

Table 17-1

SISTEMAS DE GESTIÓN DE PREMIOS Y RECOMPENSAS

## SECTION 18

## SUMMARY DATA

The comparison of the operating parameters and the performance achieved by previous missions has been difficult due to the large volume of data that results from each mission. Some of the pertinent characteristics from prior missions have been summarized in Tables 18-1 through 18-3.

The summary data was started with Mission 1004 as the J-05 camera system was the first to incorporate the major modifications of the titanium drum and scan arm, four roller scan head and Corona J capabilities. Only those missions that culminated in the recovery of some photography have been listed, therefore Missions 1003 and 1005 are deleted.

# MISSION SUMMARY

TOP SECRET//SI

MISSION NUMBER	PAYOUT NUMBER	VEHICLE NUMBER	LAUNCH DATE	LAUNCH TIME	ORBIT INCLINATION (°)	PERIGEE LOCATION (°N)	ALTITUDE (NM)	RECOVERY PASS	MASTER CAMERA NUMBER	MASTER CAMERA SLIT FILTER TYPE	SLAVE CAMERA NUMBER	SLAVE CAMERA SLIT FILTER TYPE	STELLAR INDEX CAMERA NUMBER		
1004	J-05	1174	2/13/64	2150 Z	74.9	99.9	29.0	49	124	0.250	W-21	125	0.250	W-21	029/29/29
1006	J-09	1176	6/4/64	2259 Z	79.9	84.0	63.2	65	148	0.200	W-21	149	0.200	W-21	042/42/37
1007	J-07	1609	6/19/64	2310 Z	85.0	99.2	41.9	65	128	0.250	W-25	145	0.200	W-21	049/43/43
1008	J-10	1177	7/10/64	2354 Z	89.0	99.4	40.8	49	112	0.200	W-21	161	0.200	W-21	043/47/43
1009	J-12	1605	8/5/64	2310 Z	80.1	99.6	39.3	49	128	0.200	W-21	155	0.200	W-21	050/44/54
1010	J-11	1178	9/14/64	2254 Z	84.9	97.4	42.5	65	144	0.175	W-21	153	0.175	W-21	047/41/41
1011	J-3X	1170	10/5/64	2150 Z	79.9	99.3	20.9	65	—	0.175	W-21	161	0.175	W-21	030/30/30
1012	J-13	1179	10/17/64	2202 Z	75.0	96.2	35.4	49	—	0.200	W-21	157	0.200	W-21	037/37/37
1013	J-15	1173	11/2/64	2150 Z	80.0	100.0	25.0	65	158	0.225	W-21	159	0.225	W-21	052/49/55
1014	J-16	1180	11/18/64	2036 Z	70.0	103.2	65.6	61	145	0.250	W-25	139	0.175	W-21	053/59/49
1015	J-17	1607	12/19/64	2110 Z	74.9	96.7	21.5	61	175	0.250	W-25	141	0.175	W-21	061/61/61
1016	J-18	1608	1/15/65	2101 Z	74.9	99.4	30.2	61	159	0.250	W-25	133	0.175	W-21	055/55/50
1017	J-14	1611	2/25/65	2144 Z	75.0	97.2	25.9	61	145	0.250	W-25	165	0.175	W-21	021/21/21
1018	J-19	1612	3/21/65	2111 Z	96.0	100.2	40.3	66	99	0.250	W-25	123	0.175	W-21	020/20/20
															022/22/22

REC'D 4/18

Table 18-1

TOP SECRET//SI

# PERFORMANCE SUMMARY

PROSPECTIVE

MISSION NUMBER	CAMERA	SERIAL NUMBER	M-I-P VALUE	VISUAL RES	SLIT AVERAGE (μ)	SLIT AVERAGE (μ)	SLIT AVERAGE (μ)	SLIT AVERAGE (μ)	90% ATTITUDE ERROR (%)			90% ATTITUDE RATES (°/sec)			90% V/H ERROR (%)		90% CROSS TRACK	
									PITCH	ROLL	YAW	PITCH	ROLL	YAW	PITCH	ROLL	YAW	
1004-1	FWD	124	65	78	97	109	115	127	0.45	0.42	1.08	30.0	21.0	6.1	—	7.7	0.1	
1004-2	ART	125	65	88	350	43	117	124	0.50	0.50	0.91	44.0	30.0	29.0	4.9	6.6	0.6	
1004-3	FWD	125	73	76	85	113	92	95	0.74	0.50	—	—	—	—	—	—	—	
1005-1	FWD	148	50	76	74	350	43	84	0.41	0.42	1.14	28.0	27.8	15.4	13.0	6.7	0.7	
1005-2	FWD	149	50	85	64	90	320	84	0.49	0.40	1.08	31.1	27.9	30.0	11.6	10.1	7.0	
1007-1	FWD	144	65	85	350	43	67	82	0.58	0.46	1.43	37.6	23.9	23.9	5.6	3.1	0.4	
1007-2	FWD	145	65	79	350	61	61	320	68	0.64	0.47	—	43.0	25.8	—	4.6	2.1	0.6
1008-1	FWD	150	65	80	350	43	98	86	0.59	0.39	0.94	43.8	23.9	23.9	2.9	4.9	0.9	
1008-2	FWD	151	65	82	350	61	91	92	0.63	0.36	0.71	42.9	24.0	32.5	3.0	4.1	0.4	
1009-1	FWD	154	65	92	350	65	60	61	0.65	0.65	0.71	29.2	22.7	27.6	3.3	9.1	0.8	
1009-2	FWD	155	65	94	350	67	67	62	0.48	0.65	0.59	35.6	23.9	27.2	2.6	4.9	0.8	
1010-1	FWD	162	65	90	350	60	80	87	0.93	0.90	0.87	39.1	23.8	30.0	4.8	4.8	0.8	
1010-2	FWD	163	65	82	350	62	60	62	0.59	0.70	1.21	49.4	23.6	30.7	4.6	7.5	0.8	
1011-1	FWD	160	90	84	350	77	60	66	0.79	0.77	0.89	43.1	23.8	31.1	2.3	5.3	0.6	
1012-1	FWD	166	65	92	350	61	—	61	0.64	0.65	0.51	—	47.1	33.2	—	4.8	—	—
1012-2	FWD	157	65	65	350	61	67	60	0.64	0.67	0.77	48.2	30.7	40.4	5.8	6.3	0.6	
1013-1	FWD	158	65	77	350	61	—	60	0.64	0.64	0.51	—	47.1	33.2	—	4.8	—	—
1014-1	FWD	162	60	67	350	62	67	61	0.62	0.62	0.41	46.0	35.0	36.1	5.6	6.5	0.6	
1014-2	FWD	159	60	65	350	62	60	60	0.62	0.61	1.44	34.8	36.0	35.3	5.4	6.5	0.6	
1015-1	FWD	158	65	67	350	62	—	60	0.65	0.38	0.53	47.0	29.4	36.4	3.0	5.4	0.5	
1015-2	FWD	161	65	67	350	62	—	60	0.64	0.36	0.53	46.9	29.2	36.2	3.0	5.4	0.5	
1016-1	FWD	152	65	65	350	61	—	61	0.72	0.63	1.01	48.8	30.4	40.4	40.4	42.0	0.6	
1016-2	ART	153	65	65	350	61	—	60	0.65	0.63	2.01	48.4	30.1	42.2	39.9	41.5	0.6	
1017-1	FWD	140	65	65	350	61	—	60	0.75	0.69	2.19	42.2	37.3	39.9	2.3	7.1	—	
1017-2	ART	155	65	65	350	61	—	60	0.65	0.65	0.45	—	36.5	34.0	—	4.3	—	—
1018-1	FWD	152	65	65	350	61	—	60	0.61	0.61	0.48	47.4	36.7	36.7	3.4	5.7	—	
1018-2	ART	153	65	65	350	61	—	60	0.60	0.60	0.47	48.2	34.8	34.8	3.4	5.7	—	
* DATA NOT PRESENTLY AVAILABLE																		

## **EXPOSURE - PROCESSING**

## SUMMARY

~~TOP SECRET~~

SECTION A

APPENDIX

~~TOP SECRET~~

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MISSION • 1018-1 • INSTRUMENT • FWD      06-11-65      DENSITY FREQ DIST

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
0.01	0	0	0	0
0.02	0	0	0	0
0.03	0	0	0	0
0.04	0	0	0	0
0.05	0	0	0	0
0.06	0	0	0	0
0.07	0	0	0	0
0.08	0	0	0	0
0.09	0	0	0	0
0.10	0	0	0	0
0.11	0	0	0	0
0.12	0	0	0	0
0.13	0	0	0	0
0.14	0	0	0	0
0.15	0	0	0	0
0.16	0	0	0	0
0.17	0	0	0	0
0.18	0	0	0	0
0.19	0	0	0	0
0.20	0	0	0	0
0.21	0	0	0	0
0.22	0	0	0	0
0.23	0	0	0	0
0.24	0	0	0	0
0.25	0	0	0	0
0.26	0	0	0	0
0.27	0	0	0	0
0.28	0	0	0	0
0.29	0	0	0	0
0.30	0	0	0	0
0.31	0	0	0	0
0.32	0	0	0	0
0.33	0	0	0	0
0.34	0	0	0	0
0.35	0	0	0	0
0.36	0	0	0	0
0.37	0	0	0	0
0.38	0	0	0	0
0.39	0	0	0	0
0.40	0	0	0	0
0.41	0	0	0	0
0.42	0	0	0	0
0.43	0	0	0	0
0.44	0	0	0	0
0.45	0	0	0	0
0.46	0	0	0	0
0.47	0	0	0	0
0.48	0	0	0	0
0.49	0	0	0	0
0.50	0	0	0	0
SUBTOTAL	2	0	91	110
		0	0	0

~~TOP SECRET~~

~~TOP SECRET~~

MISSION • 1018-1 • INSTRUMENT • FWD

06-11-65

DENSITY FREQ DIST

DENSITY VALUE	PRIMARY			INTERMEDIATE			FULL			ALL LEVELS		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
0.51	0	0	0	2	0	0	1	1	0	0	0	0
0.52	0	0	0	11	0	0	1	1	0	0	0	0
0.53	0	0	0	0	4	0	0	0	0	0	0	0
0.54	0	0	0	3	0	0	0	0	0	0	0	0
0.55	0	0	0	0	2	0	0	0	0	0	0	0
0.56	0	0	0	0	1	0	0	0	0	0	0	0
0.57	0	0	0	0	0	0	0	0	0	0	0	0
0.58	0	0	0	0	0	0	0	0	0	0	0	0
0.59	0	0	0	0	0	0	0	0	0	0	0	0
0.60	0	0	0	0	0	0	0	0	0	0	0	0
0.61	0	0	0	0	0	0	0	0	0	0	0	0
0.62	0	0	0	0	0	0	0	0	0	0	0	0
0.63	0	0	0	0	0	0	0	0	0	0	0	0
0.64	0	0	0	0	0	0	0	0	0	0	0	0
0.65	0	0	0	0	0	0	0	0	0	0	0	0
0.66	0	0	0	0	0	0	0	0	0	0	0	0
0.67	0	0	0	0	0	0	0	0	0	0	0	0
0.68	0	0	0	0	0	0	0	0	0	0	0	0
0.69	0	0	0	0	0	0	0	0	0	0	0	0
0.70	0	0	0	0	0	0	0	0	0	0	0	0
0.71	0	0	0	0	0	0	0	0	0	0	0	0
0.72	0	0	0	0	0	0	0	0	0	0	0	0
0.73	0	0	0	0	0	0	0	0	0	0	0	0
0.74	0	0	0	0	0	0	0	0	0	0	0	0
0.75	0	0	0	0	0	0	0	0	0	0	0	0
0.76	0	0	0	0	0	0	0	0	0	0	0	0
0.77	0	0	0	0	0	0	0	0	0	0	0	0
0.78	0	0	0	0	0	0	0	0	0	0	0	0
0.79	0	0	0	0	0	0	0	0	0	0	0	0
0.80	0	0	0	0	0	0	0	0	0	0	0	0
0.81	0	0	0	0	0	0	0	0	0	0	0	0
0.82	0	0	0	0	0	0	0	0	0	0	0	0
0.83	0	0	0	0	0	0	0	0	0	0	0	0
0.84	0	0	0	0	0	0	0	0	0	0	0	0
0.85	0	0	0	0	0	0	0	0	0	0	0	0
0.86	0	0	0	0	0	0	0	0	0	0	0	0
0.87	0	0	0	0	0	0	0	0	0	0	0	0
0.88	0	0	0	0	0	0	0	0	0	0	0	0
0.89	0	0	0	0	0	0	0	0	0	0	0	0
0.90	0	0	0	0	0	0	0	0	0	0	0	0
0.91	0	0	0	0	0	0	0	0	0	0	0	0
0.92	0	0	0	0	0	0	0	0	0	0	0	0
0.93	0	0	0	0	0	0	0	0	0	0	0	0
0.94	0	0	0	0	0	0	0	0	0	0	0	0
0.95	0	0	0	0	0	0	0	0	0	0	0	0
0.96	0	0	0	0	0	0	0	0	0	0	0	0
0.97	0	0	0	0	0	0	0	0	0	0	0	0
0.98	0	0	0	0	0	0	0	0	0	0	0	0
0.99	0	0	0	0	0	0	0	0	0	0	0	0
1.00	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	106	0	0	10	0	0	20	3	13

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~~TOP SECRET~~

MISSION \* 1018-1 \* INSTRUMENT \* FWD 06-11-65 DENSITY FREQ DIST

DENSITY VALUE	PRIMARY			INTERMEDIATE			FULL			ALL LEVELS		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
1.01	0	0	0	0	0	0	0	0	0	0	0	0
1.02	0	0	0	0	0	0	0	0	0	0	0	0
1.03	0	0	0	0	0	0	0	0	0	0	0	0
1.04	0	0	0	0	0	0	0	0	0	0	0	0
1.05	0	0	0	0	0	0	0	0	0	0	0	0
1.06	0	0	0	0	0	0	0	0	0	0	0	0
1.07	0	0	0	0	0	0	0	0	0	0	0	0
1.08	0	0	0	0	0	0	0	0	0	0	0	0
1.09	0	0	0	0	0	0	0	0	0	0	0	0
1.10	0	0	0	0	0	0	0	0	0	0	0	0
1.11	0	0	0	0	0	0	0	0	0	0	0	0
1.12	0	0	0	0	0	0	0	0	0	0	0	0
1.13	0	0	0	0	0	0	0	0	0	0	0	0
1.14	0	0	0	0	0	0	0	0	0	0	0	0
1.15	0	0	0	0	0	0	0	0	0	0	0	0
1.16	0	0	0	0	0	0	0	0	0	0	0	0
1.17	0	0	0	0	0	0	0	0	0	0	0	0
1.18	0	0	0	0	0	0	0	0	0	0	0	0
1.19	0	0	0	0	0	0	0	0	0	0	0	0
1.20	0	0	0	0	0	0	0	0	0	0	0	0
1.21	0	0	0	0	0	0	0	0	0	0	0	0
1.22	0	0	0	0	0	0	0	0	0	0	0	0
1.23	0	0	0	0	0	0	0	0	0	0	0	0
1.24	0	0	0	0	0	0	0	0	0	0	0	0
1.25	0	0	0	0	0	0	0	0	0	0	0	0
1.26	0	0	0	0	0	0	0	0	0	0	0	0
1.27	0	0	0	0	0	0	0	0	0	0	0	0
1.28	0	0	0	0	0	0	0	0	0	0	0	0
1.29	0	0	0	0	0	0	0	0	0	0	0	0
1.30	0	0	0	0	0	0	0	0	0	0	0	0
1.31	0	0	0	0	0	0	0	0	0	0	0	0
1.32	0	0	0	0	0	0	0	0	0	0	0	0
1.33	0	0	0	0	0	0	0	0	0	0	0	0
1.34	0	0	0	0	0	0	0	0	0	0	0	0
1.35	0	0	0	0	0	0	0	0	0	0	0	0
1.36	0	0	0	0	0	0	0	0	0	0	0	0
1.37	0	0	0	0	0	0	0	0	0	0	0	0
1.38	0	0	0	0	0	0	0	0	0	0	0	0
1.39	0	0	0	0	0	0	0	0	0	0	0	0
1.40	0	0	0	0	0	0	0	0	0	0	0	0
1.41	0	0	0	0	0	0	0	0	0	0	0	0
1.42	0	0	0	0	0	0	0	0	0	0	0	0
1.43	0	0	0	0	0	0	0	0	0	0	0	0
1.44	0	0	0	0	0	0	0	0	0	0	0	0
1.45	0	0	0	0	0	0	0	0	0	0	0	0
1.46	0	0	0	0	0	0	0	0	0	0	0	0
1.47	0	0	0	0	0	0	0	0	0	0	0	0
1.48	0	0	0	0	0	0	0	0	0	0	0	0
1.49	0	0	0	0	0	0	0	0	0	0	0	0
1.50	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0	0	0	0	0

~~TOP SECRET~~

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MISSION • 1018-1 • INSTRUMENT • FWD      06-11-65      DENSITY FREQ DIST

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
1.51	0	0	0	0
1.52	0	0	0	0
1.53	0	0	0	0
1.54	0	0	0	0
1.55	0	0	0	0
1.56	0	0	0	0
1.57	0	0	0	0
1.58	0	0	0	0
1.59	0	0	0	0
1.60	0	0	0	0
1.61	0	0	0	0
1.62	0	0	0	0
1.63	0	0	0	0
1.64	0	0	0	0
1.65	0	0	0	0
1.66	0	0	0	0
1.67	0	0	0	0
1.68	0	0	0	0
1.69	0	0	0	0
1.70	0	0	0	0
1.71	0	0	0	0
1.72	0	0	0	0
1.73	0	0	0	0
1.74	0	0	0	0
1.75	0	0	0	0
1.76	0	0	0	0
1.77	0	0	0	0
1.78	0	0	0	0
1.79	0	0	0	0
1.80	0	0	0	0
1.81	0	0	0	0
1.82	0	0	0	0
1.83	0	0	0	0
1.84	0	0	0	0
1.85	0	0	0	0
1.86	0	0	0	0
1.87	0	0	0	0
1.88	0	0	0	0
1.89	0	0	0	0
1.90	0	0	0	0
1.91	0	0	0	0
1.92	0	0	0	0
1.93	0	0	0	0
1.94	0	0	0	0
1.95	0	0	0	0
1.96	0	0	0	0
1.97	0	0	0	0
1.98	0	0	0	0
1.99	0	0	0	0
2.00	0	0	0	0
SUBTOTAL	0	0	123	82
			21	10
			21	10
			146	93

~~TOP SECRET~~

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MISSION • 1018-1 • INSTRUMENT • FWD 06-11-65 DENSITY FREQ DIST

TOP SECRET

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MISSION • 1018-1 • INSTRUMENT • FWD      06-11-65      DENSITY FREQ DISTR

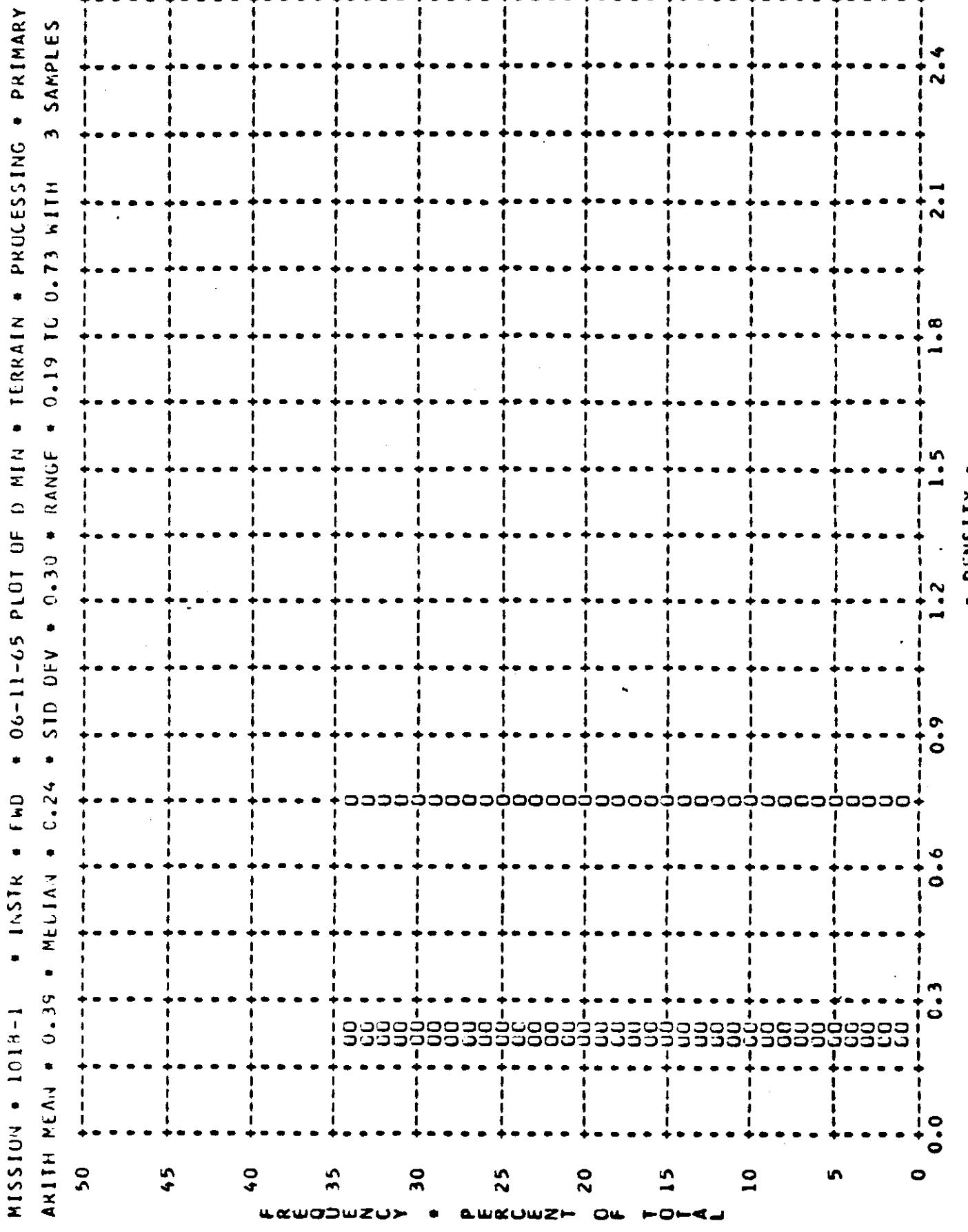
DENSITY VALUE	PRIMARY			INTERMEDIATE			FULL			ALL LEVELS		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
2.51	0	C	0	0	0	0	0	0	0	0	0	0
2.52	0	CC	0	0	0	0	0	0	0	0	0	0
2.53	0	CCC	0	0	0	0	0	0	0	0	0	0
2.54	0	CCCC	0	0	0	0	0	0	0	0	0	0
2.55	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.56	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.57	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.58	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.59	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.60	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.61	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.62	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.63	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.64	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.65	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.66	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.67	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.68	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.69	0	CCCCC	0	0	0	0	0	0	0	0	0	0
2.70	0	CCCCC	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	C	0	0	0	0	0	0	0	0	0	0
TOTAL	3	3	1	202	202	147	37	37	26	242	242	174

MISSION 1018-1      INSTR - FWD      06-11-65      PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	3	0 PC	67 PC	33 PC	0 PC	0 PC
INTERMEDIATE	202	0 PC	22 PC	69 PC	8 PC	0 PC
FULL	37	11 PC	0 PC	86 PC	3 PC	0 PC
ALL LEVELS	242	2 PC	19 PC	71 PC	7 PC	0 PC
PROCESS LEVEL	BASE + FOG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-C.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND UI
INTERMED	0.10-C.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UI
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND UI

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Figure A-1

MISSION • 1018-1 • INSTR • FWD • 06-11-65 PLOT OF D MAX • TERRAIN • PROCESSING • PRIMARY  
ARITH MEAN • 1.62 • MEDIAN • 1.67 • STD DEV • 0.18 • RANGE • 1.42 TO 1.76 WITH 3 SAMPLES

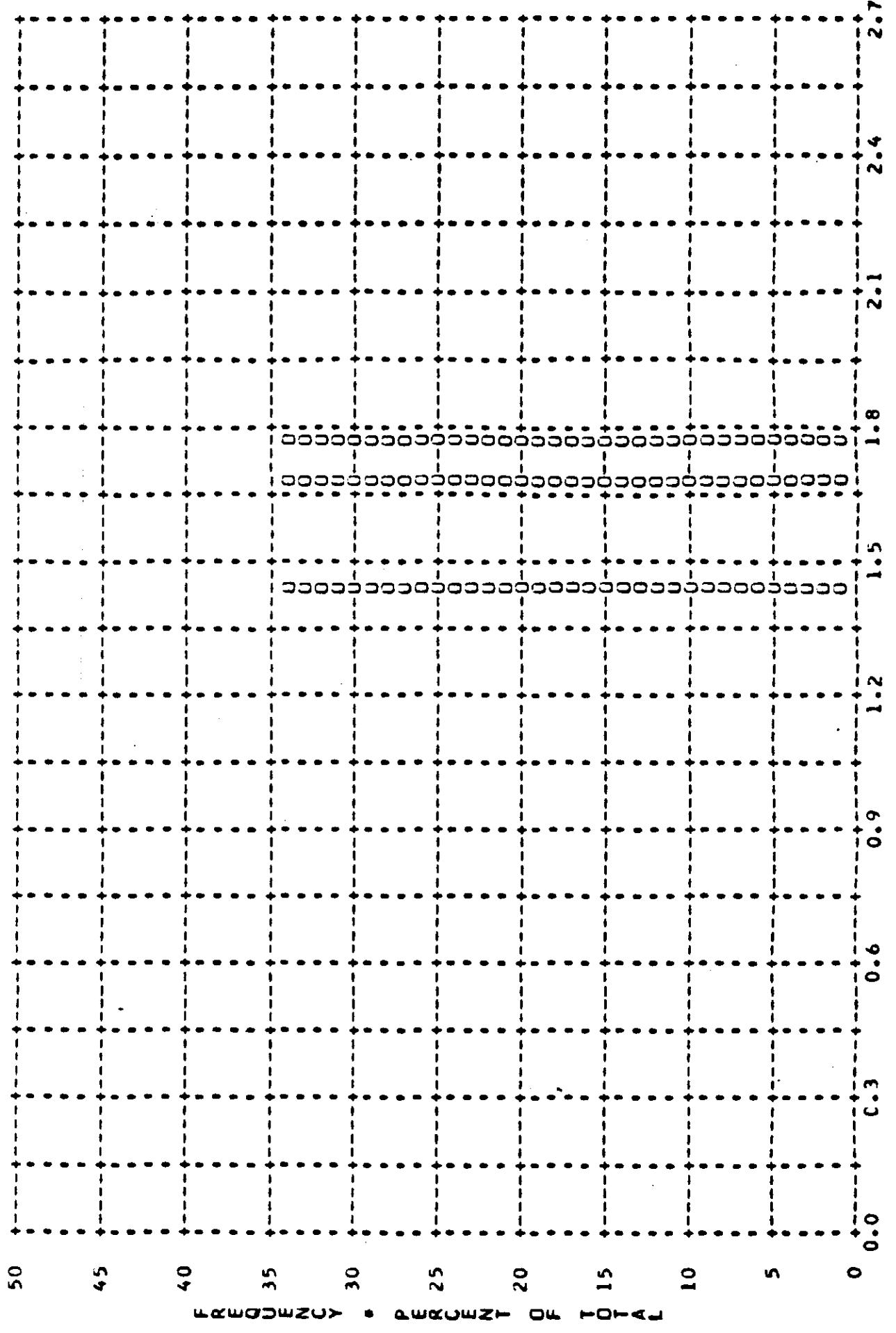


Figure A-2

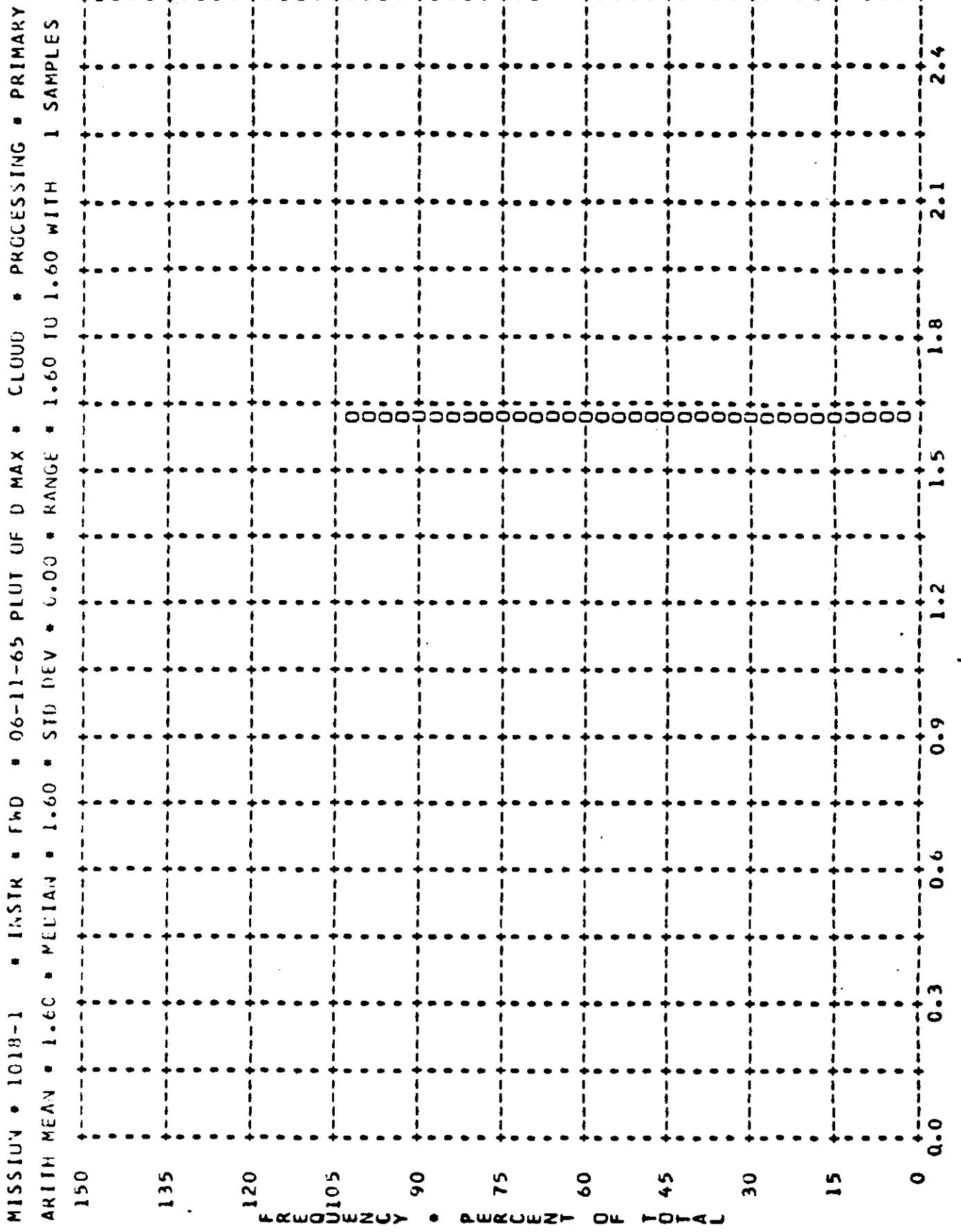


Figure A-3

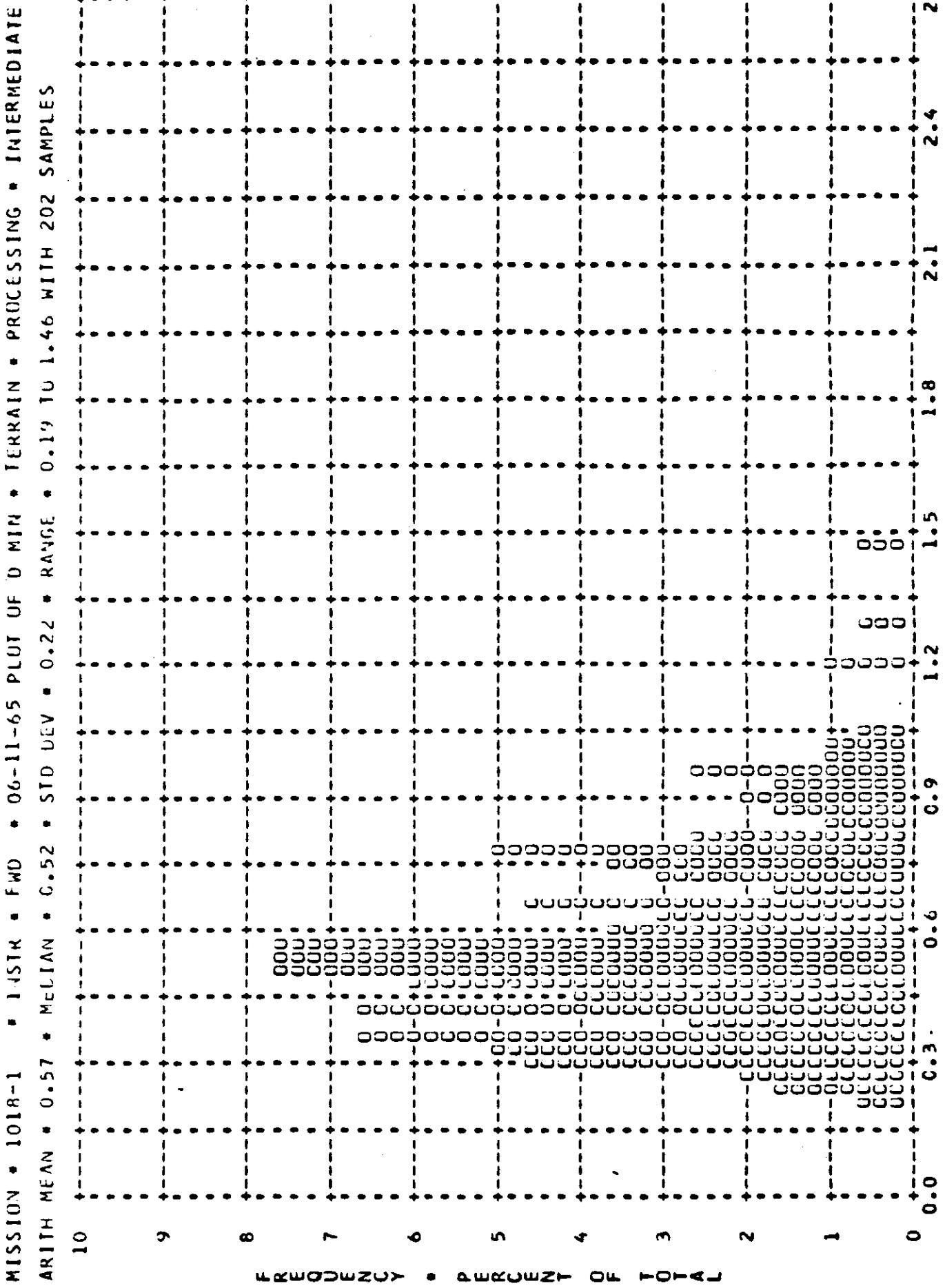
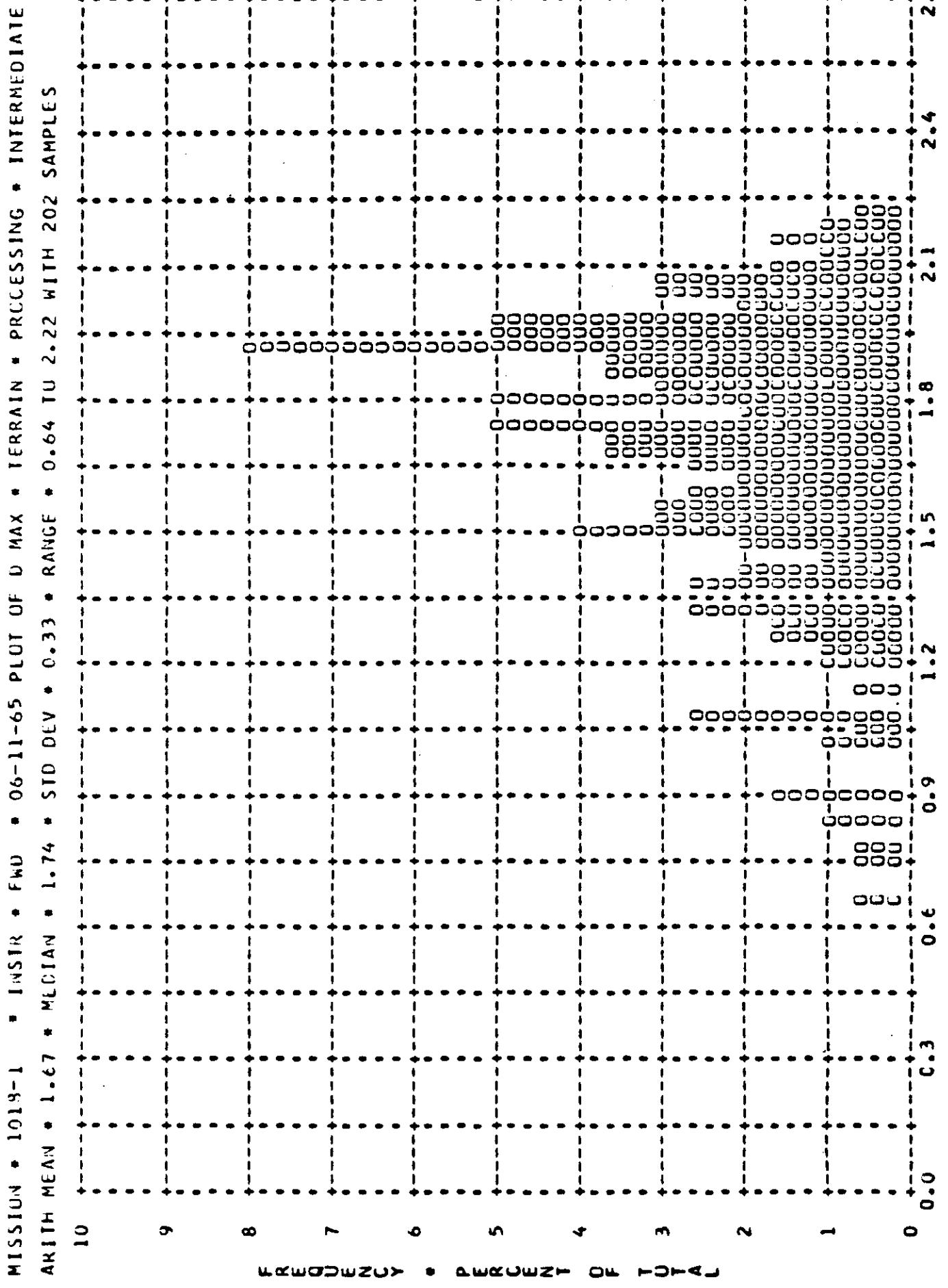


Figure A-4



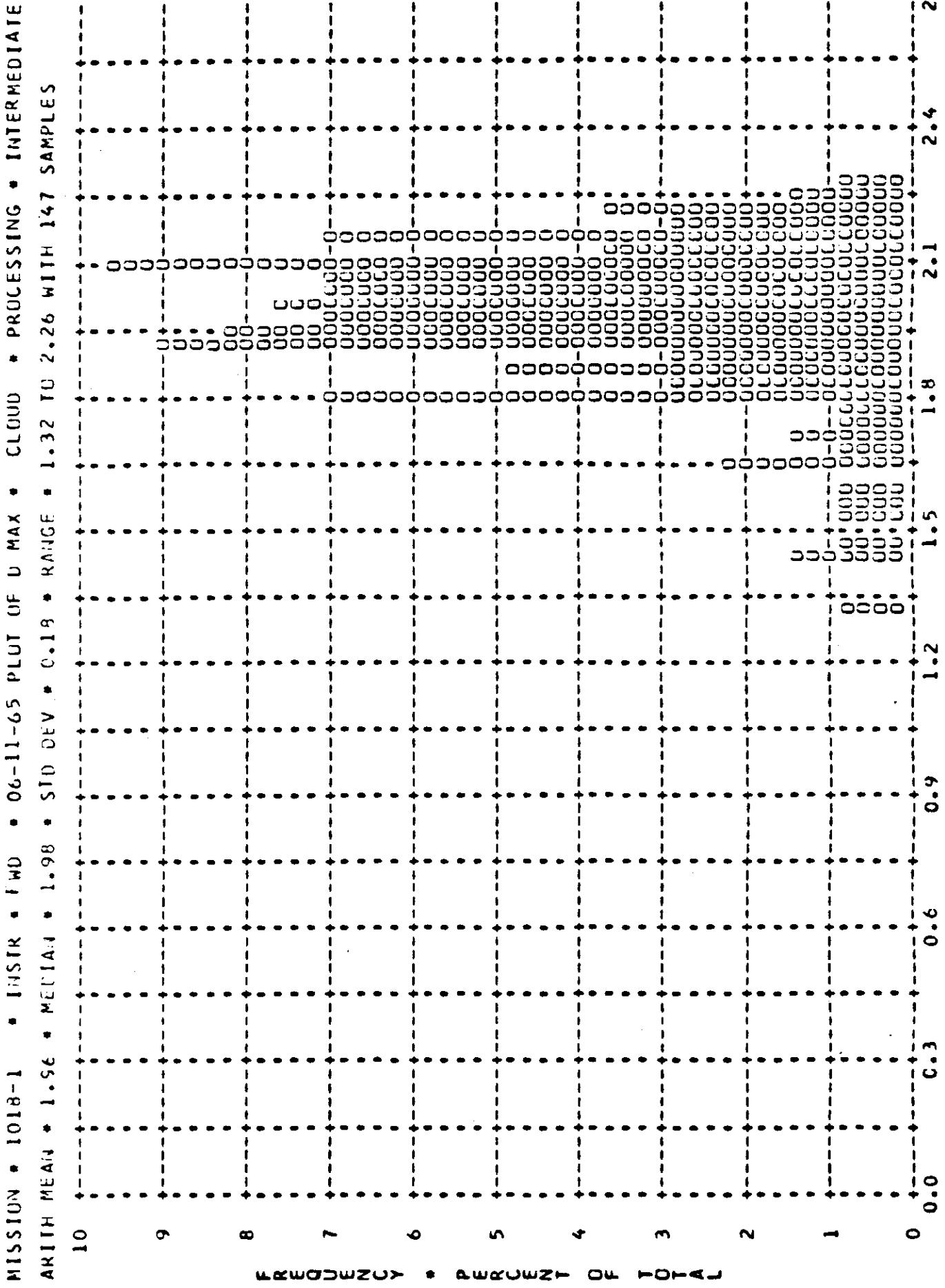
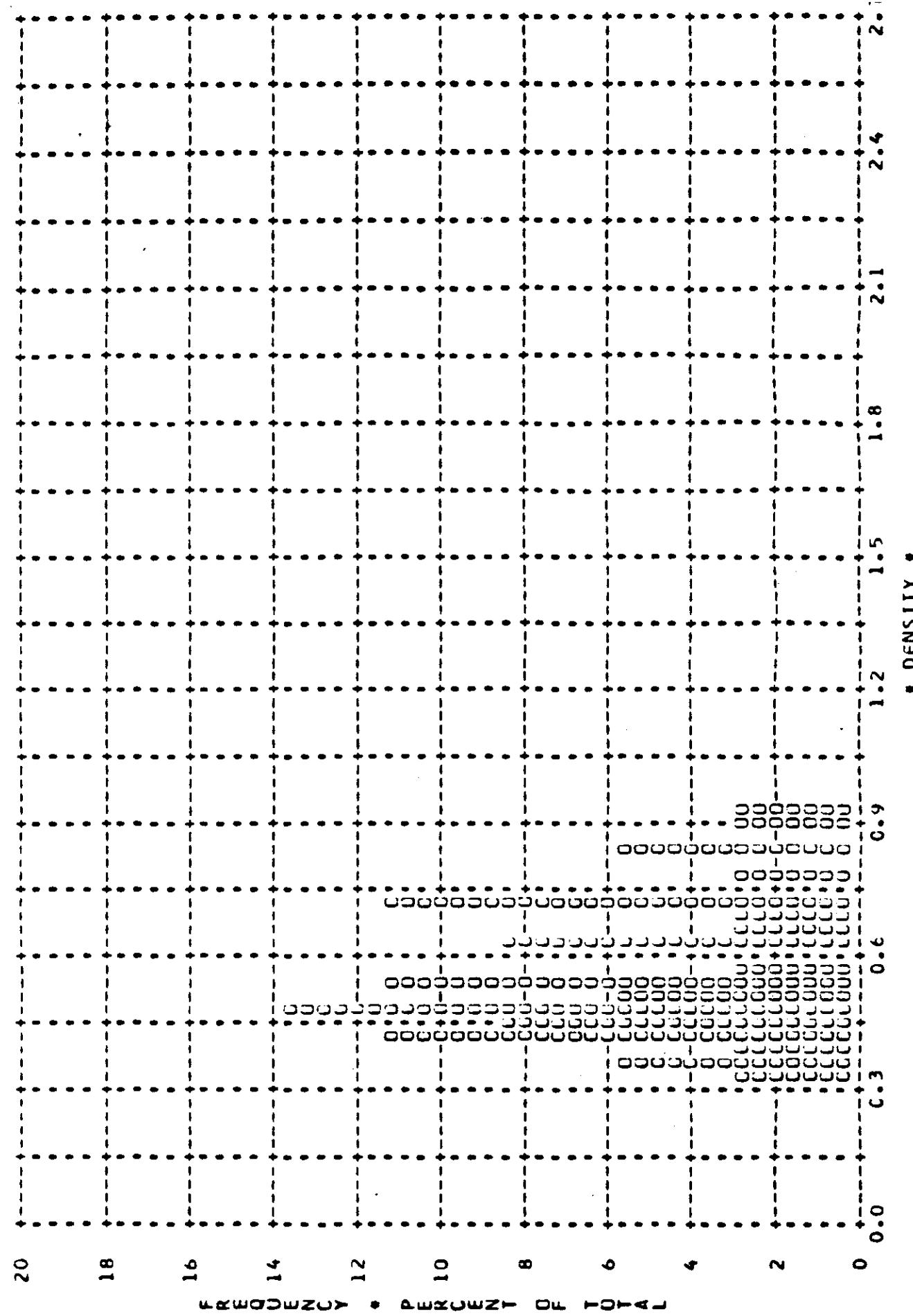


Figure A-1

MISSION \* 101W-1 \* INSTR \* FWD \* 06-11-65 PLOT OF D MIN \* TERRAIN \* PROCESSING \* FULL  
ARITH MEAN \* 0.56 \* MEDIAN \* C.52 \* STD DEV \* C.16 \* RANGE \* 0.33 TO 0.92 WITH 37 SAMPLES



MISSION • 1018-1 • INSTR • FWD • INSTR • FWD • 06-11-65 PLOT OF D MAX • TERRAIN • PROCESSING • FULL  
ARITH MEAN • 1.62 • MEDIAN • 1.68 • STD Dev • 0.35 • RANG • 0.94 TC 2.14 WITH 37 SAMPLES

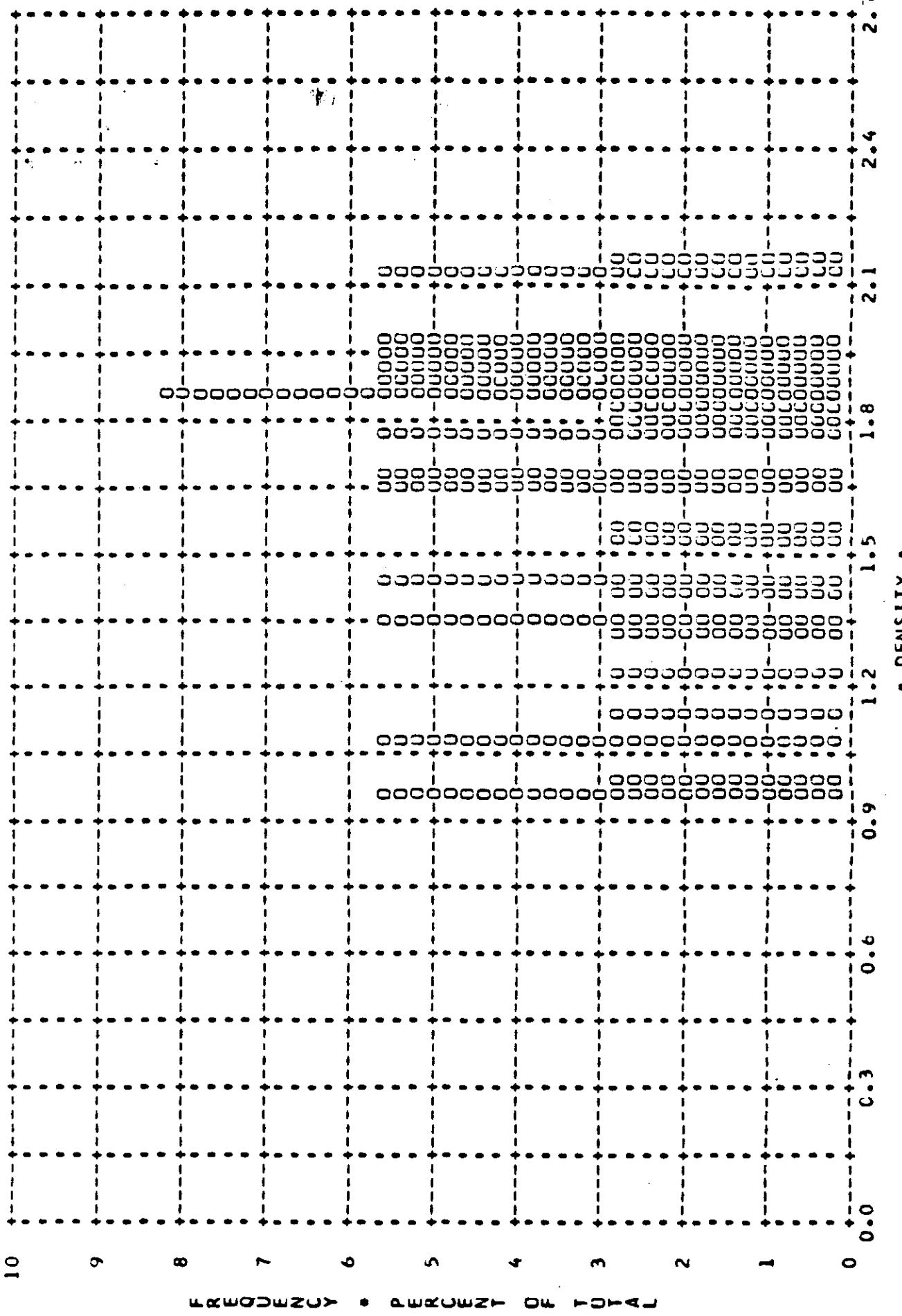


FIGURE A-2C

FIGURE A-2B

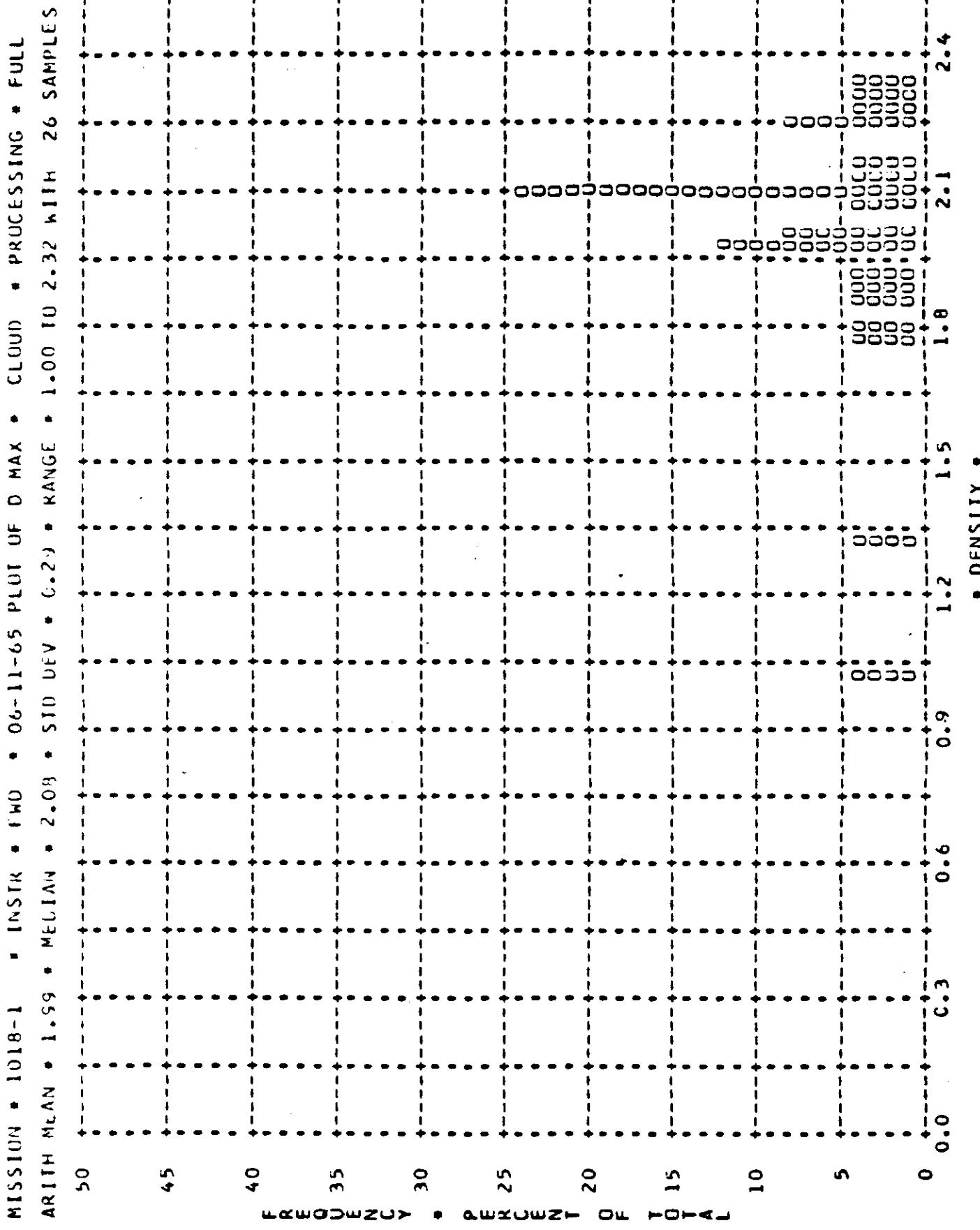


FIGURE 2-10

MISSION \* 1018-1 \* INSTR \* FRWD \* 06-11-65 PLOT OF D MIN \* TERRAIN \* PROCESSING \* ALL LEVELS  
ARITH MEAN \* 0.56 \* MEDIAN \* 0.52 \* STD DEV \* 0.21 \* RANGE \* 0.19 TO 1.46 WITH 242 SAMPLES

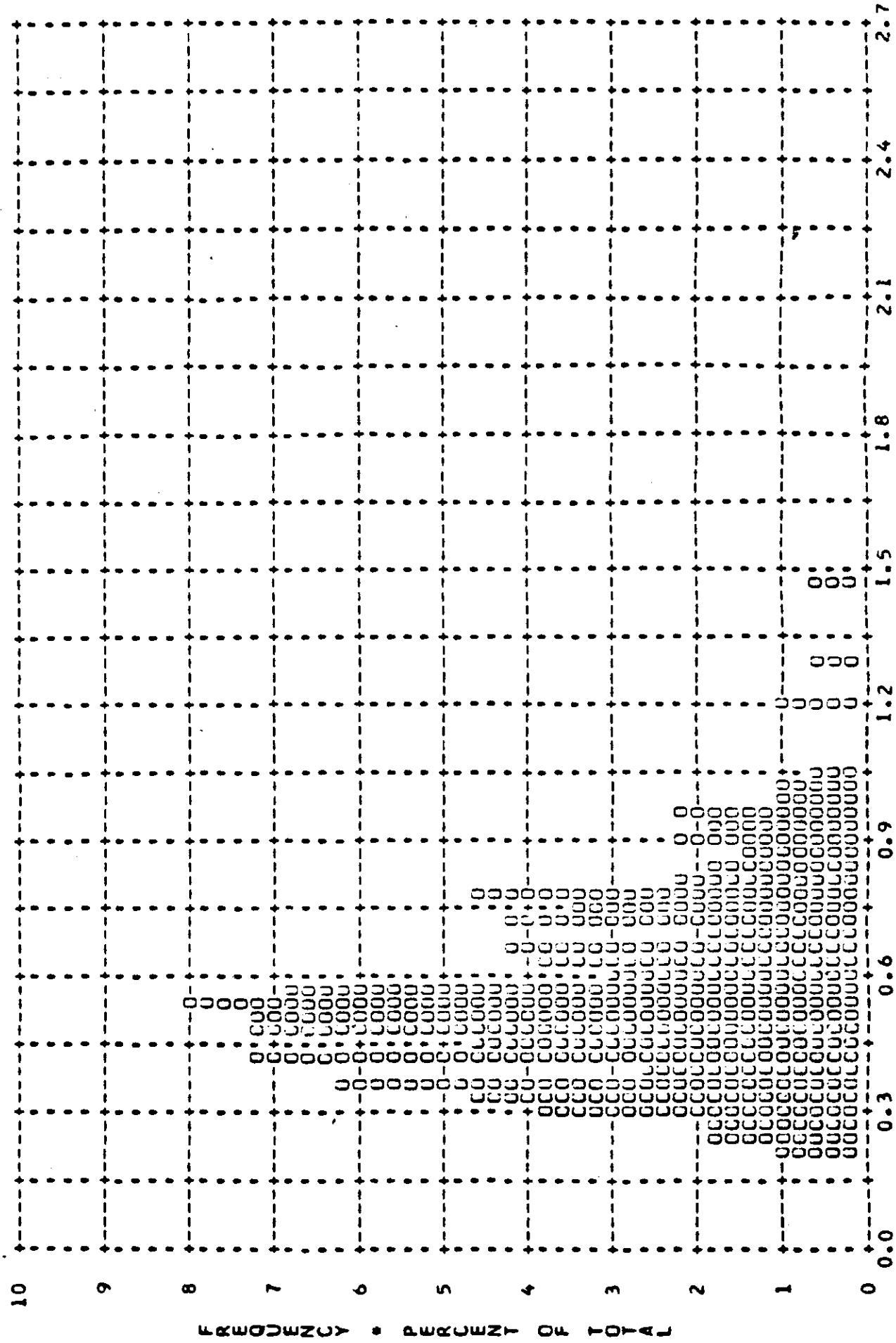
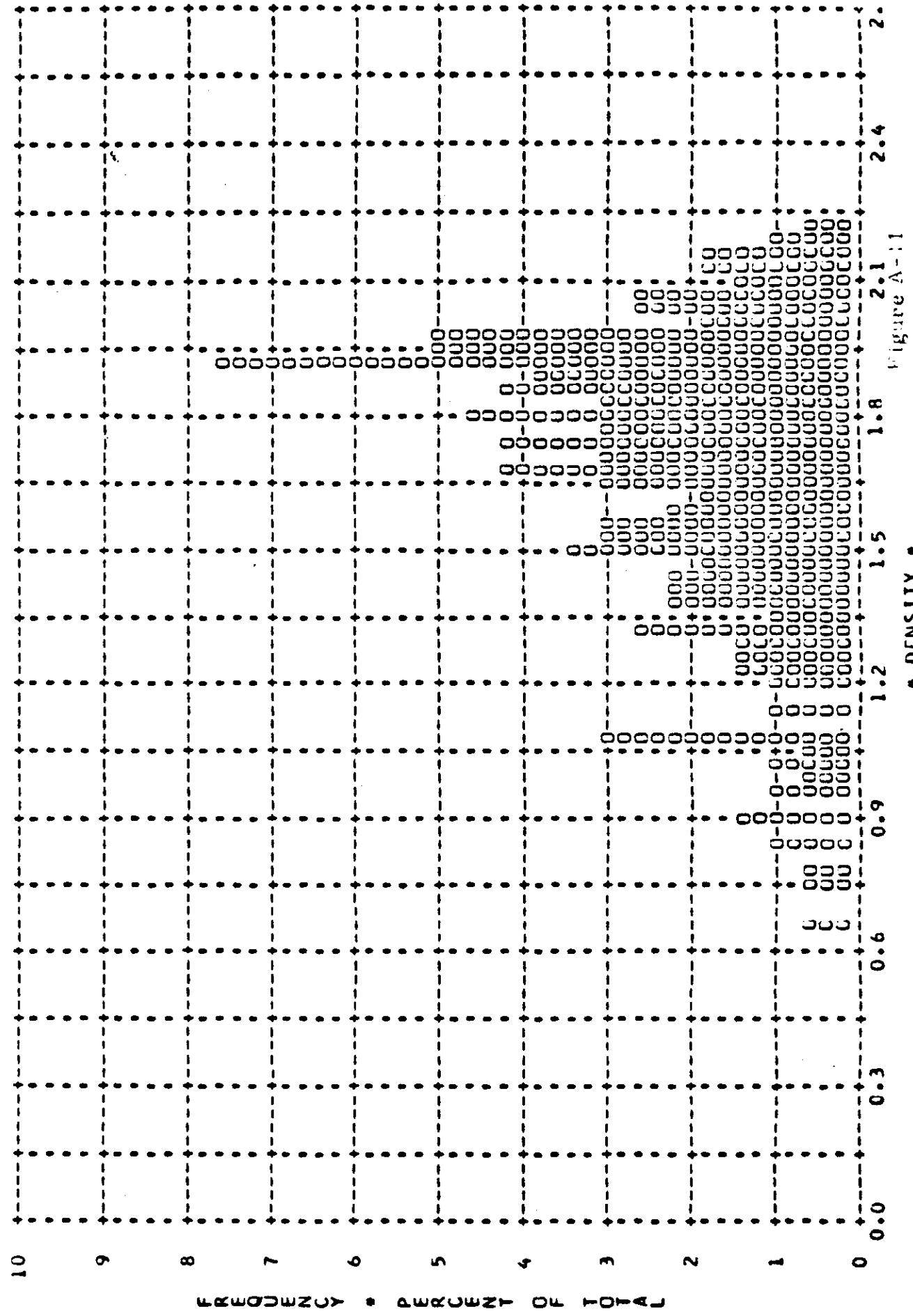


Figure A-10

MISSION • 1018-1 • INSTR • FWD • 06-11-65 PLUT OF D MAX • TERRAIN • PROCESSING • ALL LEVELS  
ARITH MEAN • 1.66 • MEDIAN • 1.73 • STD DEV • 0.33 • RANGE • 0.64 TO 2.22 WITH 242 SAMPLES



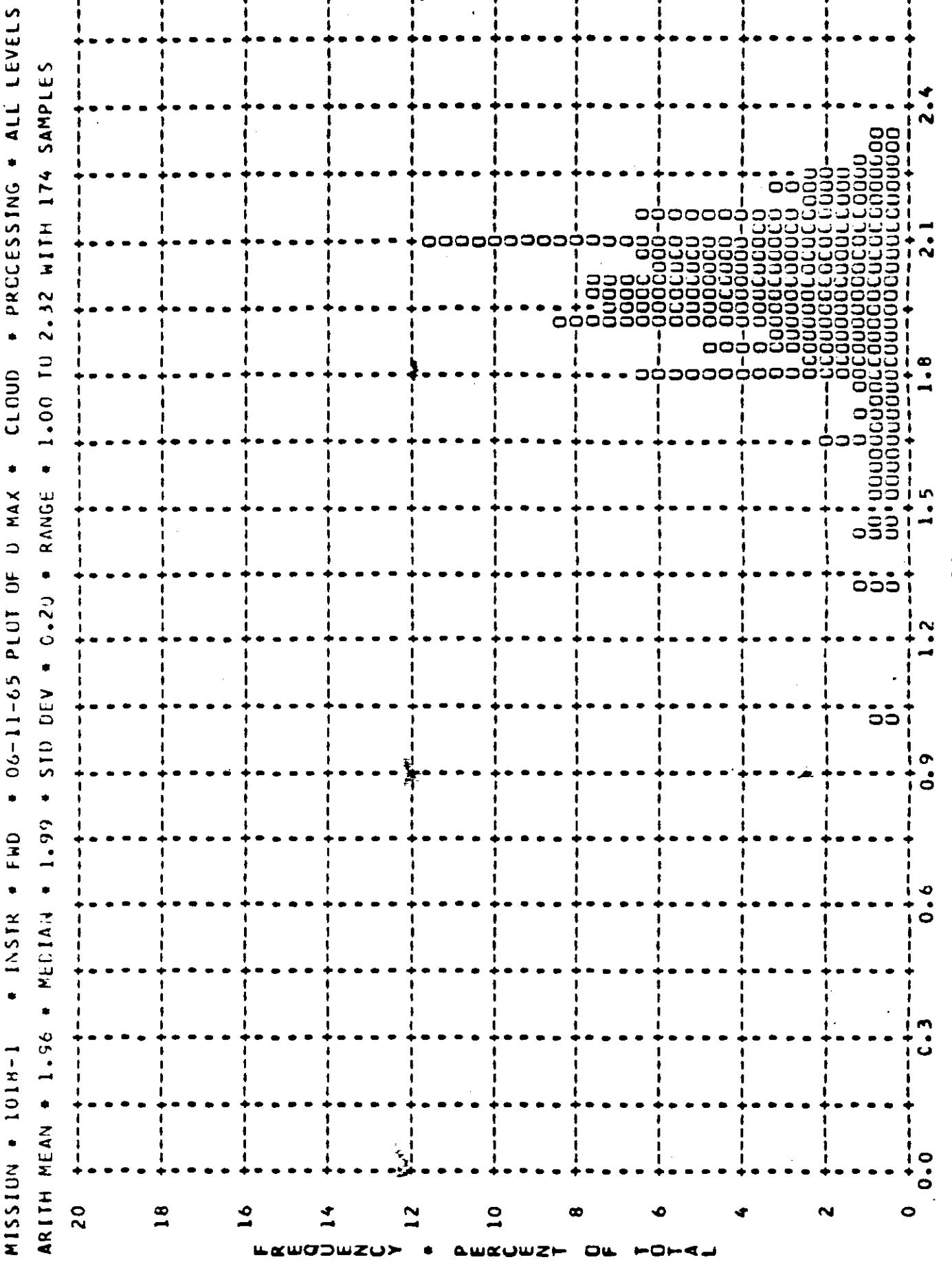


Figure A-1c

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MISSION \* 1018-1 \* INSTRUMENT \* AFT 06-11-65 DENSITY FREQ DISTR

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MISSION • 1018-1 • INSTRUMENT • AFT 06-11-65 DENSITY FREQ DISTR

100-XXXXXX

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MISSION • 1018-1 • INSTRUMENT • AFT      06-11-65      DENSITY FREQ DISTR.

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
1.01		0	0	00
1.02			00	1000
1.03			00	1100
1.04			00	1110
1.05			00	11110
1.06			00	111110
1.07			00	1111110
1.08			00	11111110
1.09			00	111111110
1.10			00	1111111110
1.11			00	11111111110
1.12			00	111111111110
1.13			00	1111111111110
1.14			00	11111111111110
1.15			00	111111111111110
1.16			00	1111111111111110
1.17			00	11111111111111110
1.18			00	111111111111111110
1.19			00	1111111111111111110
1.20			00	11111111111111111110
1.21			00	111111111111111111110
1.22			00	1111111111111111111110
1.23			00	11111111111111111111110
1.24			00	111111111111111111111110
1.25			00	1111111111111111111111110
1.26			00	11111111111111111111111110
1.27			00	111111111111111111111111110
1.28			00	1111111111111111111111111110
1.29			00	11111111111111111111111111110
1.30			00	111111111111111111111111111110
1.31		1	1	11
1.32				
1.33				
1.34				
1.35				
1.36				
1.37				
1.38				
1.39				
1.40				
1.41				
1.42				
1.43				
1.44				
1.45				
1.46				
1.47				
1.48				
1.49				
1.50				
SUBTOTAL	2	10	1	10
	12	45	7	68
			1	15
			2	13

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MISSION \* 1018-1 \* INSTRUMENT \* AFT 06-11-65 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
1.51	0 0	0 0	0 0	0 0
1.52	0 0	0 0	0 0	0 0
1.53	0 0	0 0	0 0	0 0
1.54	0 0	0 0	0 0	0 0
1.55	0 0	0 0	0 0	0 0
1.56	0 0	0 0	0 0	0 0
1.57	0 0	0 0	0 0	0 0
1.58	0 0	0 0	0 0	0 0
1.59	0 0	0 0	0 0	0 0
1.60	0 0	0 0	0 0	0 0
1.61	0 0	0 0	0 0	0 0
1.62	0 0	0 0	0 0	0 0
1.63	0 0	0 0	0 0	0 0
1.64	0 0	0 0	0 0	0 0
1.65	0 0	0 0	0 0	0 0
1.66	0 0	0 0	0 0	0 0
1.67	0 0	0 0	0 0	0 0
1.68	0 0	0 0	0 0	0 0
1.69	0 0	0 0	0 0	0 0
1.70	0 0	0 0	0 0	0 0
1.71	0 0	0 0	0 0	0 0
1.72	0 0	0 0	0 0	0 0
1.73	0 0	0 0	0 0	0 0
1.74	0 0	0 0	0 0	0 0
1.75	0 0	0 0	0 0	0 0
1.76	0 0	0 0	0 0	0 0
1.77	0 0	0 0	0 0	0 0
1.78	0 0	0 0	0 0	0 0
1.79	0 0	0 0	0 0	0 0
1.80	0 0	0 0	0 0	0 0
1.81	0 0	0 0	0 0	0 0
1.82	0 0	0 0	0 0	0 0
1.83	0 0	0 0	0 0	0 0
1.84	0 0	0 0	0 0	0 0
1.85	0 0	0 0	0 0	0 0
1.86	0 0	0 0	0 0	0 0
1.87	0 0	0 0	0 0	0 0
1.88	0 0	0 0	0 0	0 0
1.89	0 0	0 0	0 0	0 0
1.90	0 0	0 0	0 0	0 0
1.91	0 0	0 0	0 0	0 0
1.92	0 0	0 0	0 0	0 0
1.93	0 0	0 0	0 0	0 0
1.94	0 0	0 0	0 0	0 0
1.95	0 0	0 0	0 0	0 0
1.96	0 0	0 0	0 0	0 0
1.97	0 0	0 0	0 0	0 0
1.98	0 0	0 0	0 0	0 0
1.99	0 0	0 0	0 0	0 0
2.00	0 0	0 0	0 0	0 0
SUBTOTAL	0 4	0 100	0 68	0 35
				0 135 .79

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MISSION • 1018-1 • INSTRUMENT • AFT      06-11-65      DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
2.01	0	0	0	0
2.02	0	0	0	0
2.03	0	0	0	0
2.04	0	0	0	0
2.05	0	0	0	0
2.06	0	0	0	0
2.07	0	0	0	0
2.08	0	0	0	0
2.09	0	0	0	0
2.10	0	0	0	0
2.11	0	0	0	0
2.12	0	0	0	0
2.13	0	0	0	0
2.14	0	0	0	0
2.15	0	0	0	0
2.16	0	0	0	0
2.17	0	0	0	0
2.18	0	0	0	0
2.19	0	0	0	0
2.20	0	0	0	0
2.21	0	0	0	0
2.22	0	0	0	0
2.23	0	0	0	0
2.24	0	0	0	0
2.25	0	0	0	0
2.26	0	0	0	0
2.27	0	0	0	0
2.28	0	0	0	0
2.29	0	0	0	0
2.30	0	0	0	0
2.31	0	0	0	0
2.32	0	0	0	0
2.33	0	0	0	0
2.34	0	0	0	0
2.35	0	0	0	0
2.36	0	0	0	0
2.37	0	0	0	0
2.38	0	0	0	0
2.39	0	0	0	0
2.40	0	0	0	0
2.41	0	0	0	0
2.42	0	0	0	0
2.43	0	0	0	0
2.44	0	0	0	0
2.45	0	0	0	0
2.46	0	0	0	0
2.47	0	0	0	0
2.48	0	0	0	0
2.49	0	0	0	0
2.50	0	0	0	0
SUBTOTAL	0	0	0	0
	27	59	4	25
			31	85

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MISSION • 1018-1 • INSTRUMENT • AFT      06-11-65      DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY			INTERMEDIATE			FULL			ALL LEVELS		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
2.51	C	C	0	C	C	0	0	0	0	0	0	0
2.52	CCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.53	CCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.54	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.55	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.56	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.57	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.58	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.59	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.60	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.61	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.62	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.63	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.64	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.65	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.66	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.67	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.68	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.69	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
2.70	CCCCC	CCC	00	CCC	CCC	00	00	00	00	00	00	00
SUBTOTAL	C	C	0	C	C	0	0	0	0	0	0	0
TOTAL	11	11	6	176	176	134	56	56	34	243	243	174

MISSION 1018-1      INSTR - AFT      06-11-65      PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDEK EXPCSEC	UNDER PROCESSED		CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED	
			UNDEK EXPCSED	UNDER PROCESSED				
PRIMARY	11	0 PC	36	PC	36	PC	0 PC	27 PC
INTERMEDIATE	176	1 PC	14	PC	75	PC	10 PC	1 PC
FULL	56	4 PC	0	PC	89	PC	7 PC	0 PC
ALL LEVELS	243	1 PC	12	PC	77	PC	9 PC	2 PC
PROCESS LEVEL	BASE + FOG	UNDEK EXPCSED	UNDER PROCESSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED	
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	-----	0.91 AND UP	
INTERMED	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UP	1.70 AND UP	
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	-----	-----	

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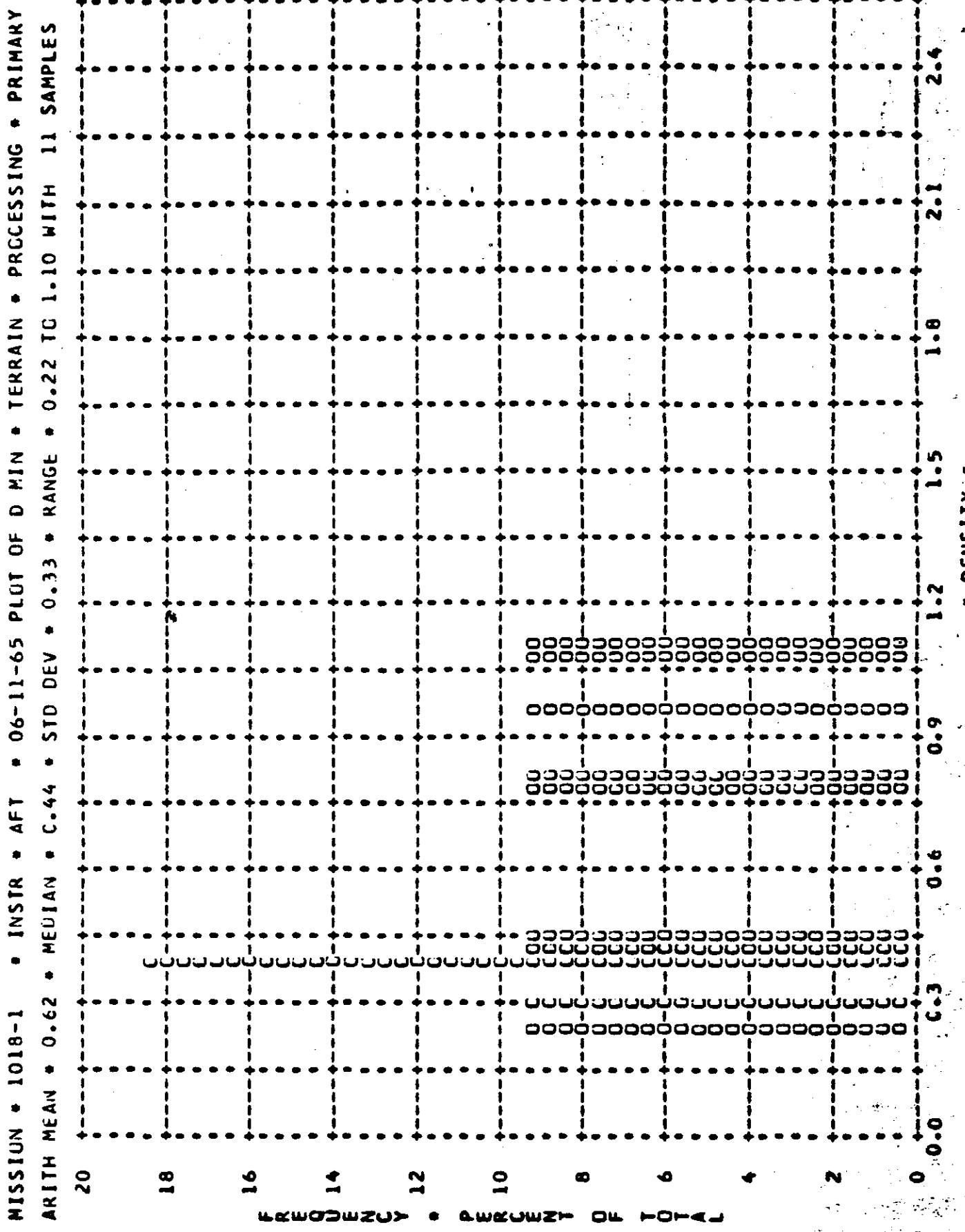
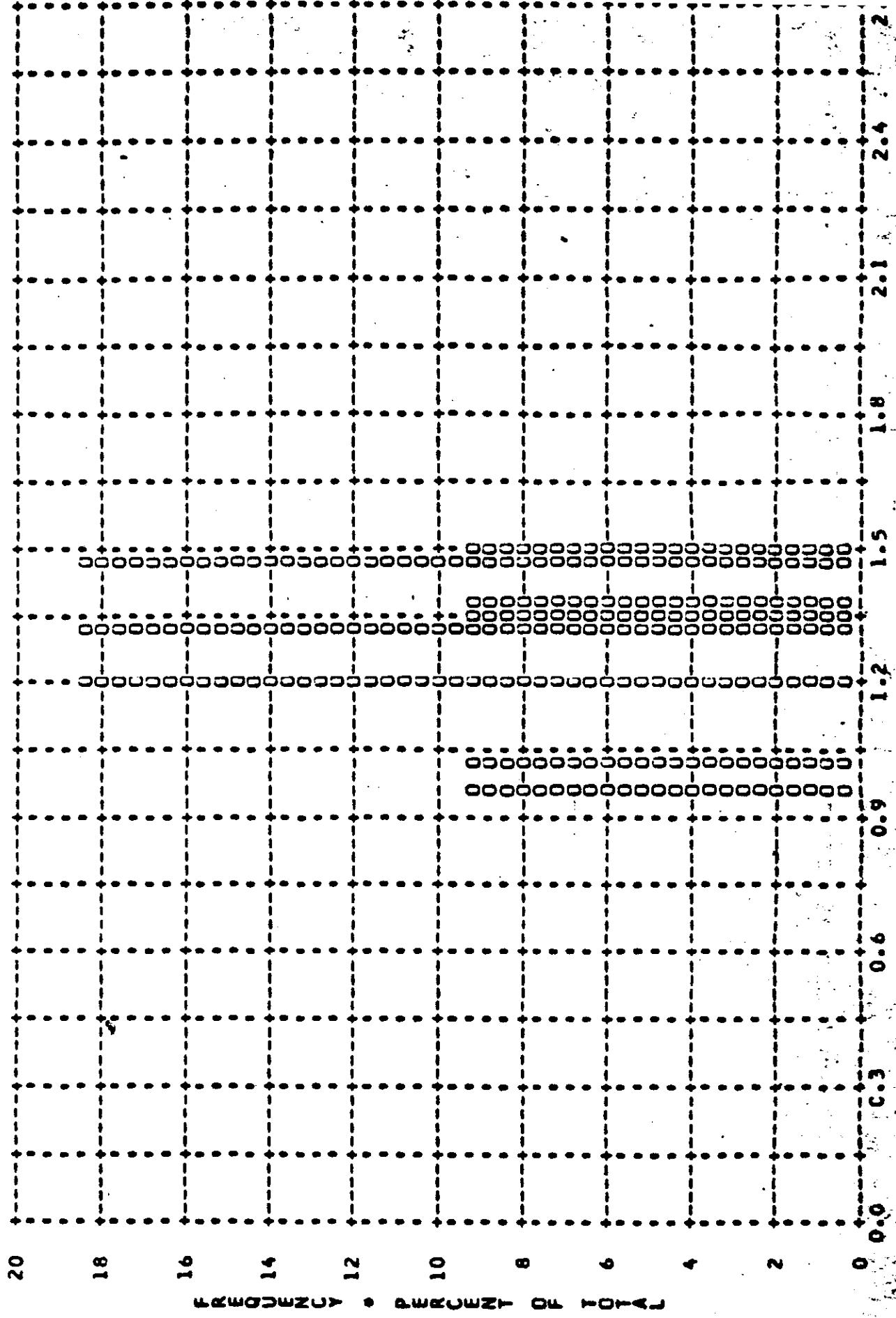


Figure A-13

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MISSION \* 1018-1 \* INSTR \* AFT \* 06-11-65 PLUT OF D MAX \* TERRAIN \* PROCESSING \* PRIMARY  
ARITH MEAN \* 1.28 \* MEDIAN \* 1.32 \* STD DEV \* 0.18 \* RANGE \* 0.94 TO 1.48 WITH 11 SAMPLES



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MISSION \* 1018-1 \* INSTR \* AFT \* 06-11-65 PLOT OF D MAX \* CLOUD \* PROCESSING \* PRIMARY  
ARITH MEAN \* 1.83 \* MEDIAN \* 1.90 \* STD DEV \* 0.24 \* RANGE \* 1.44 TO 2.12 WITH 6 SAMPLES

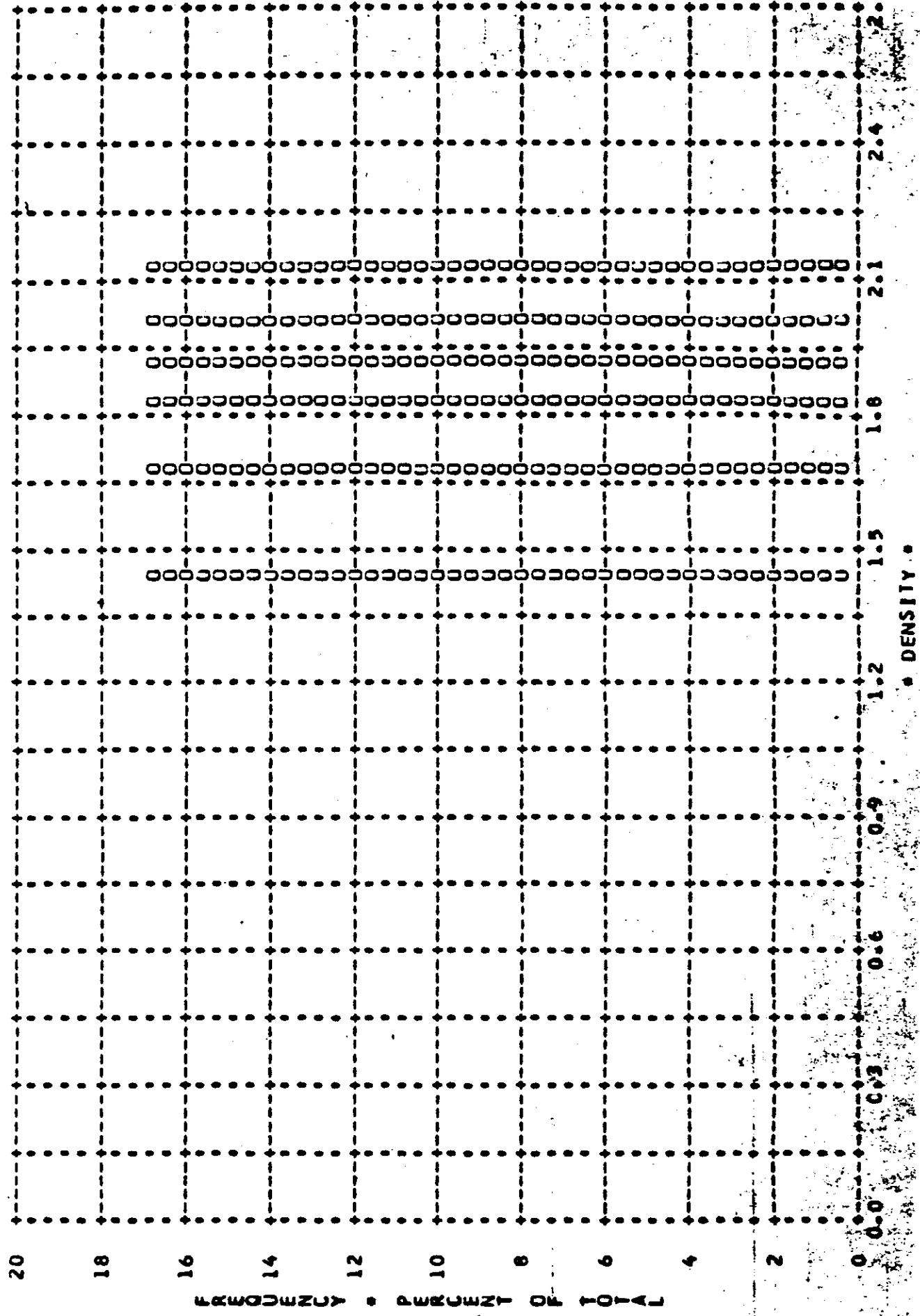
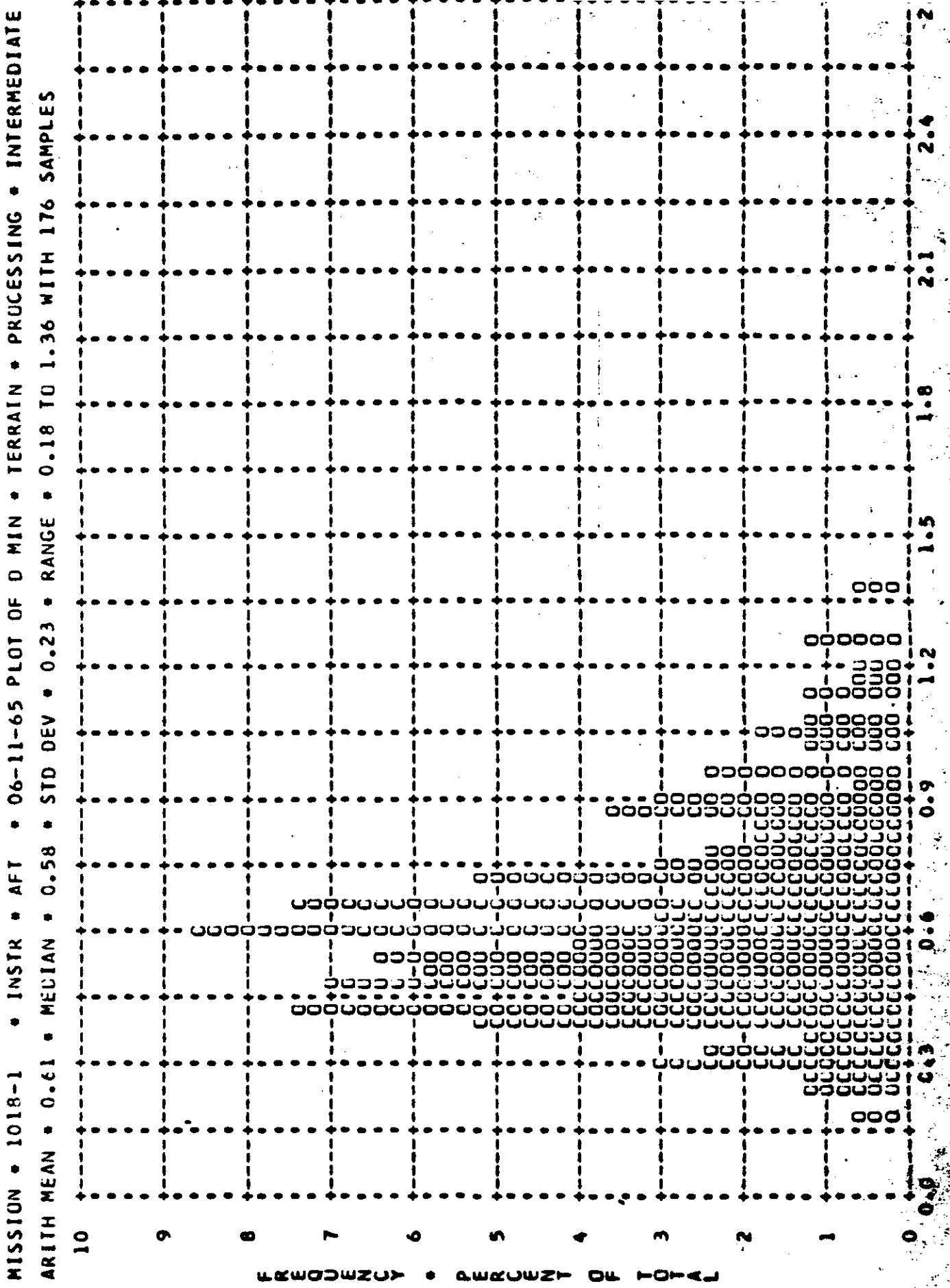
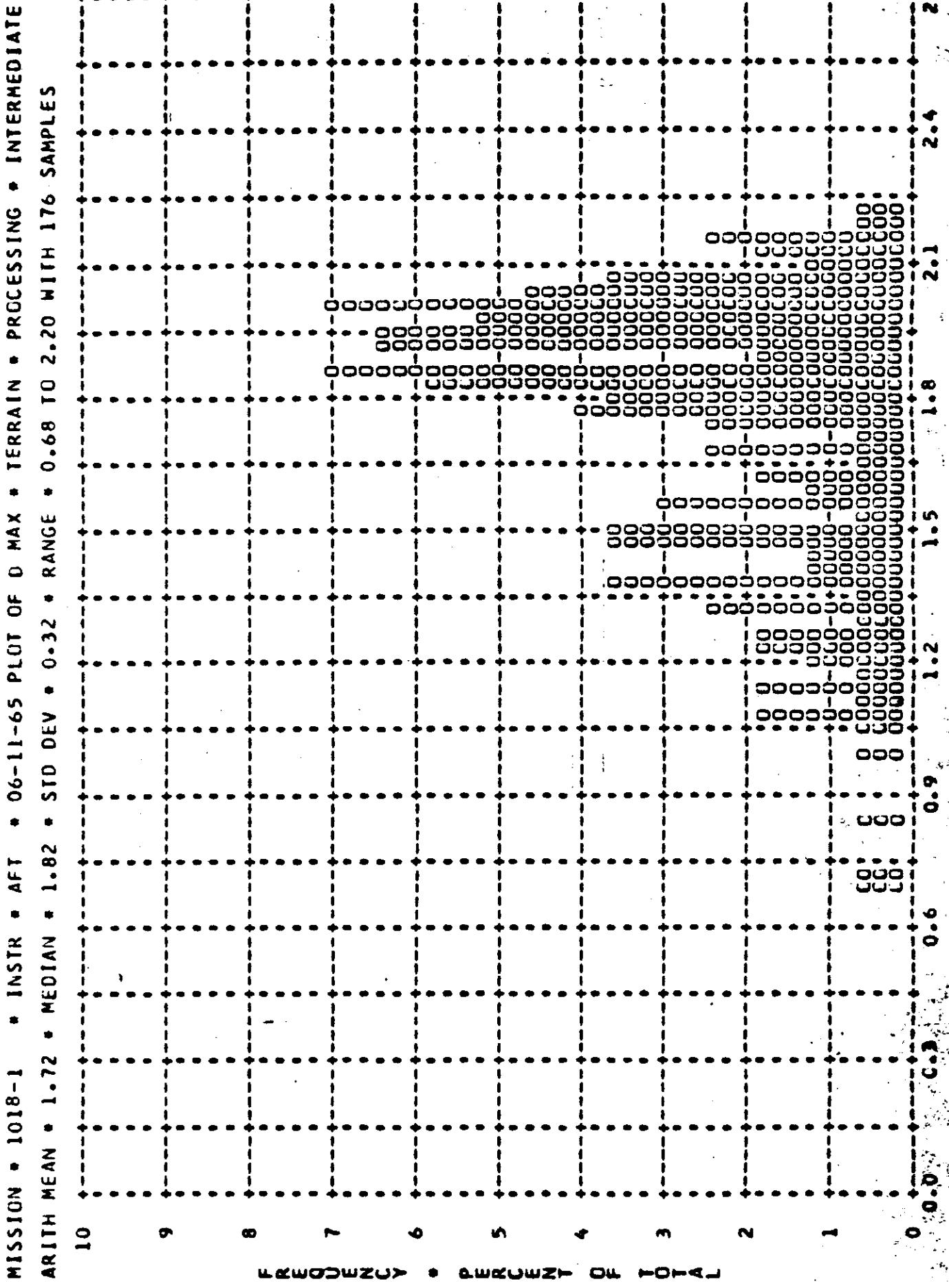


Figure A-16



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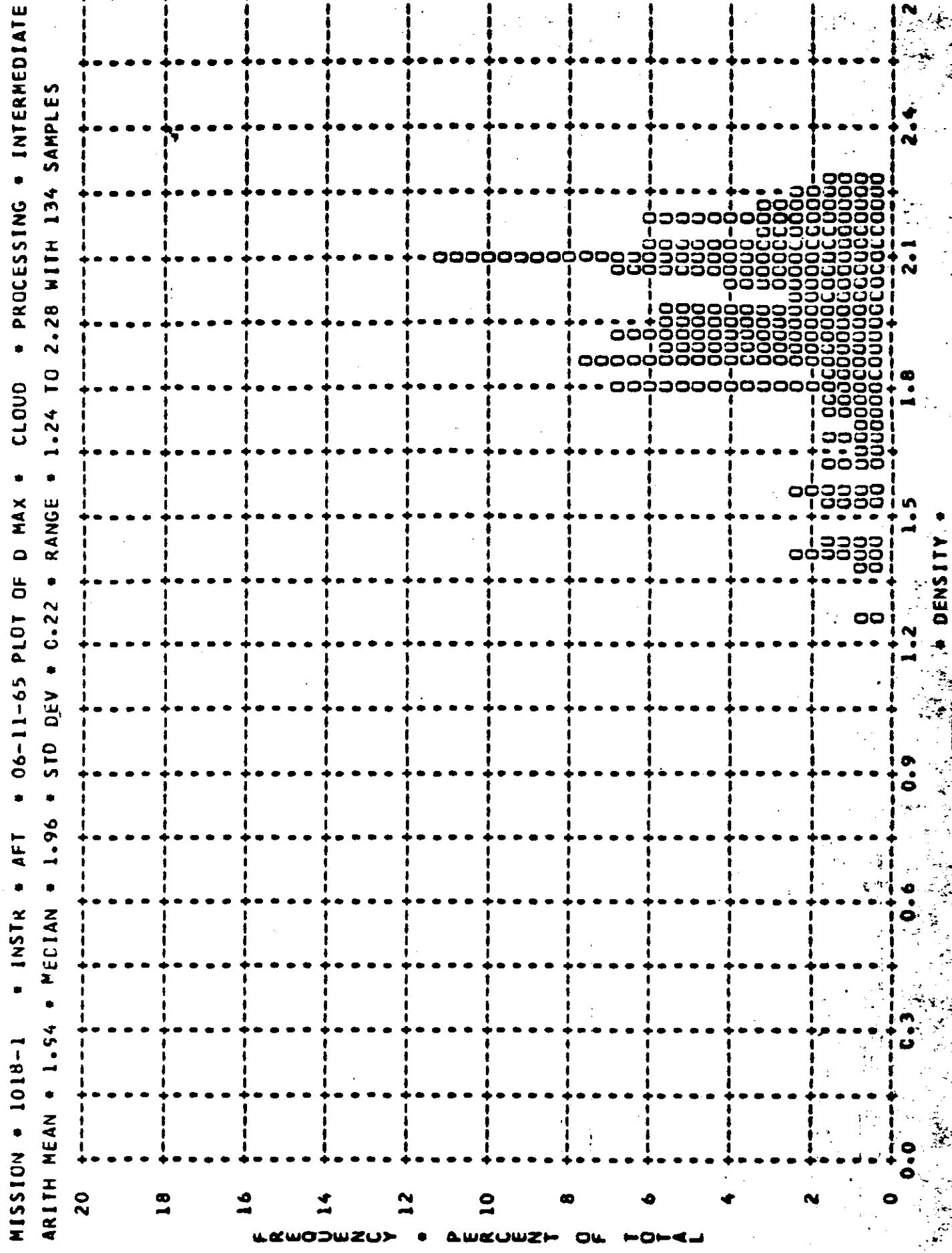


A-29

Figure A-17

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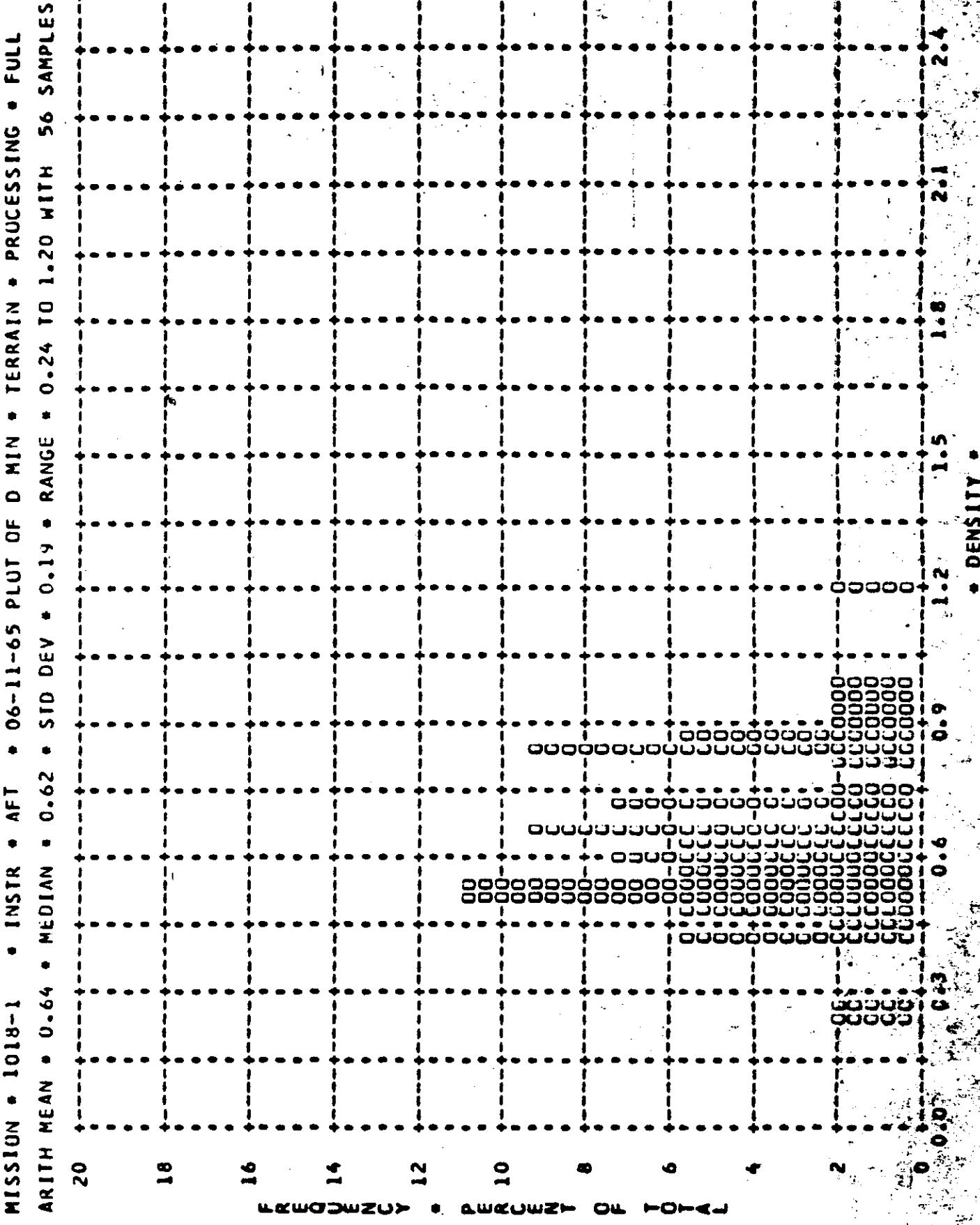
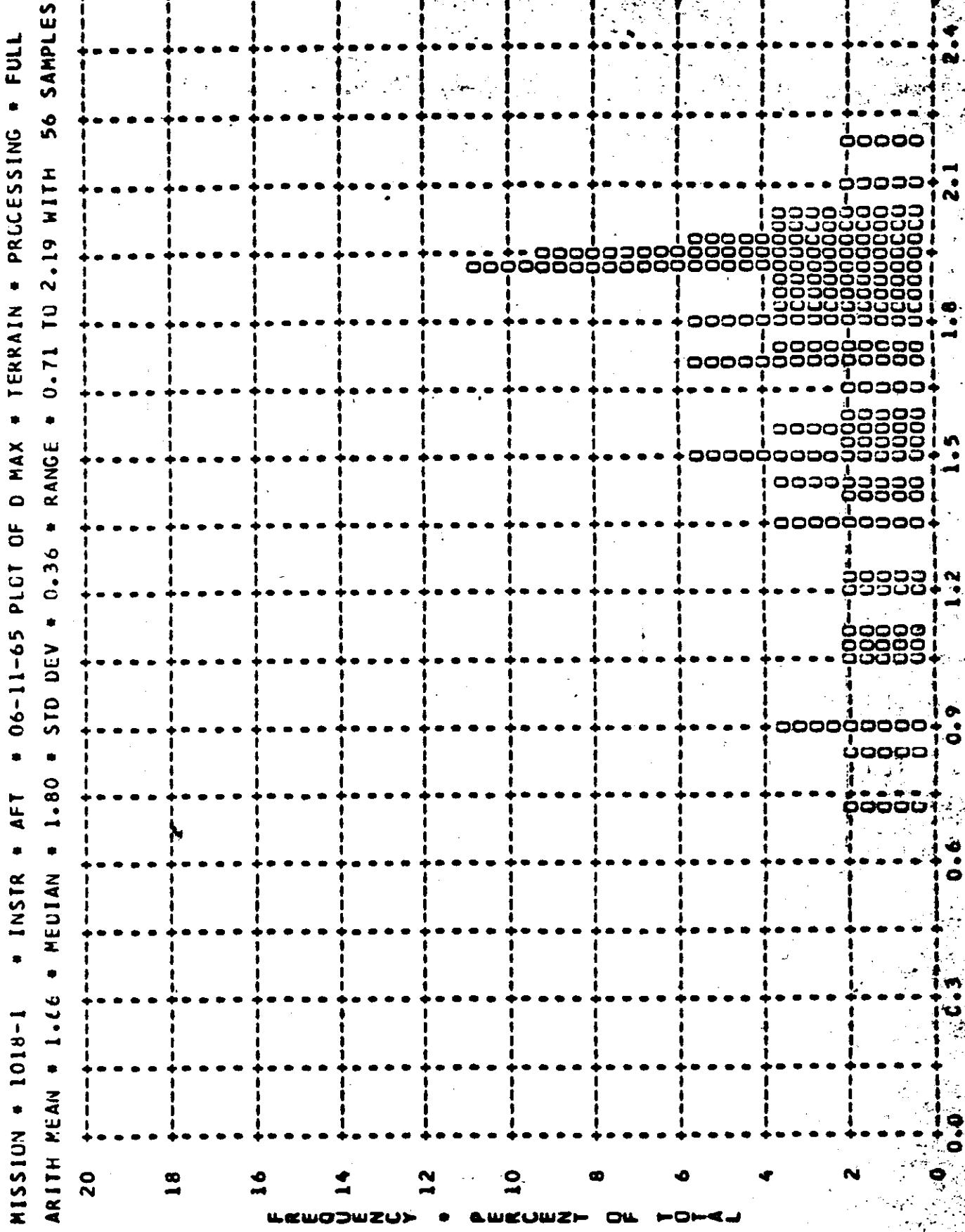
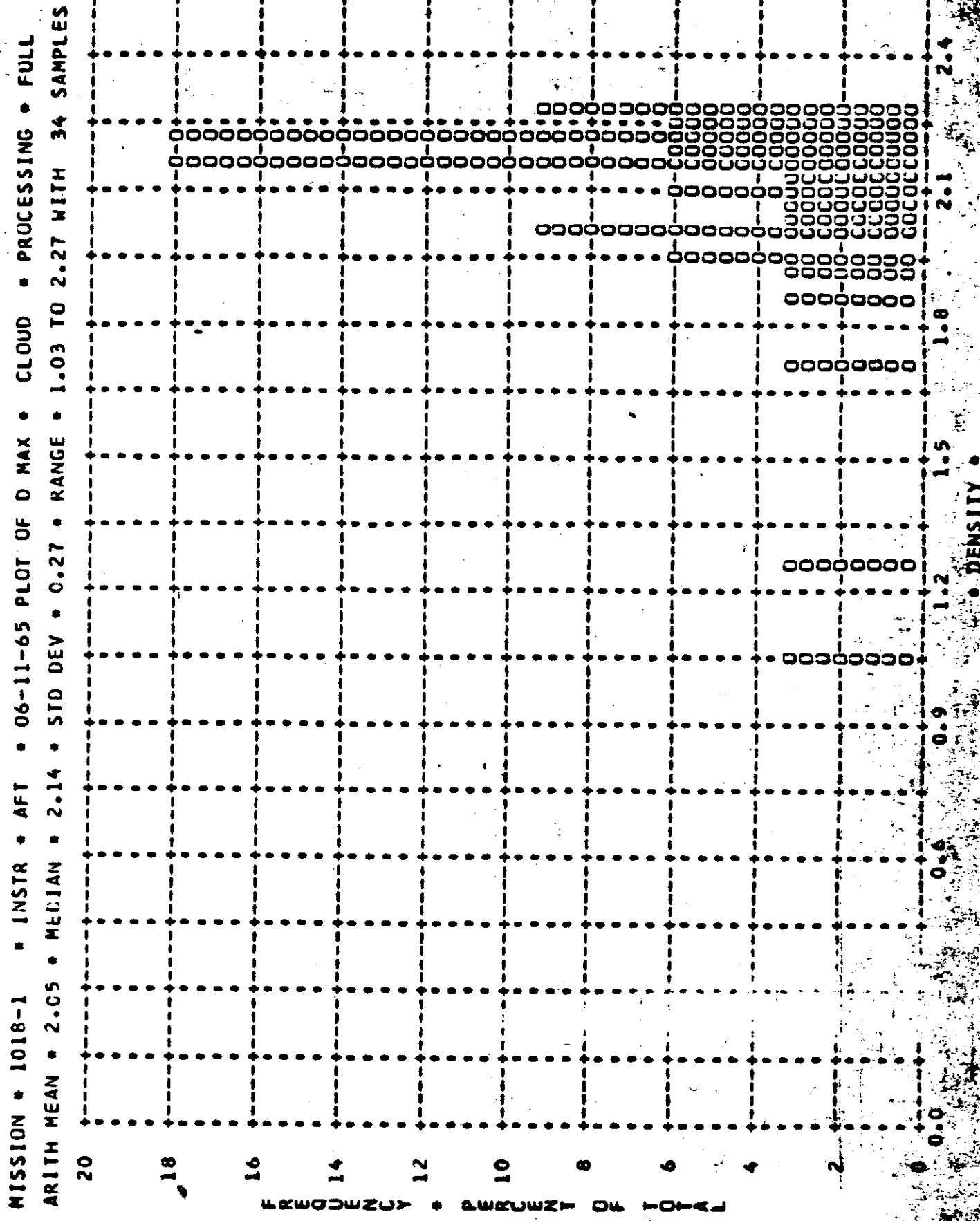


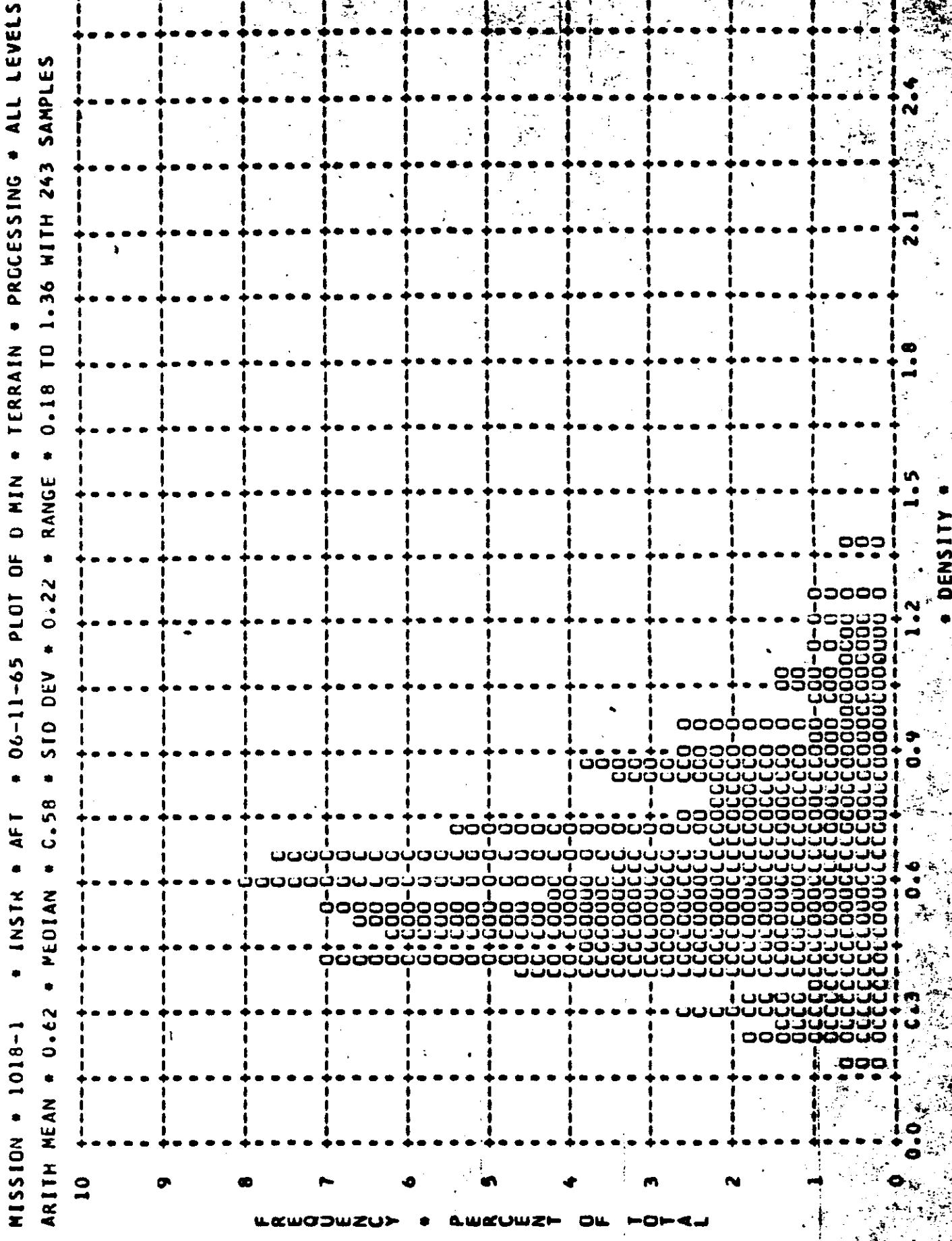
Figure A-19

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A-34

Figure A-22

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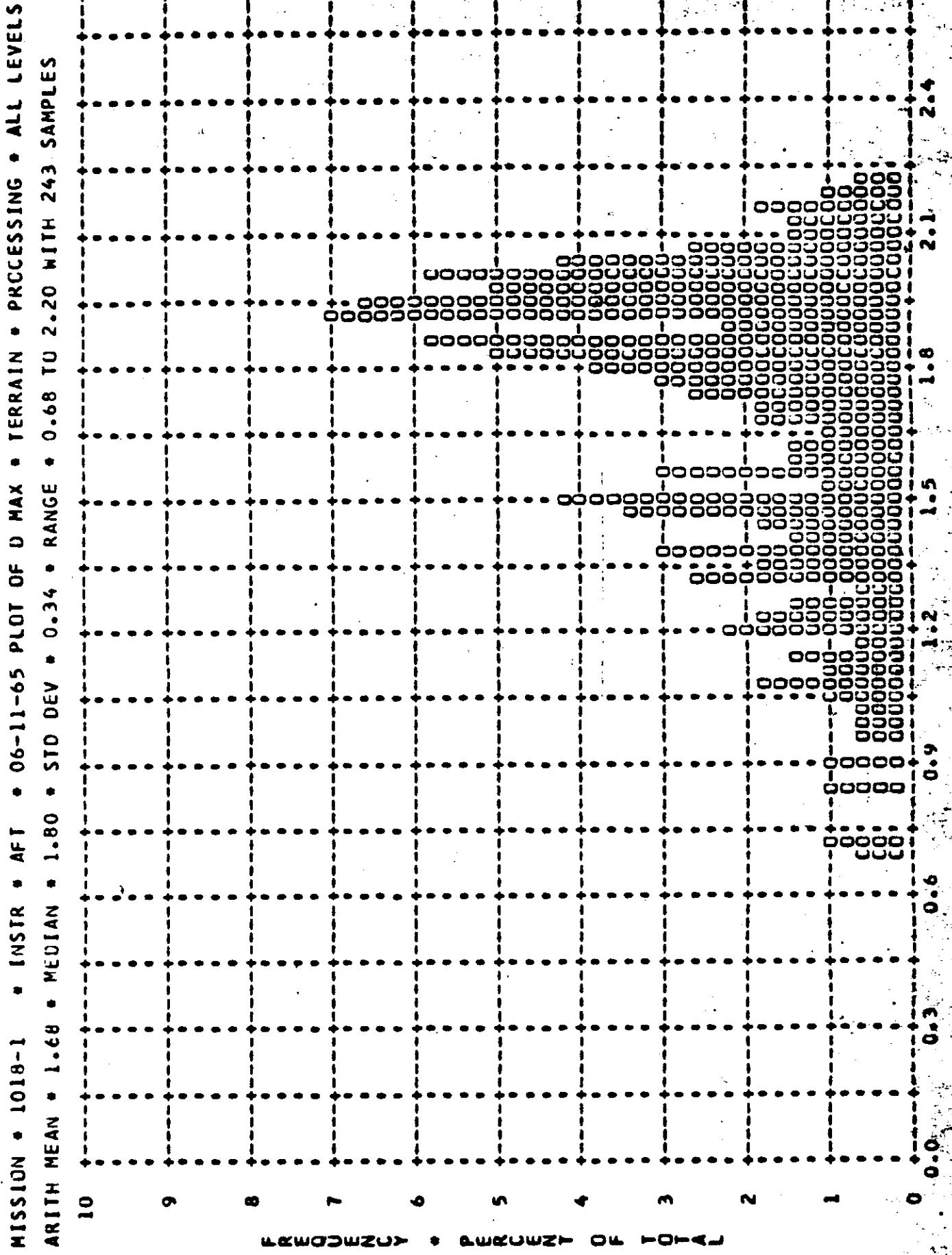


Figure A-23

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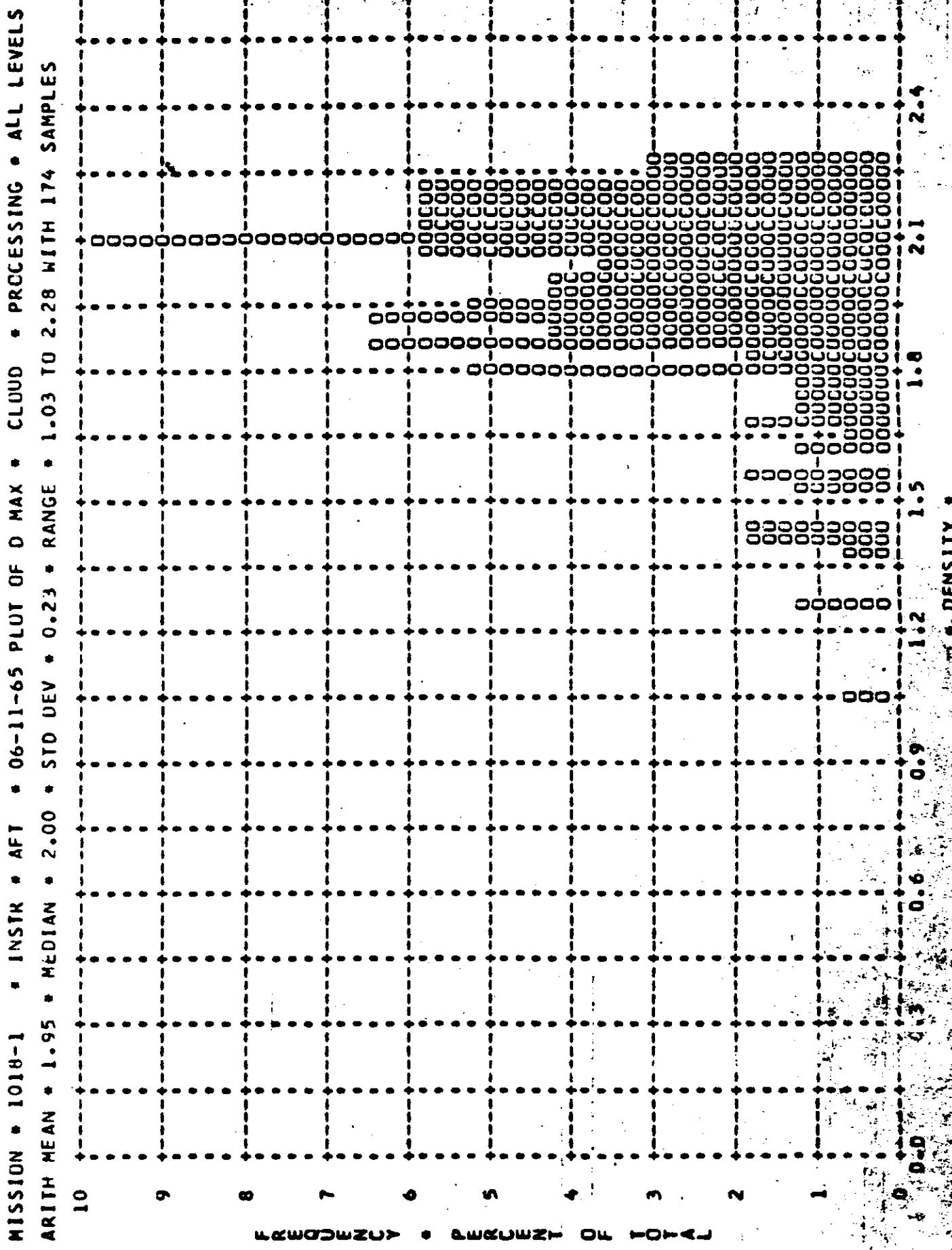


Figure A-24

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MISSION • 1018-2    INSTRUMENT • FWD    06-11-65    DENSITY FREQ DISTR.

DENSITY VALUE	PRIMARY			INTERMEDIATE			FULL			ALL LEVELS		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
0.01	0	0	0	0	0	0	0	0	0	0	0	0
0.02	0	0	0	0	0	0	0	0	0	0	0	0
0.03	0	0	0	0	0	0	0	0	0	0	0	0
0.04	0	0	0	0	0	0	0	0	0	0	0	0
0.05	0	0	0	0	0	0	0	0	0	0	0	0
0.06	0	0	0	0	0	0	0	0	0	0	0	0
0.07	0	0	0	0	0	0	0	0	0	0	0	0
0.08	0	0	0	0	0	0	0	0	0	0	0	0
0.09	0	0	0	0	0	0	0	0	0	0	0	0
0.10	0	0	0	0	0	0	0	0	0	0	0	0
0.11	0	0	0	0	0	0	0	0	0	0	0	0
0.12	0	0	0	0	0	0	0	0	0	0	0	0
0.13	0	0	0	0	0	0	0	0	0	0	0	0
0.14	0	0	0	0	0	0	0	0	0	0	0	0
0.15	0	0	0	0	0	0	0	0	0	0	0	0
0.16	0	0	0	0	0	0	0	0	0	0	0	0
0.17	0	0	0	0	0	0	0	0	0	0	0	0
0.18	0	0	0	0	0	0	0	0	0	0	0	0
0.19	0	0	0	0	0	0	0	0	0	0	0	0
0.20	0	0	0	0	0	0	0	0	0	0	0	0
0.21	0	0	0	0	0	0	0	0	0	0	0	0
0.22	0	0	0	0	0	0	0	0	0	0	0	0
0.23	0	0	0	0	0	0	0	0	0	0	0	0
0.24	0	0	0	0	0	0	0	0	0	0	0	0
0.25	0	0	0	0	0	0	0	0	0	0	0	0
0.26	0	0	0	0	0	0	0	0	0	0	0	0
0.27	0	0	0	0	0	0	0	0	0	0	0	0
0.28	0	0	0	0	0	0	0	0	0	0	0	0
0.29	0	0	0	0	0	0	0	0	0	0	0	0
0.30	0	0	0	0	0	0	0	0	0	0	0	0
0.31	0	0	0	0	0	0	0	0	0	0	0	0
0.32	0	0	0	0	0	0	0	0	0	0	0	0
0.33	0	0	0	0	0	0	0	0	0	0	0	0
0.34	0	0	0	0	0	0	0	0	0	0	0	0
0.35	0	0	0	0	0	0	0	0	0	0	0	0
0.36	0	0	0	0	0	0	0	0	0	0	0	0
0.37	0	0	0	0	0	0	0	0	0	0	0	0
0.38	0	0	0	0	0	0	0	0	0	0	0	0
0.39	0	0	0	0	0	0	0	0	0	0	0	0
0.40	0	0	0	0	0	0	0	0	0	0	0	0
0.41	0	0	0	0	0	0	0	0	0	0	0	0
0.42	0	0	0	0	0	0	0	0	0	0	0	0
0.43	0	0	0	0	0	0	0	0	0	0	0	0
0.44	0	0	0	0	0	0	0	0	0	0	0	0
0.45	0	0	0	0	0	0	0	0	0	0	0	0
0.46	0	0	0	0	0	0	0	0	0	0	0	0
0.47	0	0	0	0	0	0	0	0	0	0	0	0
0.48	0	0	0	0	0	0	0	0	0	0	0	0
0.49	0	0	0	0	0	0	0	0	0	0	0	0
0.50	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	23			104						133		

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MISSION • 1018-2 • INSTRUMENT • FHD 06-11-65 DENSITY FREQ DISTR

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MISSION • 1018-2 • INSTRUMENT • FWD 06-11-65 DENSITY FREQ UISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
1.01	C	C	0	0
1.02	CCCC	CCCC	0	0
1.03	CCCC	CCCC	0	0
1.04	CCCC	CCCC	0	0
1.05	CCCC	CCCC	0	0
1.06	CCCC	CCCC	0	0
1.07	CCCC	CCCC	0	0
1.08	CCCC	CCCC	0	0
1.09	CCCC	CCCC	0	0
1.10	CCCC	CCCC	0	0
1.11	CCCC	CCCC	0	0
1.12	CCCC	CCCC	0	0
1.13	CCCC	CCCC	0	0
1.14	CCCC	CCCC	0	0
1.15	CCCC	CCCC	0	0
1.16	CCCC	CCCC	0	0
1.17	CCCC	CCCC	0	0
1.18	CCCC	CCCC	0	0
1.19	CCCC	CCCC	0	0
1.20	CCCC	CCCC	0	0
1.21	CCCC	CCCC	0	0
1.22	CCCC	CCCC	0	0
1.23	CCCC	CCCC	0	0
1.24	CCCC	CCCC	0	0
1.25	CCCC	CCCC	0	0
1.26	CCCC	CCCC	0	0
1.27	CCCC	CCCC	0	0
1.28	CCCC	CCCC	0	0
1.29	CCCC	CCCC	0	0
1.30	CCCC	CCCC	0	0
1.31	CCCC	CCCC	0	0
1.32	CCCC	CCCC	0	0
1.33	CCCC	CCCC	0	0
1.34	CCCC	CCCC	0	0
1.35	CCCC	CCCC	0	0
1.36	CCCC	CCCC	0	0
1.37	CCCC	CCCC	0	0
1.38	CCCC	CCCC	0	0
1.39	CCCC	CCCC	0	0
1.40	CCCC	CCCC	0	0
1.41	CCCC	CCCC	0	0
1.42	CCCC	CCCC	0	0
1.43	CCCC	CCCC	0	0
1.44	CCCC	CCCC	0	0
1.45	CCCC	CCCC	0	0
1.46	CCCC	CCCC	0	0
1.47	CCCC	CCCC	0	0
1.48	CCCC	CCCC	0	0
1.49	CCCC	CCCC	0	0
1.50	CCCC	CCCC	0	0
SUBTOTAL-	2	61	26	22

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MISSION • 1018-2 • INSTRUMENT • FWD 06-11-65 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
1.51	0	0	0	0
1.52	0	0	0	0
1.53	0	0	0	0
1.54	0	0	0	0
1.55	0	0	0	0
1.56	0	0	0	0
1.57	0	0	0	0
1.58	0	0	0	0
1.59	0	0	0	0
1.60	0	0	0	0
1.61	0	0	0	0
1.62	0	0	0	0
1.63	0	0	0	0
1.64	0	0	0	0
1.65	0	0	0	0
1.66	0	0	0	0
1.67	0	0	0	0
1.68	0	0	0	0
1.69	0	0	0	0
1.70	0	0	0	0
1.71	0	0	0	0
1.72	0	0	0	0
1.73	0	0	0	0
1.74	0	0	0	0
1.75	0	0	0	0
1.76	0	0	0	0
1.77	0	0	0	0
1.78	0	0	0	0
1.79	0	0	0	0
1.80	0	0	0	0
1.81	0	0	0	0
1.82	0	0	0	0
1.83	0	0	0	0
1.84	0	0	0	0
1.85	0	0	0	0
1.86	0	0	0	0
1.87	0	0	0	0
1.88	0	0	0	0
1.89	0	0	0	0
1.90	0	0	0	0
1.91	0	0	0	0
1.92	0	0	0	0
1.93	0	0	0	0
1.94	0	0	0	0
1.95	0	0	0	0
1.96	0	0	0	0
1.97	0	0	0	0
1.98	0	0	0	0
1.99	0	0	0	0
2.00	0	0	0	0
SUBTOTAL	15	22	10	85

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MISSION • 1018-2 • INSTRUMENT • FWD      06-11-65      DENSITY FREQ UISTR

DENSITY VALUE	PRIMARY		INTERMEDIATE		FULL	ALL LEVELS	
	MIN	MAX LIM	MIN	MAX LIM	MIN MAX LIM	MIN MAX LIM	MIN MAX LIM
2.01	0	0	0	0	0	0	0
2.02	0	0	0	0	0	0	0
2.03	0	0	0	0	0	0	0
2.04	0	0	0	0	0	0	0
2.05	0	0	0	0	0	0	0
2.06	0	0	0	0	0	0	0
2.07	0	0	0	0	0	0	0
2.08	0	0	0	0	0	0	0
2.09	0	0	0	0	0	0	0
2.10	0	0	0	0	0	0	0
2.11	0	0	0	0	0	0	0
2.12	0	0	0	0	0	0	0
2.13	0	0	0	0	0	0	0
2.14	0	0	0	0	0	0	0
2.15	0	0	0	0	0	0	0
2.16	0	0	0	0	0	0	0
2.17	0	0	0	0	0	0	0
2.18	0	0	0	0	0	0	0
2.19	0	0	0	0	0	0	0
2.20	0	0	0	0	0	0	0
2.21	0	0	0	0	0	0	0
2.22	0	0	0	0	0	0	0
2.23	0	0	0	0	0	0	0
2.24	0	0	0	0	0	0	0
2.25	0	0	0	0	0	0	0
2.26	0	0	0	0	0	0	0
2.27	0	0	0	0	0	0	0
2.28	0	0	0	0	0	0	0
2.29	0	0	0	0	0	0	0
2.30	0	0	0	0	0	0	0
2.31	0	0	0	0	0	0	0
2.32	0	0	0	0	0	0	0
2.33	0	0	0	0	0	0	0
2.34	0	0	0	0	0	0	0
2.35	0	0	0	0	0	0	0
2.36	0	0	0	0	0	0	0
2.37	0	0	0	0	0	0	0
2.38	0	0	0	0	0	0	0
2.39	0	0	0	0	0	0	0
2.40	0	0	0	0	0	0	0
2.41	0	0	0	0	0	0	0
2.42	0	0	0	0	0	0	0
2.43	0	0	0	0	0	0	0
2.44	0	0	0	0	0	0	0
2.45	0	0	0	0	0	0	0
2.46	0	0	0	0	0	0	0
2.47	0	0	0	0	0	0	0
2.48	0	0	0	0	0	0	0
2.49	0	0	0	0	0	0	0
2.50	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	17	73	90
							19 0 89

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MISSION • 1018-2 • INSTRUMENT • FWD      06-11-65      DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
2.51	0 0	0 0	0 0	0 0
2.52	0 0	0 0	0 0	0 0
2.53	0 0	0 0	0 0	0 0
2.54	0 0	0 0	0 0	0 0
2.55	0 0	0 0	0 0	0 0
2.56	0 0	0 0	0 0	0 0
2.57	0 0	0 0	0 0	0 0
2.58	0 0	0 0	0 0	0 0
2.59	0 0	0 0	0 0	0 0
2.60	0 0	0 0	0 0	0 0
2.61	0 0	0 0	0 0	0 0
2.62	0 0	0 0	0 0	0 0
2.63	0 0	0 0	0 0	0 0
2.64	0 0	0 0	0 0	0 0
2.65	0 0	0 0	0 0	0 0
2.66	0 0	0 0	0 0	0 0
2.67	0 0	0 0	0 0	0 0
2.68	0 0	0 0	0 0	0 0
2.69	0 0	0 0	0 0	0 0
2.70	0 0	0 0	0 0	0 0
SUBTOTAL	44 44	38 0	204 204	184 0
TOTAL	44 44	38 0	204 204	184 0
			9 9	10 0
			257 257	232 0

MISSION 1C18-2      INSTR - FWD      06-11-65      PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXP+SED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	44	0 PC	32 PC	61 PC	0 PC	7 PC
INTERMEDIATE	204	0 PC	32 PC	54 PC	11 PC	3 PC
FULL	9	22 PC	0 PC	7A PC	0 PC	0 PC
ALL LEVELS	257	1 PC	31 PC	56 PC	9 PC	4 PC
PROCESS LEVEL	BASE + FCG	UNDER EXP+SED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-C.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND UP
INTERMED	0.10-C.17	0.01-C.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UP
FULL	0.18 AND UP	0.01-C.39	-----	0.40-0.90	0.91-1.69	1.70 AND UP

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MISSION • 1018-2 • INSTR • FWD • 06-11-65 PLOT OF L. MIN • TERRAIN • PROCESSING • PRIMARY  
ARITH MEAN • 0.52 • MEDIAN • 0.50 • STD DEV • 0.23 • RANGE • 0.16 TO 1.09 WITH 44 SAMPLES

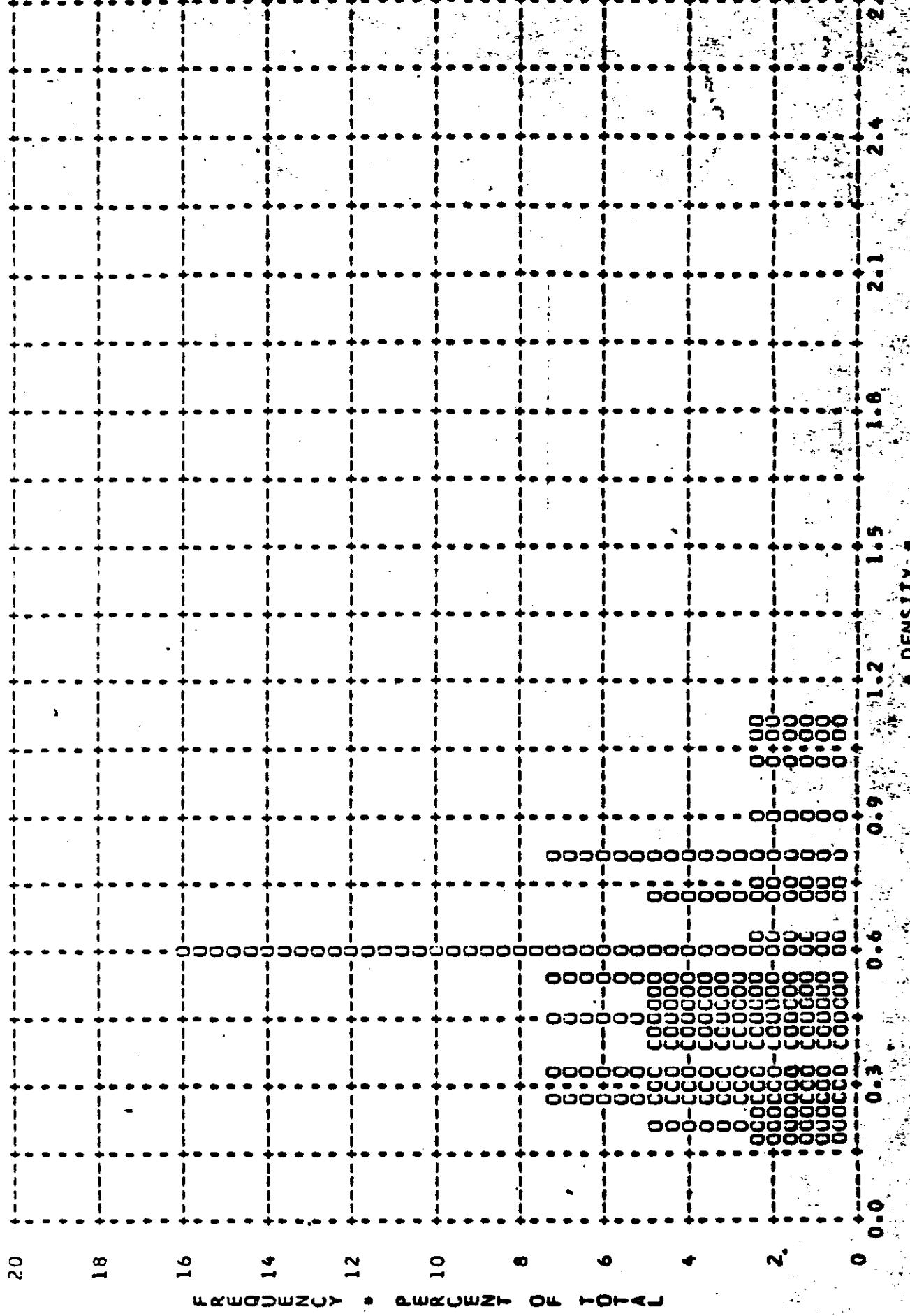
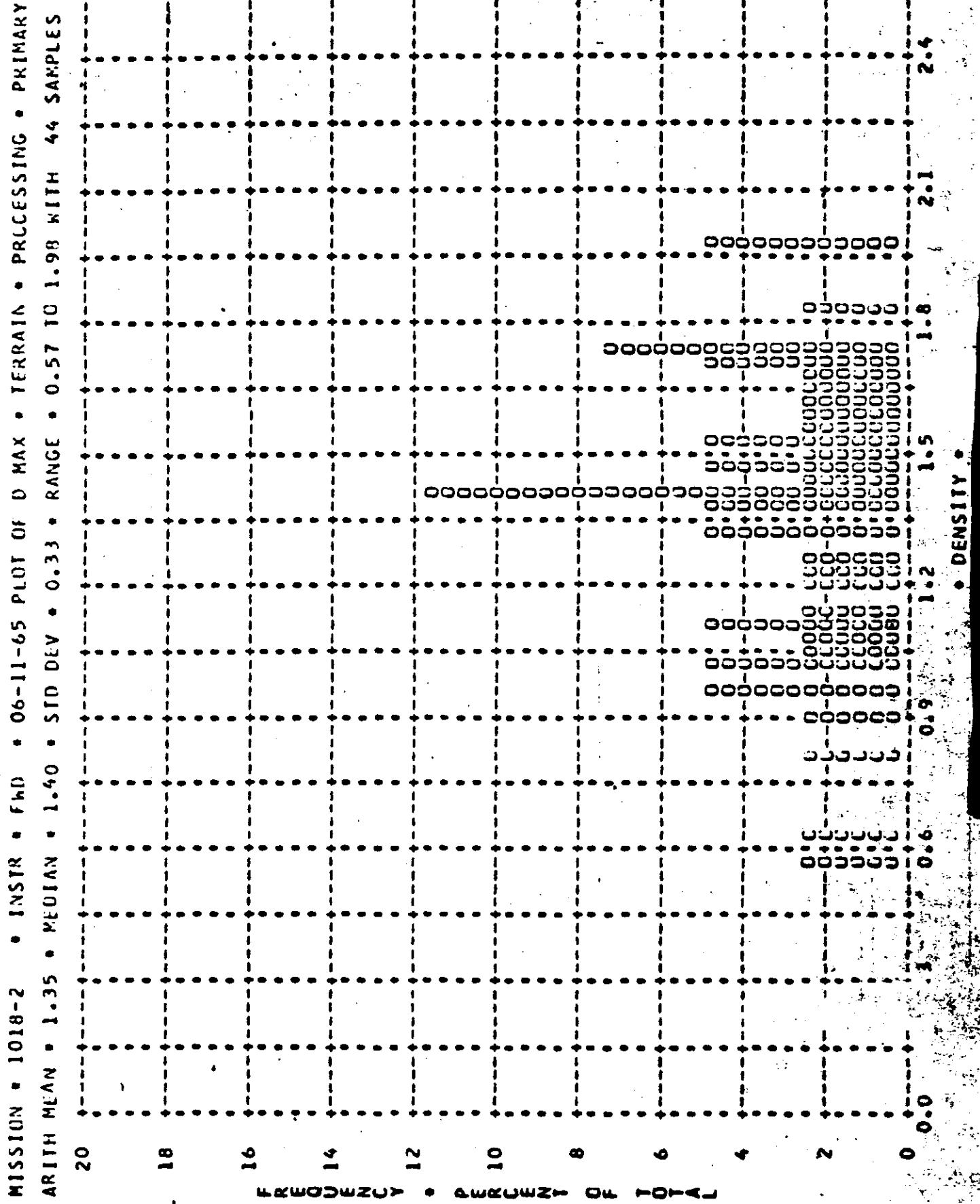
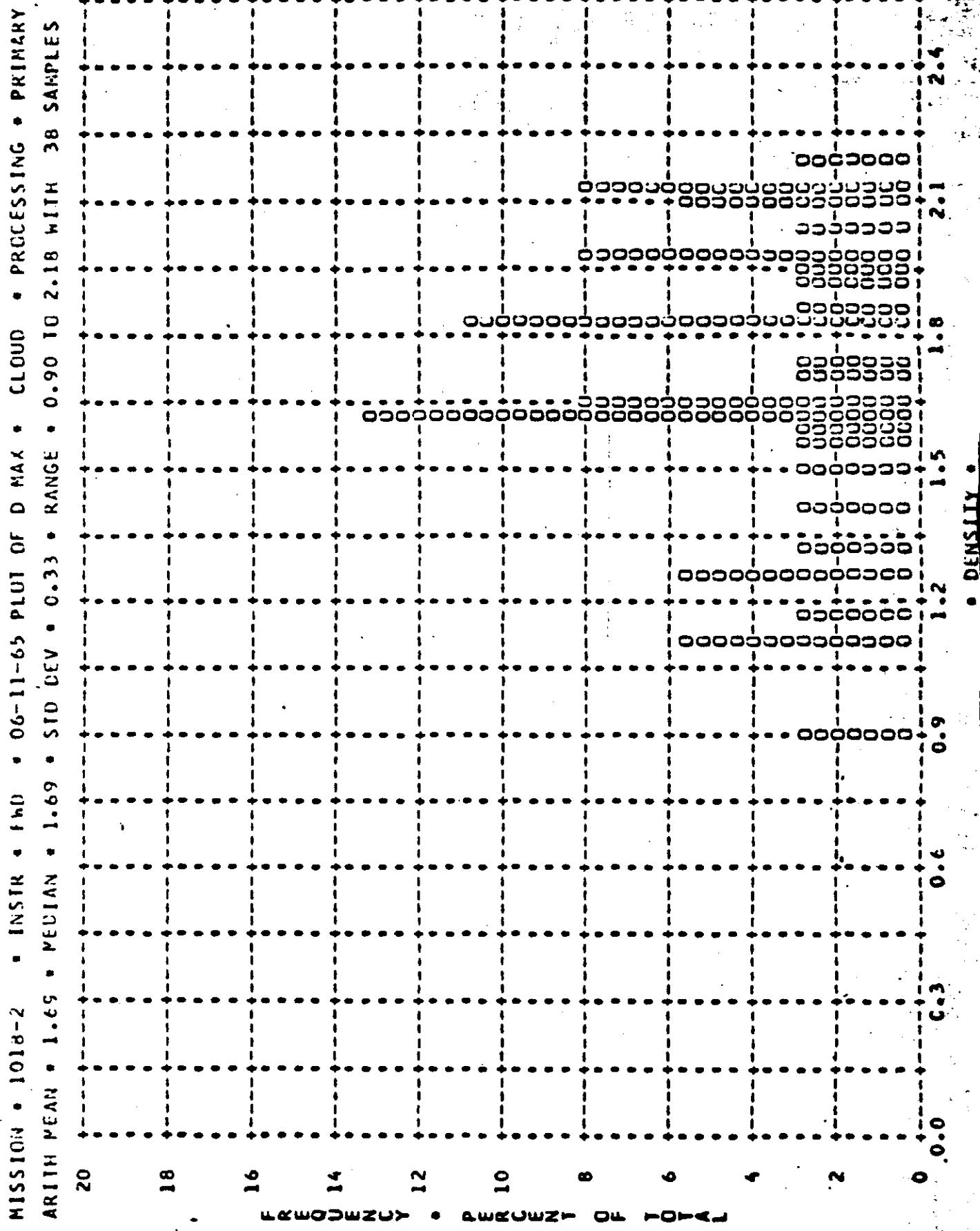


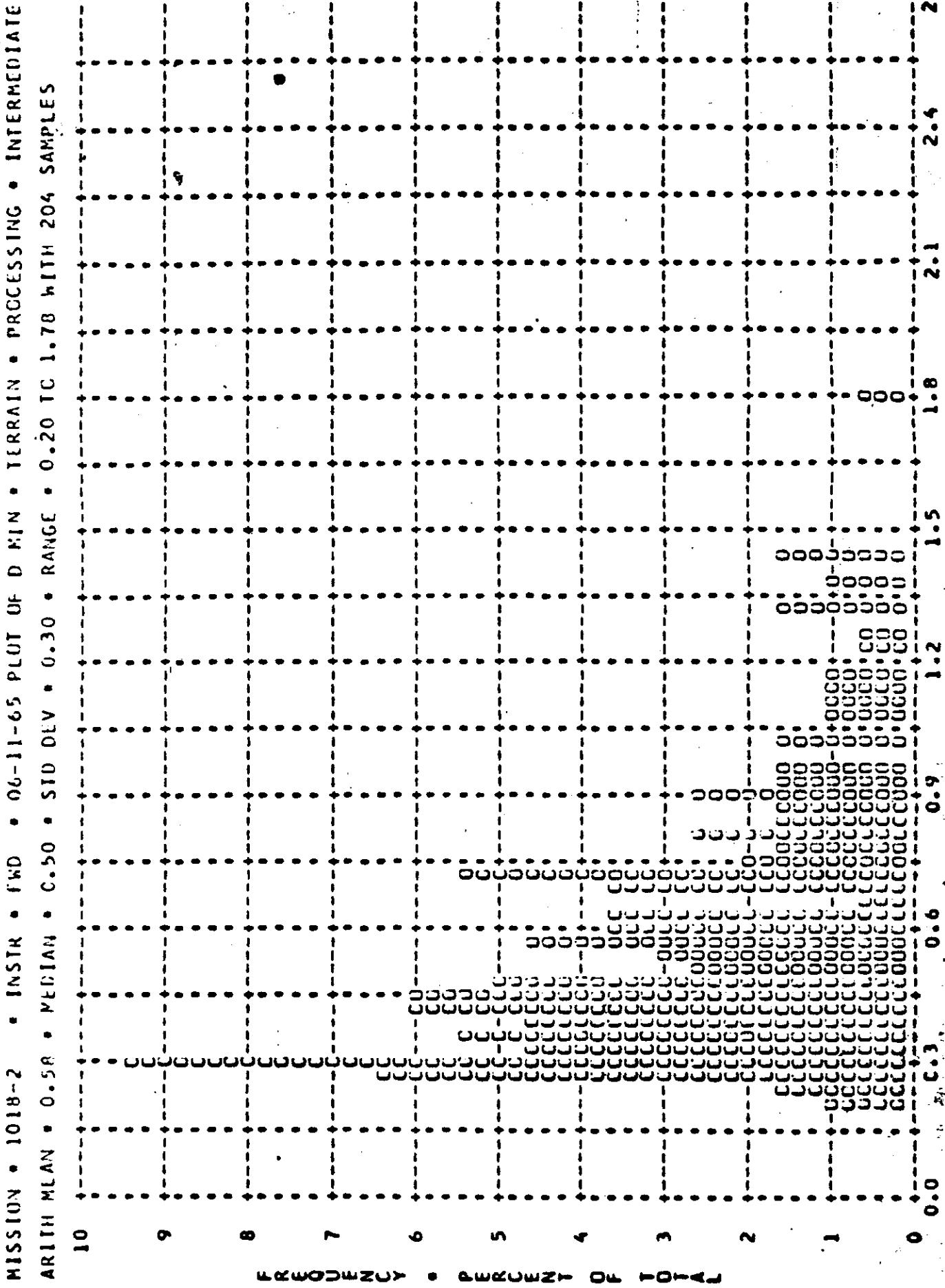
Figure A-25



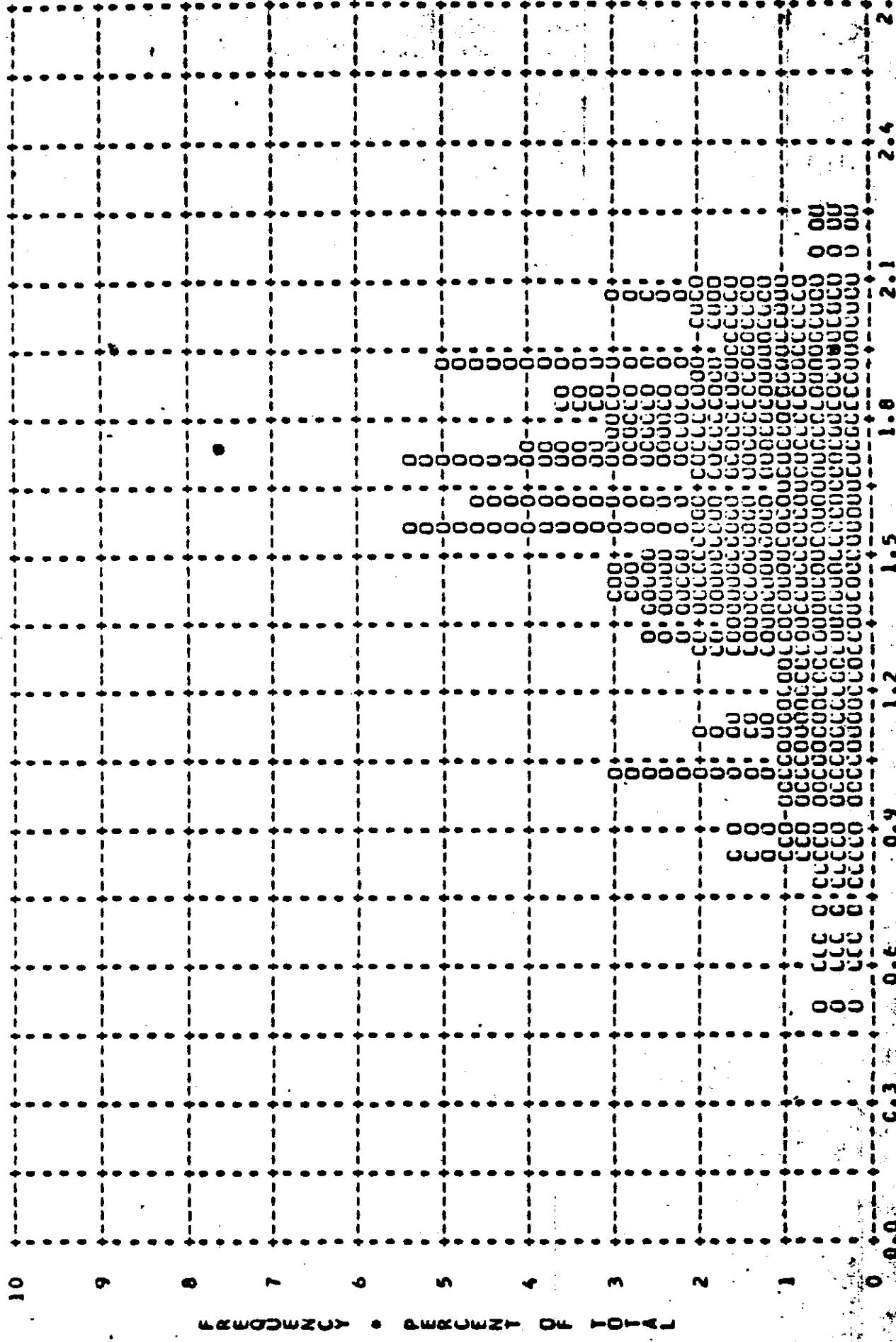
A-44

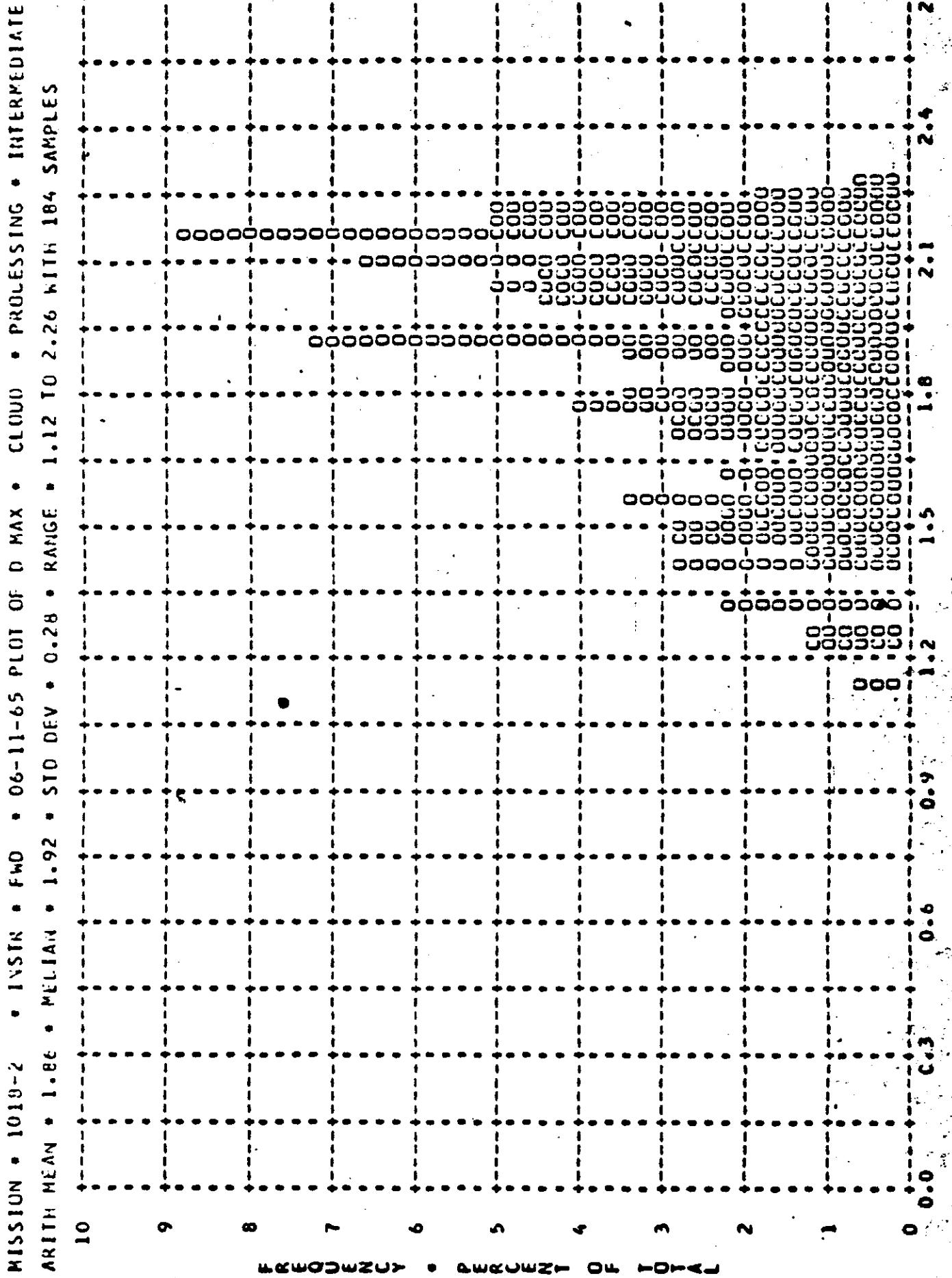
Figure A-26



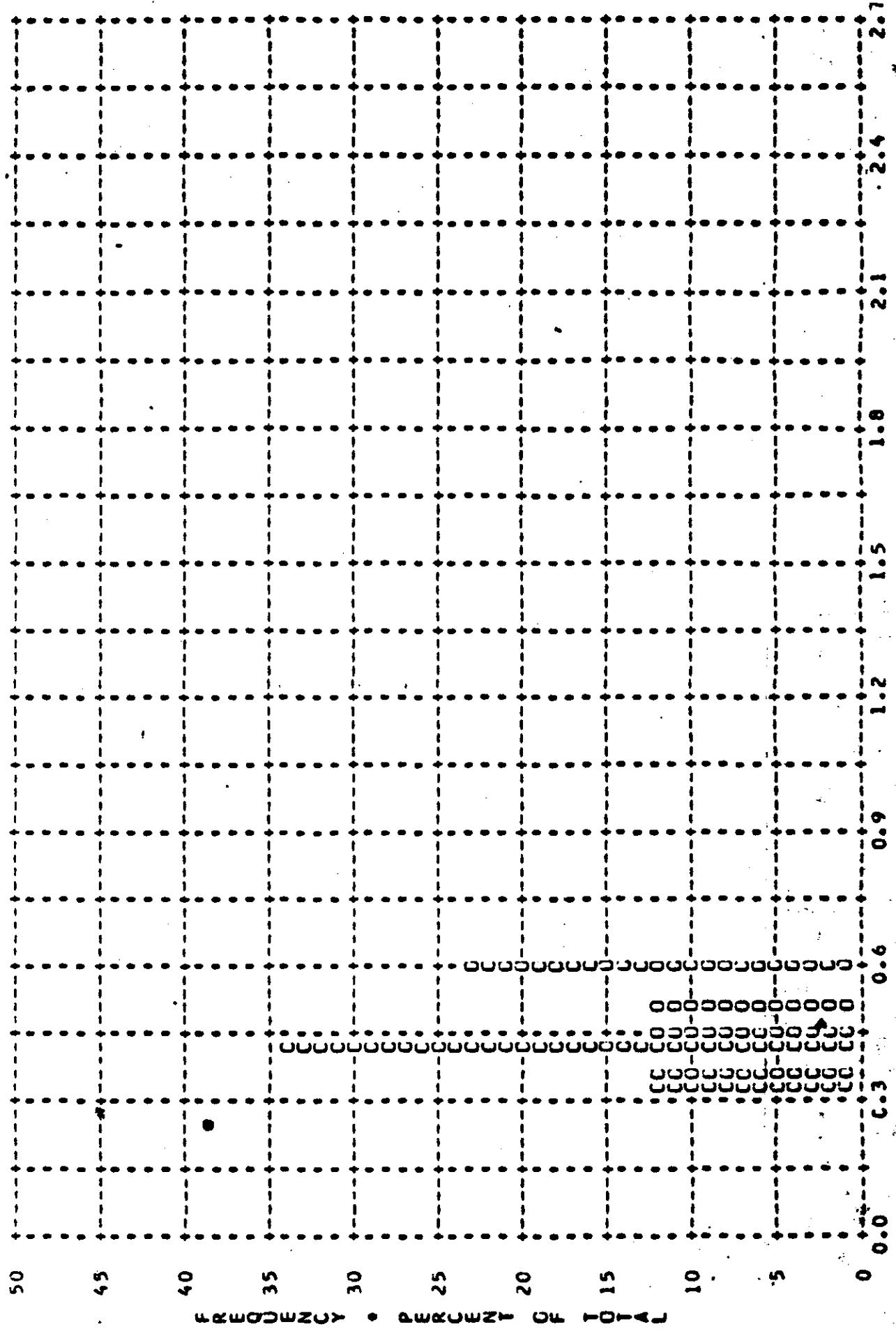


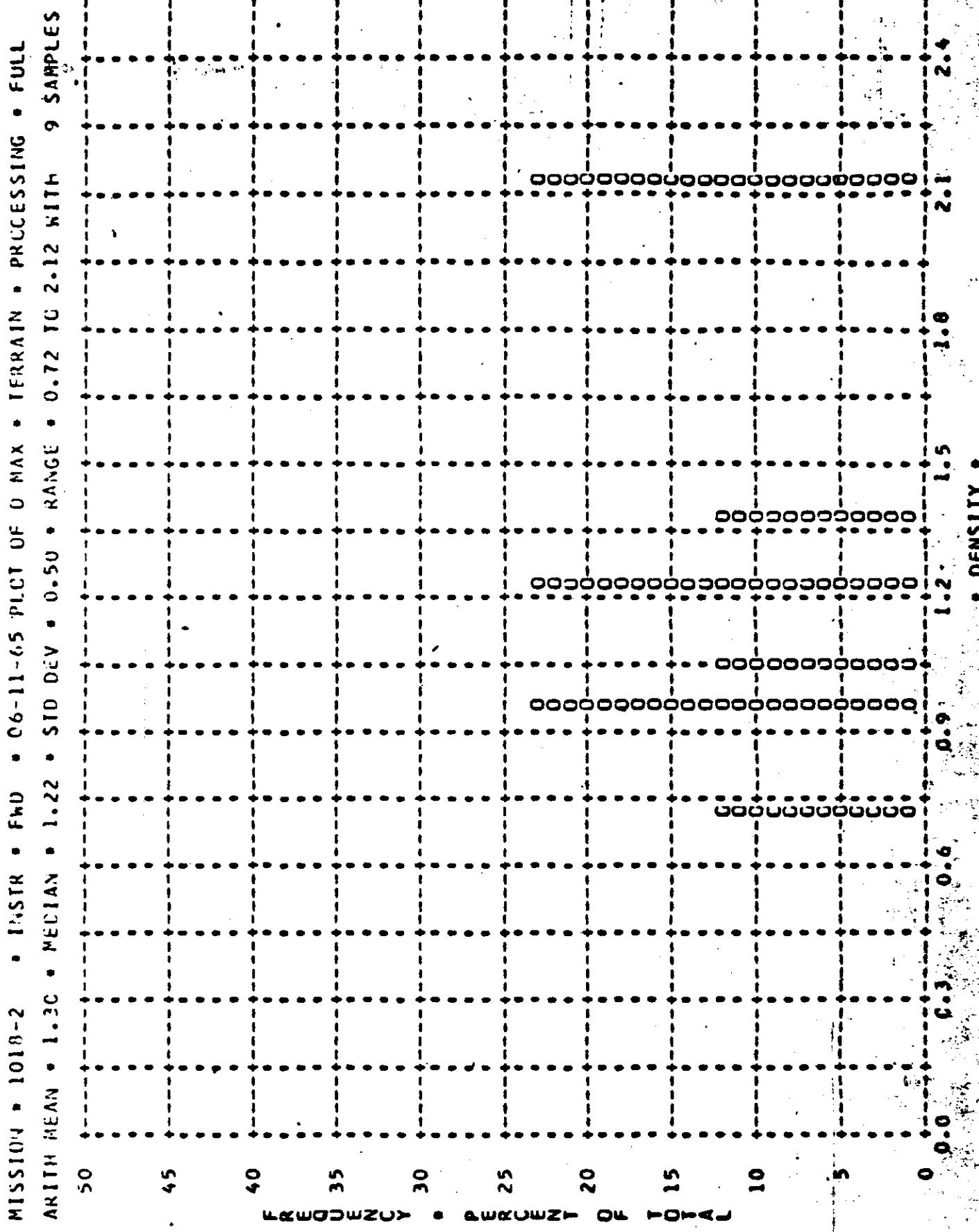
MISSION • 101B-2 • INSTR • FAD • 06-11-65 PLCT OF D MAX • TERRAIN • PROCESSING • INTERMEDIATE  
ARITH MEAN • 1.54 • MEDIAN • 1.59 • STD DEV • 0.37 • RANGE • 0.50 TC 2.24 WITH 204 SAMPLES





MISSION • LC18-2 • LASTR • FWD • 06-11-65 PLCT OF U MIN • TERRAIN • PROCESSING • FULL  
ARITH MEAN • 0.45 • MEDIAN • C.42 • STD DEV • C.10 • RANGE • 0.33 TO 0.60 WITH 9 SAMPLES

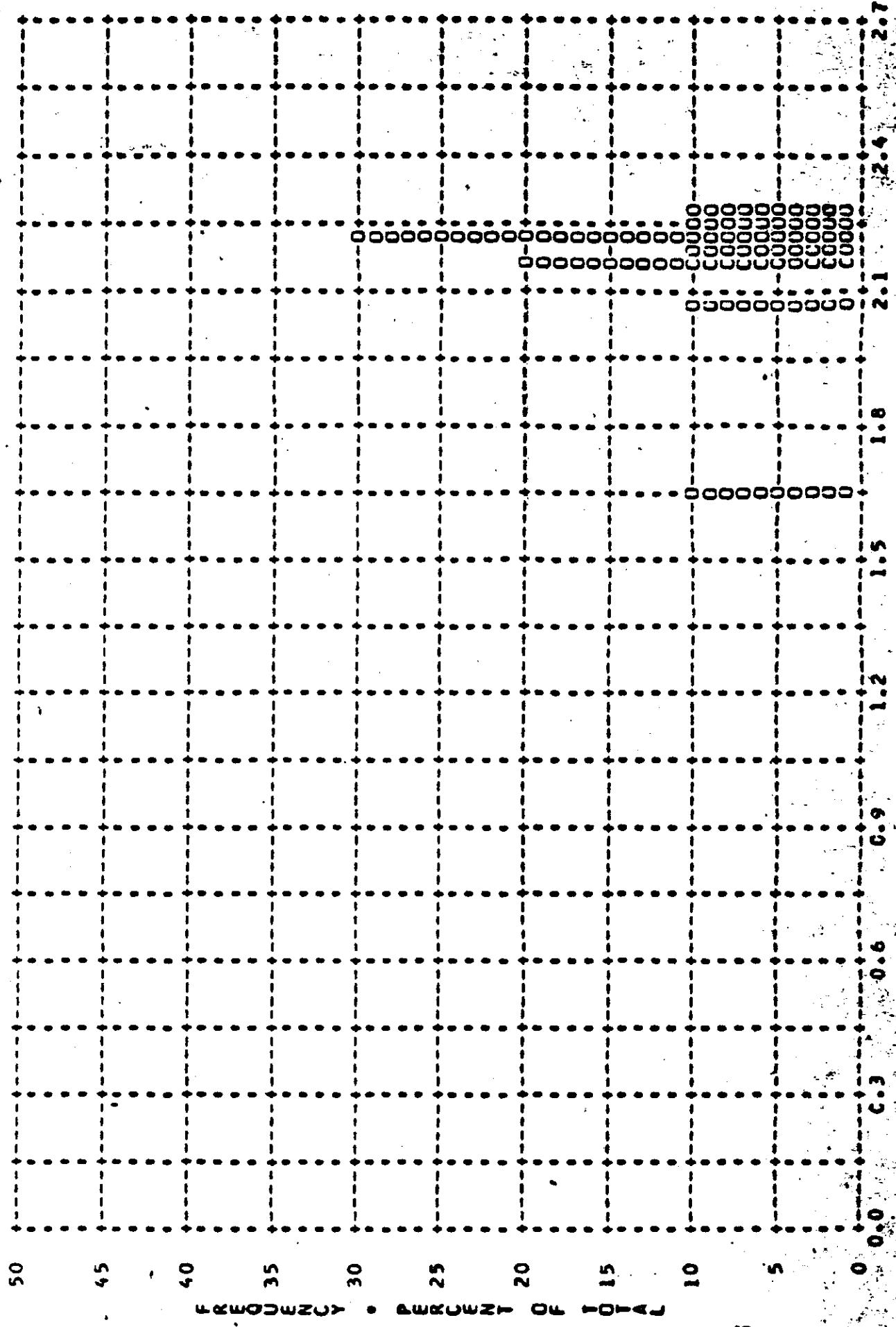


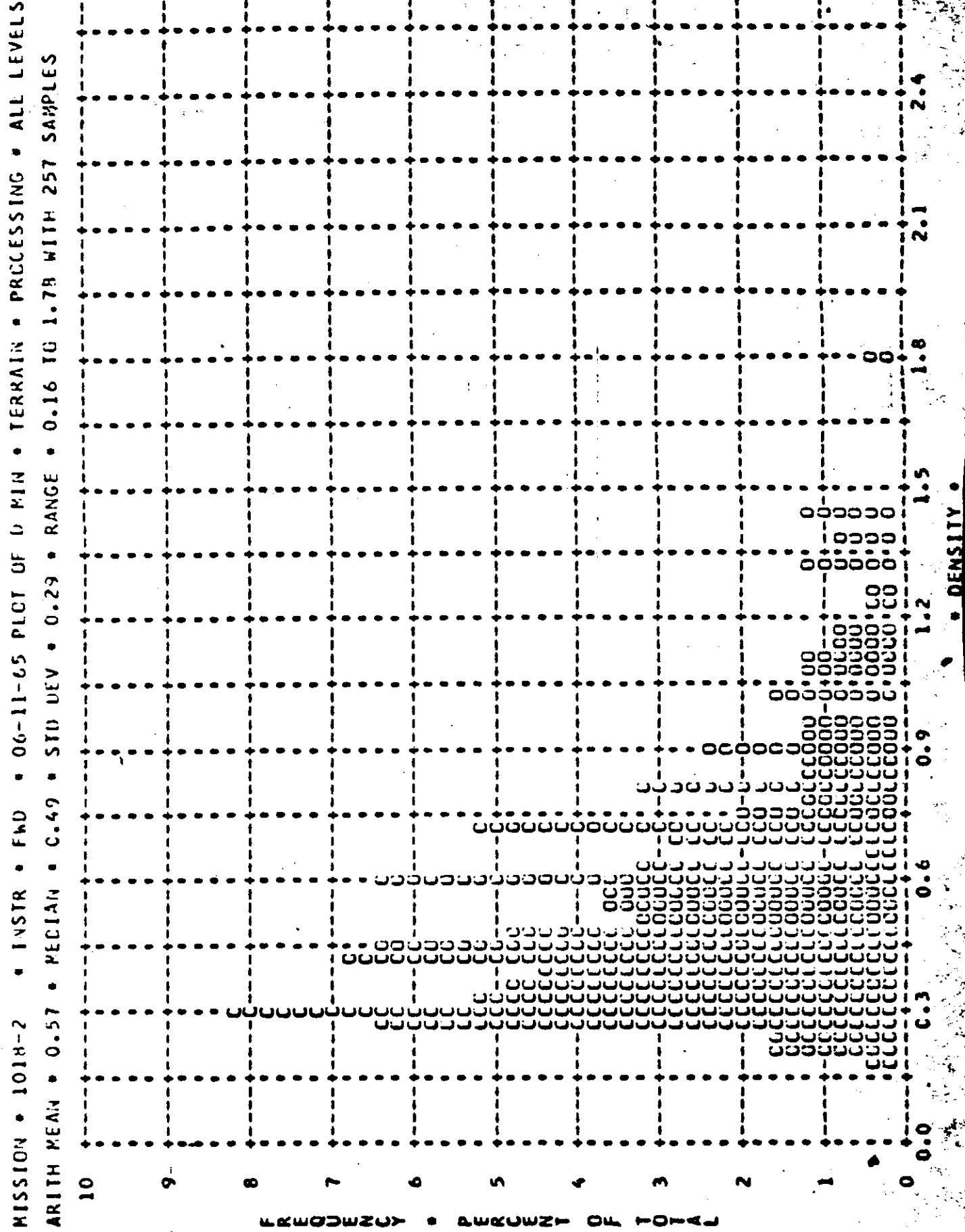


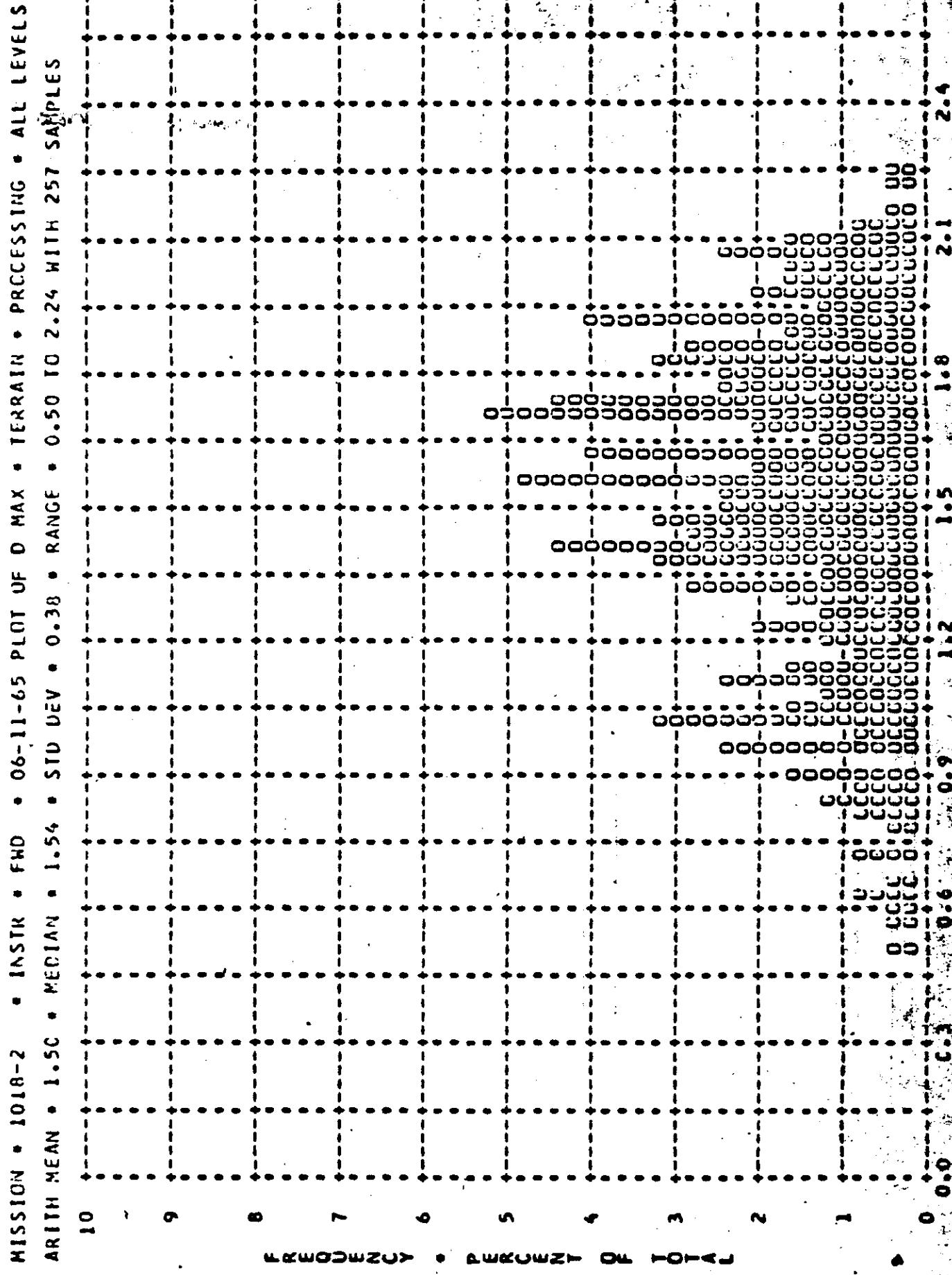
A-50

Figure A-32

MISSION \* 1018-2 \* INSTR \* RAD \* 06-11-65 PLOT OF D MAX \* CLOUD \* PROCESSING \* FULL  
ARITH MEAN \* 2.13 \* MEDIAN \* 2.20 \* STD DEV \* 0.18 \* RANGE \* 1.64 TO 2.26 WITH 10 SAMPLES







A-53.

Figure A-35.

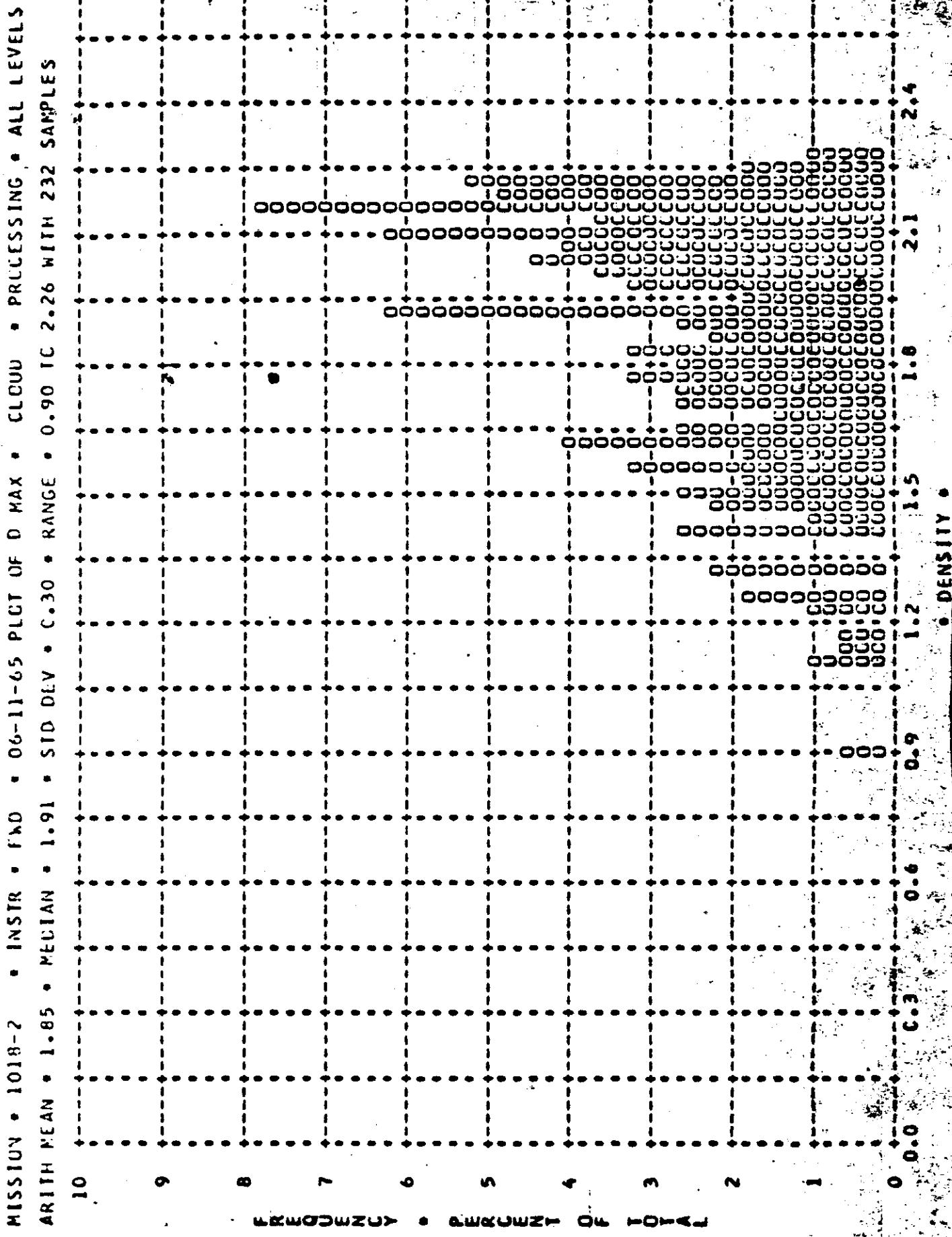


Figure A-36

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MISSION \* 1018-2 \* INSTRUMENT \* AFT 06-11-65 DENSITY FREQ. DISTR.

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
0.01	C	0	0	0
0.02	C	0	0	0
0.03	C	0	0	0
0.04	C	0	0	0
0.05	C	0	0	0
0.06	C	0	0	0
0.07	C	0	0	0
0.08	C	0	0	0
0.09	C	0	0	0
0.10	C	0	0	0
0.11	C	0	0	0
0.12	C	0	0	0
0.13	C	0	0	0
0.14	C	0	0	0
0.15	C	0	0	0
0.16	C	0	0	0
0.17	C	0	0	0
0.18	C	0	0	0
0.19	C	0	0	0
0.20	C	0	0	0
0.21	C	0	0	0
0.22	C	0	0	0
0.23	C	0	0	0
0.24	C	0	0	0
0.25	C	0	0	0
0.26	C	0	0	0
0.27	C	0	0	0
0.28	C	0	0	0
0.29	C	0	0	0
0.30	C	0	0	0
0.31	C	0	0	0
0.32	C	0	0	0
0.33	C	0	0	0
0.34	C	0	0	0
0.35	C	0	0	0
0.36	C	0	0	0
0.37	C	0	0	0
0.38	C	0	0	0
0.39	C	0	0	0
0.40	C	0	0	0
0.41	C	0	0	0
0.42	C	0	0	0
0.43	C	0	0	0
0.44	C	0	0	0
0.45	C	0	0	0
0.46	C	0	0	0
0.47	C	0	0	0
0.48	C	0	0	0
0.49	C	0	0	0
0.50	C	0	0	0
SUBTOTAL	C	0	36	0
			6	42

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MISSION • 1018-2 • INSTRUMENT • AFT • 06-11-65 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY - MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
0.51	C	0	0	0
0.52	C	0	0	0
0.53	C	0	0	0
0.54	C	0	0	0
0.55	C	0	0	0
0.56	C	0	0	0
0.57	C	0	0	0
0.58	C	0	0	0
0.59	C	0	0	0
0.60	C	0	0	0
0.612	C	0	0	0
0.63	C	0	0	0
0.64	C	0	0	0
0.65	C	0	0	0
0.66	C	0	0	0
0.67	C	0	0	0
0.68	C	0	0	0
0.69	C	0	0	0
0.70	C	0	0	0
0.71	C	0	0	0
0.72	C	0	0	0
0.73	C	0	0	0
0.74	C	0	0	0
0.75	C	0	0	0
0.76	C	0	0	0
0.77	C	0	0	0
0.78	C	0	0	0
0.79	C	0	0	0
0.80	C	0	0	0
0.81	C	0	0	0
0.82	C	0	0	0
0.83	C	0	0	0
0.84	C	0	0	0
0.85	C	0	0	0
0.86	C	0	0	0
0.87	C	0	0	0
0.88	C	0	0	0
0.89	C	0	0	0
0.901	C	0	0	0
0.92	C	0	0	0
0.93	C	0	0	0
0.94	C	0	0	0
0.95	C	0	0	0
0.96	C	0	0	0
0.97	C	0	0	0
0.98	C	0	0	0
0.99	C	0	0	0
1.00	C	0	0	0
SUBTOTAL		112	39	151

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MISSION \* 1018-2 \* INSTRUMENT \* AFT 06-11-65 DENSITY FREQ DISTR

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MISSION • 1018-2 • INSTRUMENT • AFT 06-11-65 DENSITY FREQ DISTR

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MISSION • 1018-2 • INSTRUMENT • AFT 06-11-65 DENSITY FREQ DISTR

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MISSION \* 1018-2 \* INSTRUMENT \* AFT 06-11-65 DENSITY FREQ DISTR

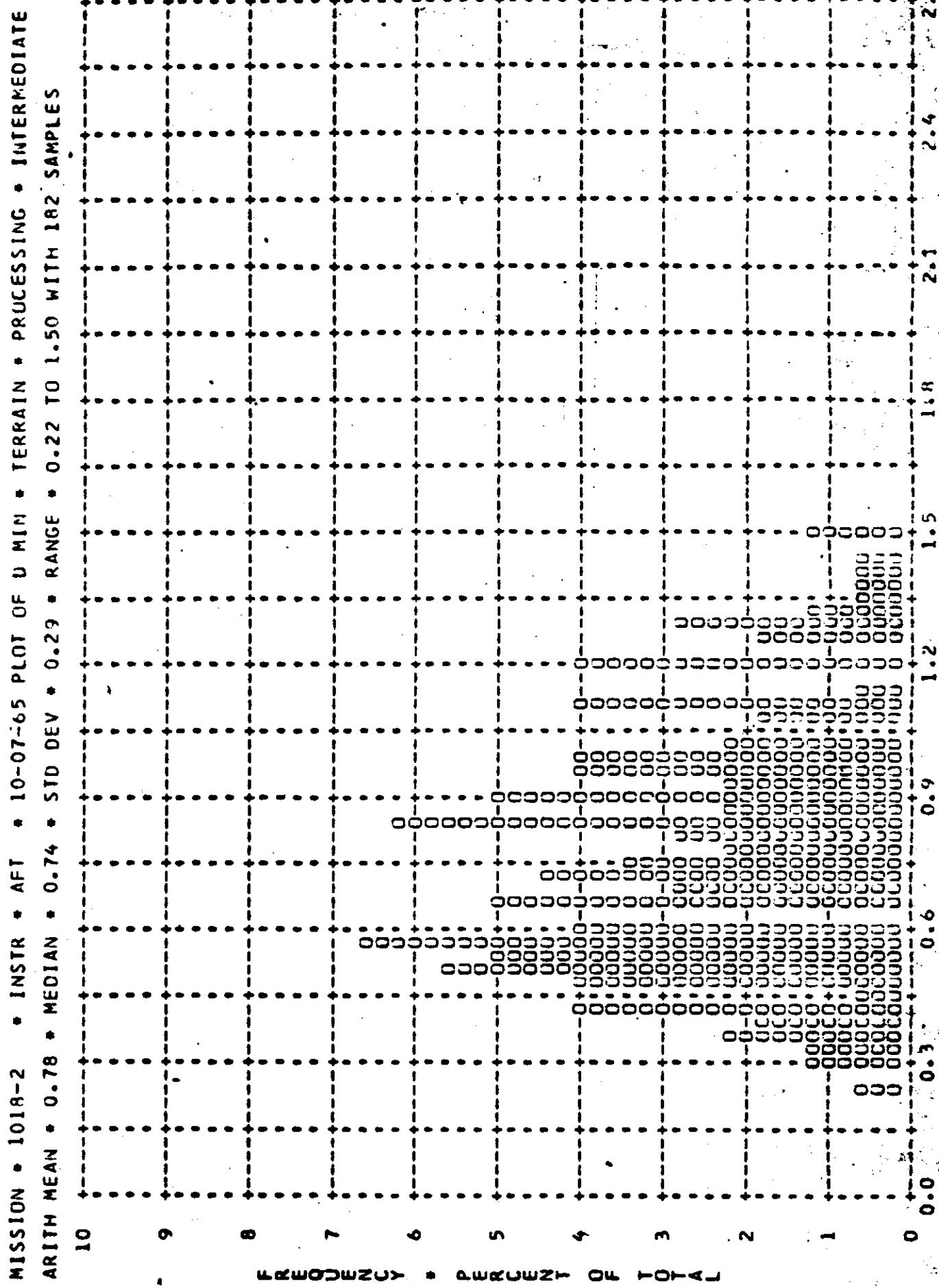
DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
2.51	C	0	0	0
2.52	CCC	00	00	00
2.53	CCCC	00	00	00
2.54	CCCCC	00	00	00
2.55	CCCCC	00	00	00
2.56	CCCCC	00	00	00
2.57	CCCCC	00	00	00
2.58	CCCCC	00	00	00
2.59	CCCCC	00	00	00
2.60	CCCCC	00	00	00
2.61	CCCCC	00	00	00
2.62	CCCCC	00	00	00
2.63	CCCCC	00	00	00
2.64	CCCCC	00	00	00
2.65	CCCCC	00	00	00
2.66	CCCCC	00	00	00
2.67	CCCCC	00	00	00
2.68	CCCCC	00	00	00
2.69	CCCCC	00	00	00
2.70	CCC	00	00	00
SUBTOTAL	C	0	0	0
TOTAL	C	0	182 182 175	65 65 61 247 247 236

MISSION 1018-2 INSTR - AFT 06-11-65 PROCESSING AND EXPOSURE ANALYSIS

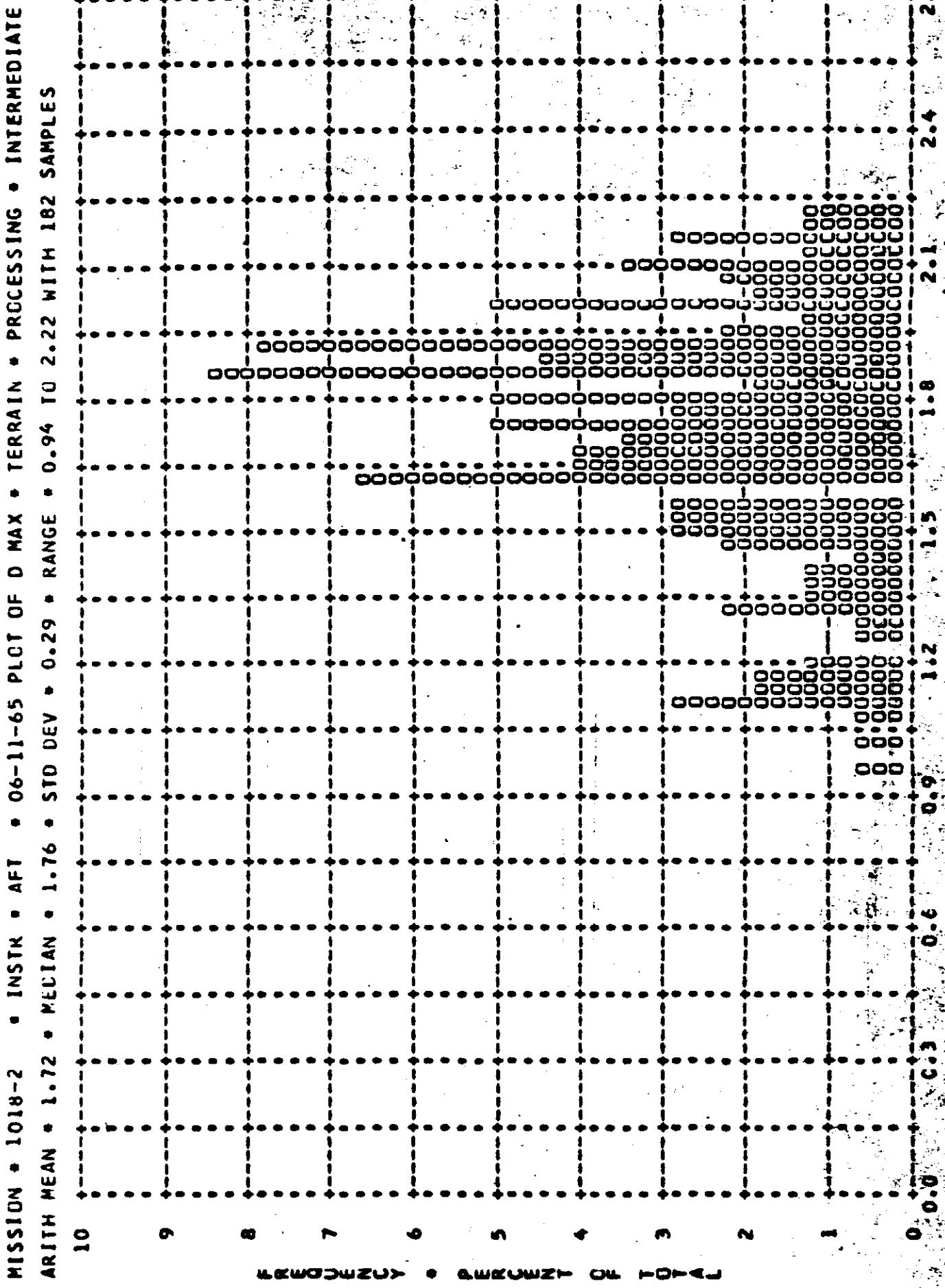
PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	C	C PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	182	0 PC	7 PC	63 PC	28 PC	3 PC
FULL	65	C PC	0 PC	45 PC	55 PC	0 PC
ALL LEVELS	247	C PC	5 PC	58 PC	35 PC	2 PC
PROCESS LEVEL	BASE + FOG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND UP
INTERMED	0.10-C.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UP
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND UP

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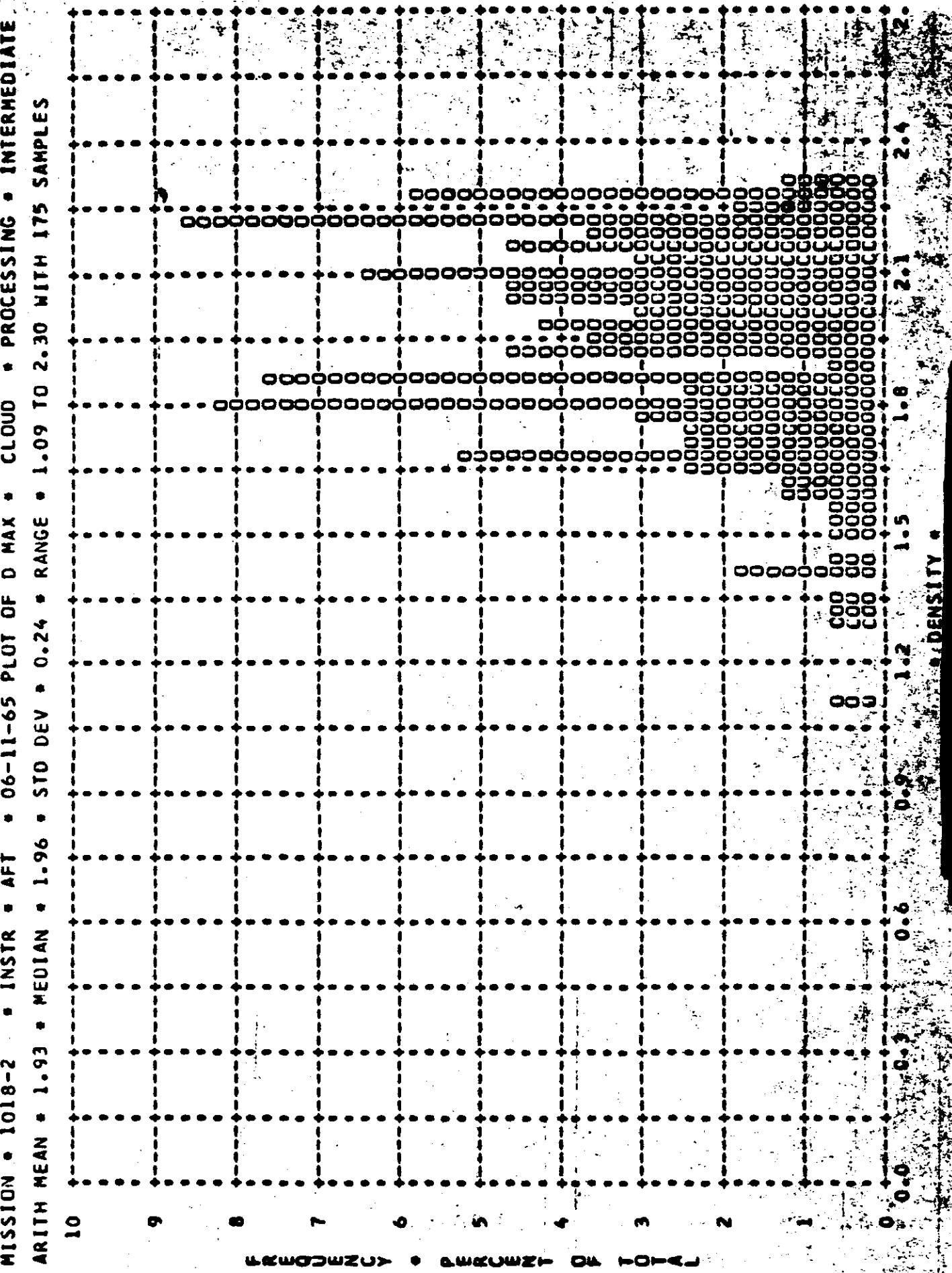
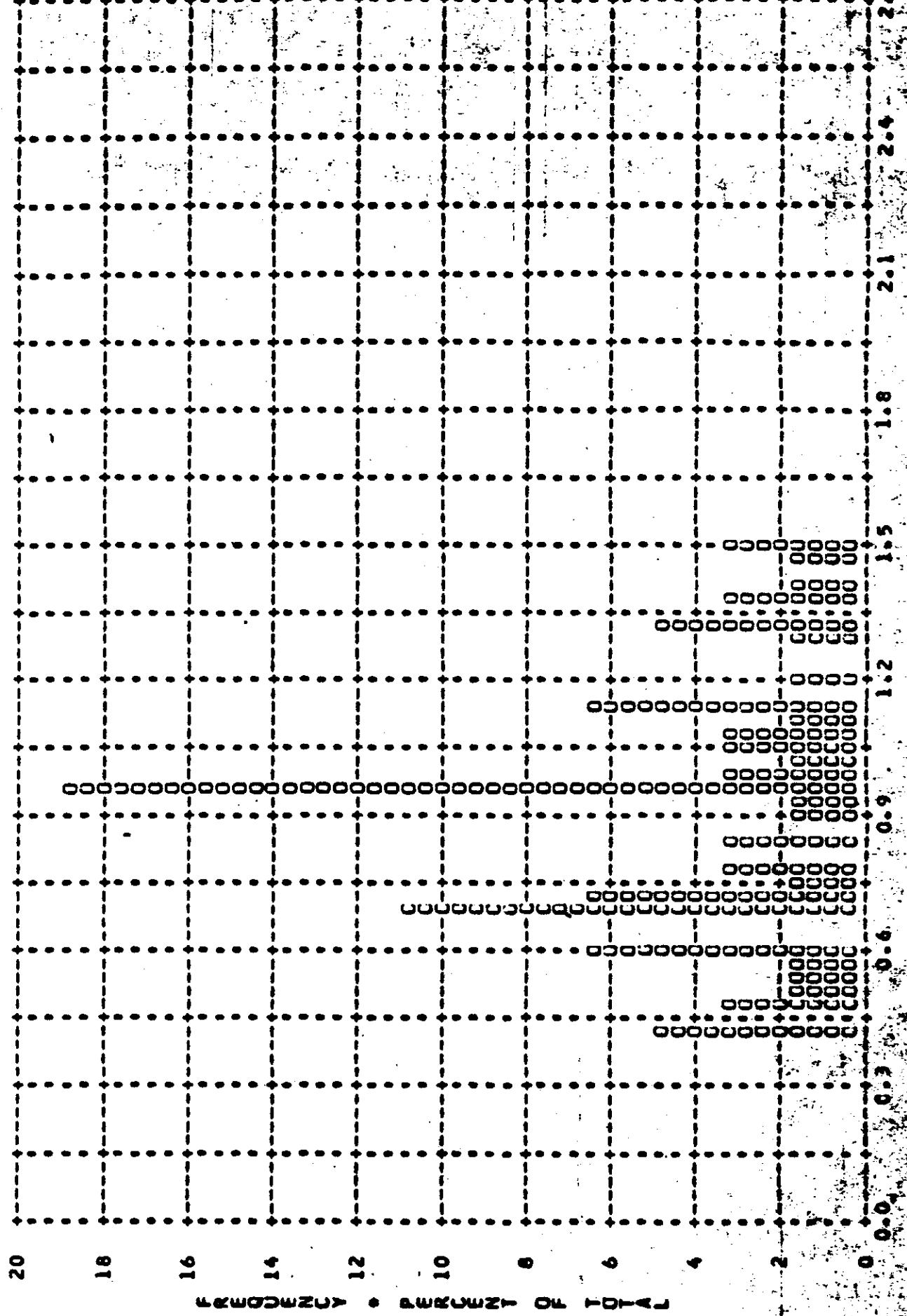


Figure A-38

MISSION \* 1018-2 \* INSTR. \* AFT \* 06-11-65 PLCT OF 0 MIN \* TERRAIN \* PROCESSING \* FULL  
ARITH MEAN \* 0.9C \* MEDIAN \* 0.94 \* STD DEV \* 0.29 \* RANGE \* 0.42 TO 1.48 WITH 65 SAMPLES



TOP SECRET

MISSION \* 1018-2 \* INSTR \* AFT \* 06-11-65 PLOT OF D MAX \* TERRAIN \* PROCESSING \* FULL  
ARITH MEAN \* 1.65 \* MEDIAN \* 1.68 \* STD DEV \* 0.28 \* RANGE \* 0.94 TO 2.26 WITH 65 SAMPLES

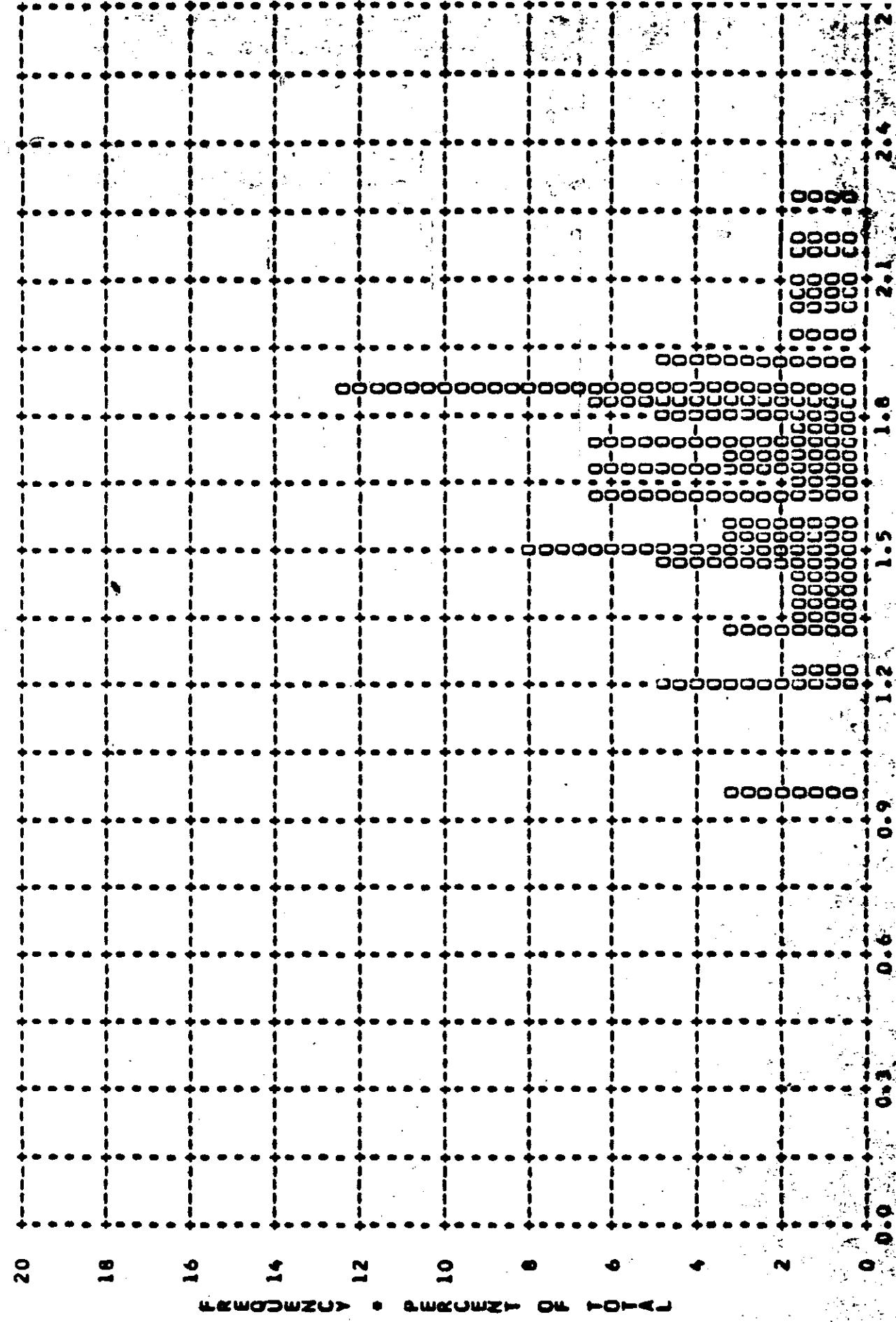
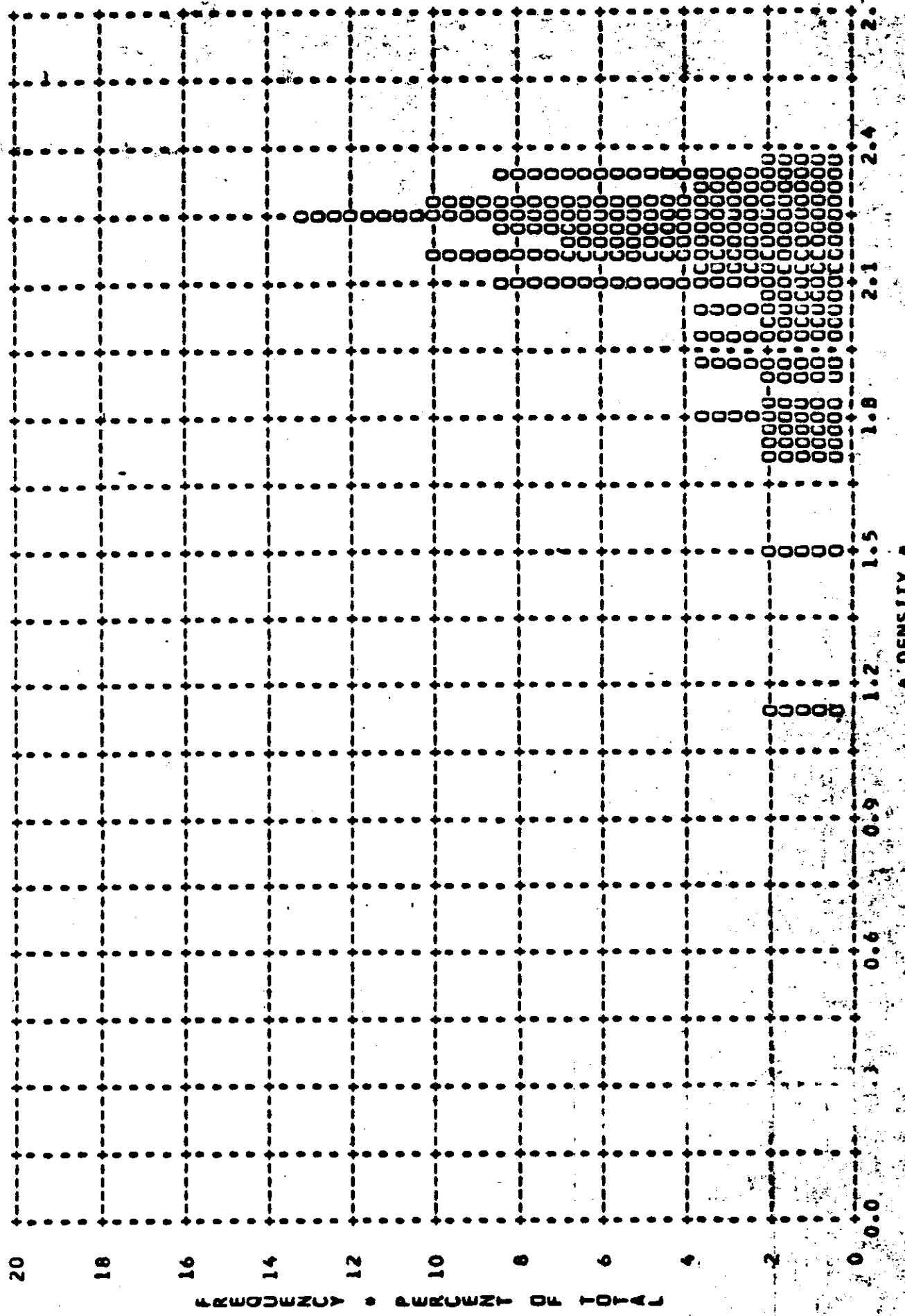


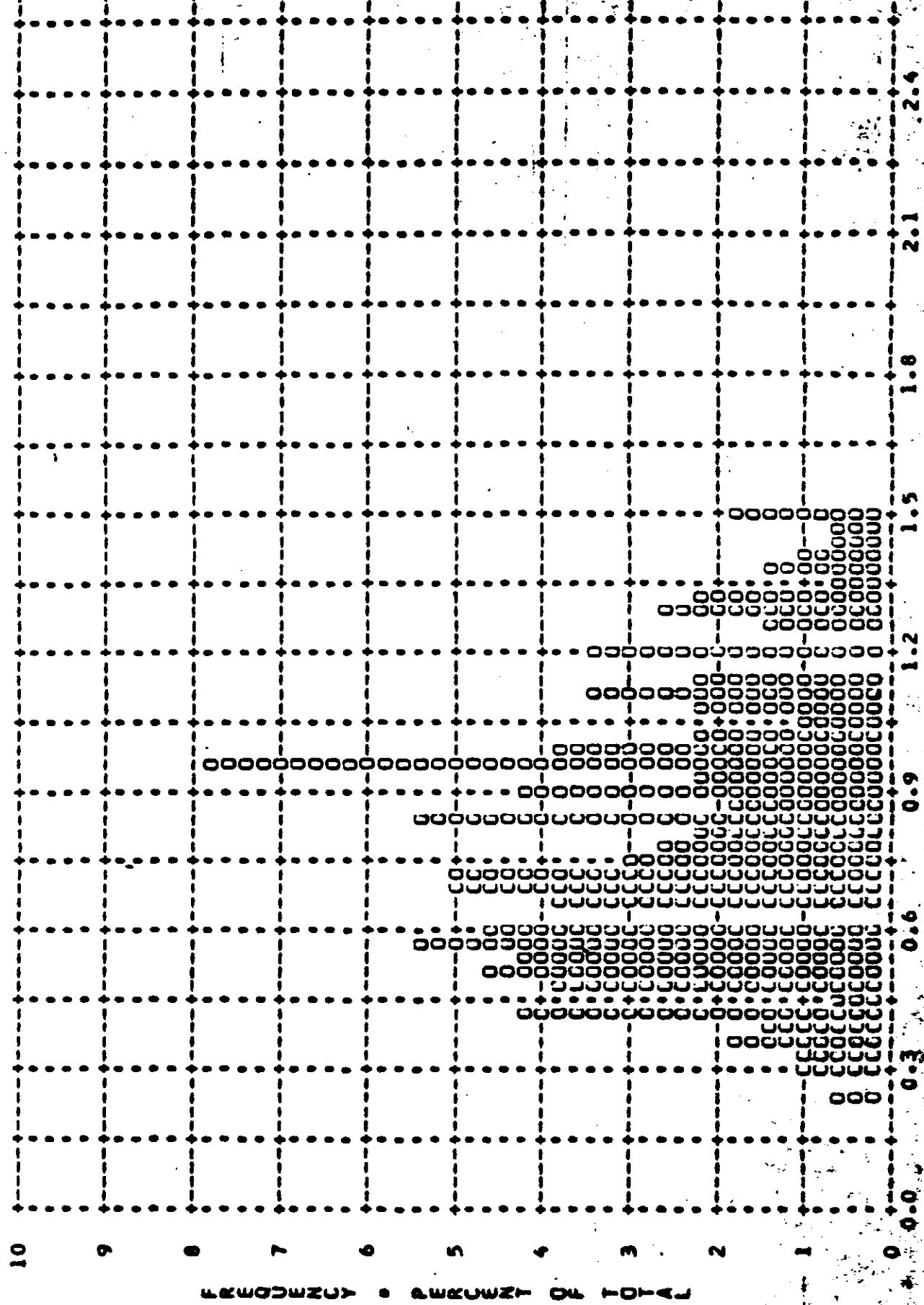
Figure A-41

~~TOP SECRET~~  
MISSION • 1018-2 • INSTR • AFT • 06-11-65 PLOT OF D MAX • CLOUD • PROCESSING • FULL  
ARITH MEAN • 2.11 • MEDIAN • 2.17 • STD DEV • 0.22 • RANGE • 1.12 TO 2.35 WITH 61 SAMPLES



TOP SECRET

MISSION \* 1018-2 \* INSTK \* AFI \* 06-11-65 PLOT OF D MIN \* TERRAIN \* PROCESSING \* ALL LEVELS  
ARITH MEAN \* 0.81 \* MEDIAN \* C.79 \* STD DEV \* 0.29 \* RANGE \* 0.22 TG 1.50 WITH 247 SAMPLES

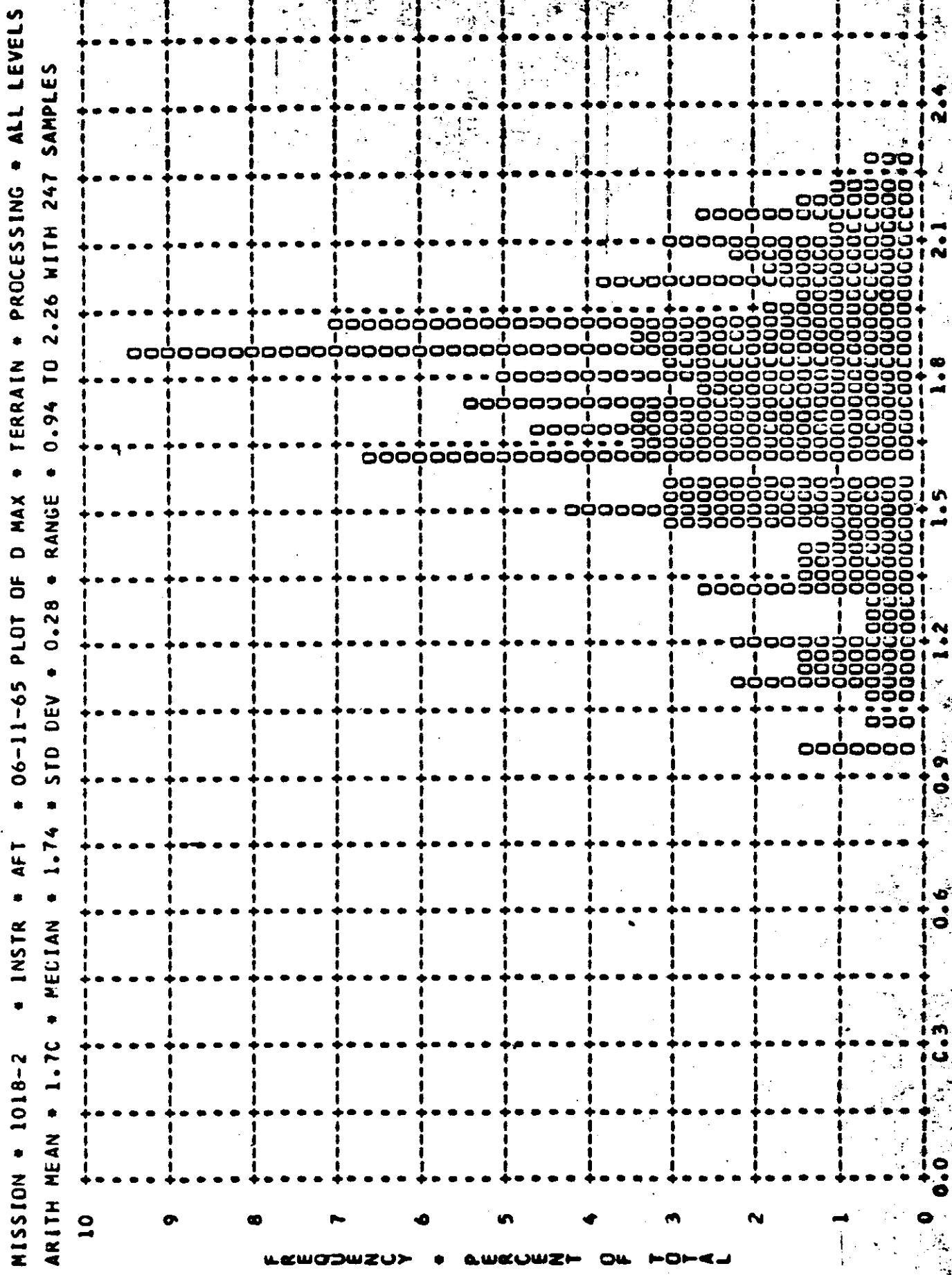


TOP SECRET

• DENSITY •

Figure A-43

~~TOP SECRET~~

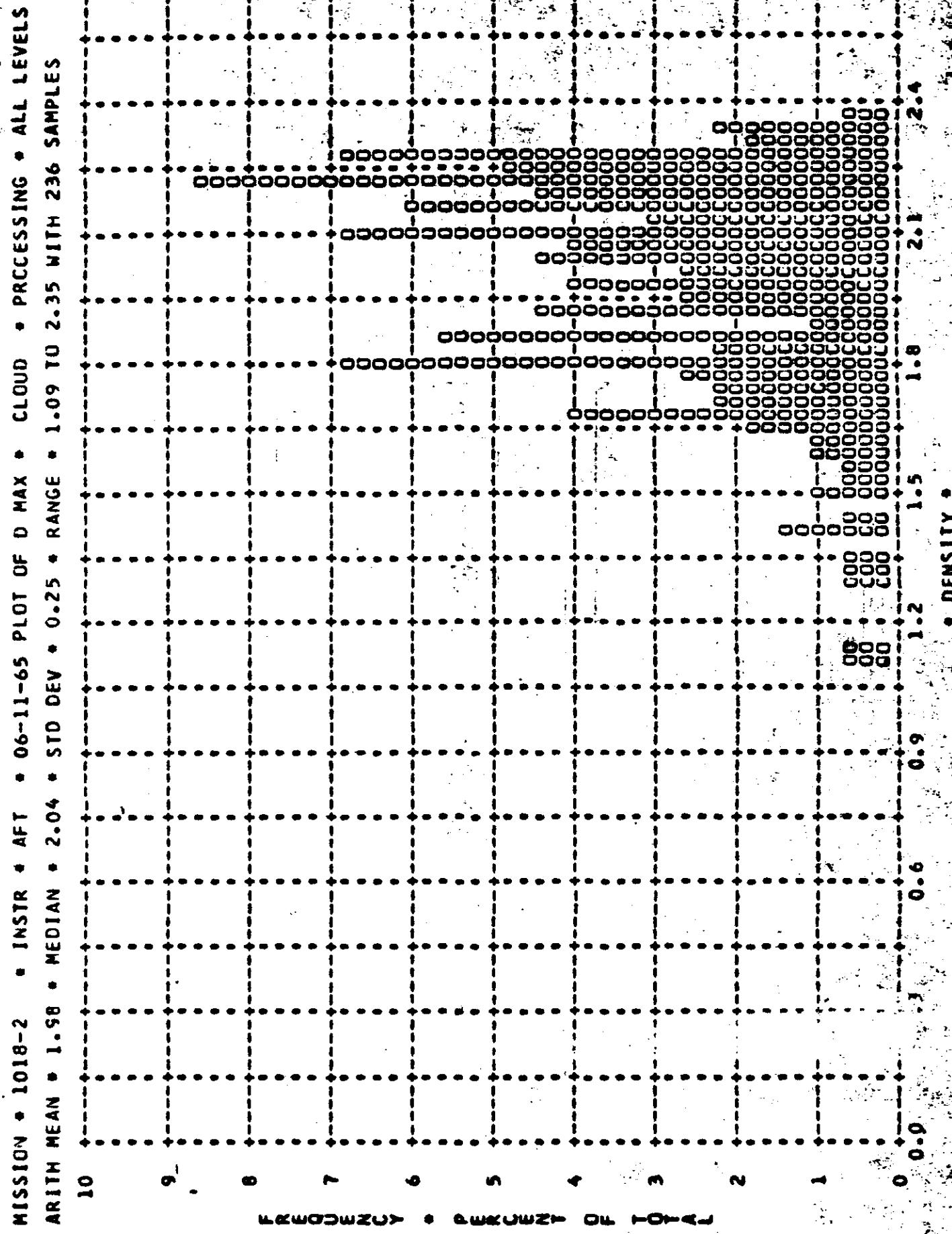


~~TOP SECRET~~

\* DENSITY \*

Figure A-44.

-TOP SECRET-



~~TOP SECRET C/~~

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